

FRACTURE OF COMPOSITE ORTHOTROPIC  
PLATES CONTAINING PERIODIC  
BUFFER STRIPS

F. Delale, 1976

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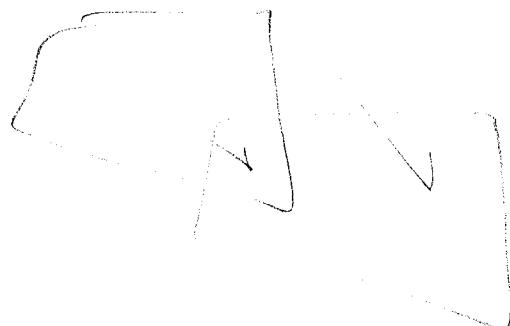
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FRACTURE OF COMPOSITE ORTHOTROPIC PLATES  
CONTAINING PERIODIC BUFFER STRIPS

by

Feridun Delale

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Fazil Erdogan

Professor in Charge  
Prof. Fazil Erdogan

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Dean P. Updike

Chairman  
Prof. Dean P. Updike

Arturs Kalnins

Prof. Arturs Kalnins

Peter Hilton

Prof. Peter Hilton

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## NOMENCLATURE

$A_{ij}$	- material constants for orthotropic materials
$E_x, E_x^*$	- elasticity moduli in x direction
$E_y, E_y^*$	- elasticity moduli in y direction
$E_z$	- elasticity modulus in z direction
$G_{xy}, G_{xy}^*$	- shear moduli in x-y plane
$G_{xz}$	- shear modulus in x-z plane
$G_{yz}$	- shear modulus in y-z plane
$a_{ij}$	- material constants for orthotropic materials
$a$	- half crack length for the crack in the first strip
$b$	- half crack length for the crack in the second strip
$c, d$	- crack tip coordinates in second strip
$h_1$	- half width of the first strip
$h_2$	- half width of the second strip
$\ell$	- half crack length for the case when the crack crosses the interface
$p_1, p_2$	- crack surface tractions
$u_i$	- displacements in x- direction
$v_i$	- displacements in y- direction
$w$	- displacements in z- direction
$x, y, z$	- cartesian coordinate system
$x_1, y$	- coordinates for the first strip
$x_2, y$	- coordinates for the second strip
$\beta_i, \beta_i^*$	- material constants defined by equations (2.15), (3.5) and (3.8)

$\epsilon_x, \epsilon_y, \epsilon_z,$	- components of strain tensor
$\gamma_{xy}, \gamma_{yz}, \gamma_{xz}$	
$\gamma_i, \gamma_i^*$	- material constants defined in Appendix A
$\lambda_i$	- material constants defined in Appendix A
$v_{xy}, v_{xy}^*$	- material constants for orthotropic materials
$v_{yx}, v_{yx}^*$	
$\phi, \phi^*$	- crack surface displacement derivatives
$w_i$	- material constants defined by equations (3.5)
$\sigma_{ix}, \sigma_{iy}, \sigma_z$	- components of stress tensor
$\tau_{ixy}, \tau_{xz}, \tau_{yz}$	

## ABSTRACT

The fracture problem of laminated plates which consist of bonded orthotropic layers is studied. It is assumed that the medium contains periodic cracks normal to the bimaterial interfaces and the external loads are applied away from the crack region. The field equations for an elastic orthotropic body are transformed to give the displacement and stress expressions for each layer or strip. The unknown functions in these expressions are found by satisfying the remaining boundary and continuity conditions. A system of singular integral equations is obtained from the mixed boundary conditions. Three cases are considered:

- a) The case of internal cracks
- b) The case of broken laminates
- c) The case of a crack crossing the interface.

The singular behavior around the crack tip and at the bimaterial interface is studied. It is shown that the crack surface displacement derivative has a power singularity for practical orthotropic materials when the crack touches the interface, i.e., for case (b). In studying the singular behavior at the bimaterial interfaces in case (c), it is found that for some orthotropic material combinations there is no singularity in the crack surface displacement derivatives and the stresses. In each case the stress intensity

factors are computed for various material combinations and various crack geometries. The results for orthotropic materials are discussed and are compared with those for isotropic materials.

## 1. INTRODUCTION

In structural design one of the most important considerations is the fracture of individual components. Although, fracture may not always mean total failure, it is considered in modern engineering as an important problem for safe and economic design of structures. It would be very attractive to develop special types of designs for which the structural resistance to fatigue crack propagation is improved. In the aerospace industry, the use of composite sheet materials with buffer strips parallel to the main load-carrying laminates seems to be such a design practice. The process of manufacturing composites gives the opportunity to improve the structural resistance to fatigue crack propagation by strengthening the material in certain directions. The increasing use of composites in structures generates new problems for the structural designer. Among these problems, we are mainly interested in the fracture of layered composite materials.

There are two main problems in studying the fracture of composites: the development of an appropriate failure criterion and a mathematical model for the calculation of the related load factor. The failure criterion affects the course of the analytical work in the sense that it is the failure criterion which generally determines the physical quantities that one should compute (such as the stress

intensity factor, the strain energy release rate, COD, etc.). There are many failure criteria or theories which are used to predict failure of structures. In Elastic Fracture Mechanics where only small scale yielding is allowed,  $K \leq K_{IC}$  is such a criterion. In this case failure occurs when the calculated value of the stress intensity factor reaches a critical value,  $K_{IC}$ , which can be determined experimentally as a material property. There are also other one-parameter failure criteria (such as critical plastic stress intensity factor  $K_{pc}$  and J integral) which have been recently proposed to predict failure from elastic to fully plastic range.  $K$  is a very highly effective correlation parameter in studying the fatigue crack propagation phenomena. In aerospace structures the basic problem is the nucleation and propagation of fatigue crack which may eventually reach a critical size causing catastrophic failure. That is why, in this study we focus our interest to the computation of the stress intensity factors and in the investigation of the singular behavior of the stress state around the crack tips.

In studying the fracture problem of composites, a mathematical model, which will reflect the geometrical and physical properties of the medium and the real mechanism of fracture, is needed. Because of mathematical difficulties and the lengthy computation that the analysis

requires, in the recent studies the geometry and the material properties have been considerably simplified. The problem of a multi-layered isotropic medium, which consists of many layers and where a crack normal to the interface can appear, has been treated by Hilton and Sih [1]. In this problem the geometry is simplified to a single layer between two dissimilar half-planes where the elastic properties are averaged. The same problem has been considered by Bogy [2]. The problem of a broken laminate between two half planes has been investigated by Ashbaugh [3] and Gupta [4]. The extension of the problem treated by Hilton and Sih to orthotropic media has been solved by Arin [5]. The fracture problem of a composite plate which consists of parallel load-carrying laminates and buffer strips has recently been solved by Erdogan and Bakoglu [6]. In this work the load carrying laminates and buffers are considered to be isotropic and linearly elastic. The orthotropic case of the problem treated in [3] and [4] has also been solved by Arin [7].

The objective of this work is to investigate the fracture problem of composite plates containing periodic buffer strips. The laminates and buffer strips are assumed to be linearly elastic and orthotropic. In general, this is the case in the actual plate and shell structures such as those, for example, which consist of boron-epoxy

composites. It is also assumed that the fatigue cracks may appear and propagate in main laminates, in buffer strips or in both normal to the interfaces. The external load is applied to the plate parallel to the strips and away from the crack region. Three different problems are studied: the internal crack problem, the case of broken laminates and the case of a crack crossing the interface. A general formulation of the problem is given for plane strain and generalized plane stress cases by the use of Fourier Integral Transform Technique. The singular behavior around ends and at the bimaterial interfaces is studied. The resulting singular integral equations are solved numerically and the stress intensity factors are calculated for various crack geometries and various material combinations.

## 2. ELASTICITY OF AN ANISOTROPIC ELASTIC BODY

For an anisotropic elastic body, in the absence of body forces, the equations which relate the field quantities can be written as follows:

### 2.1 The Equilibrium Equations

$$\begin{aligned}\frac{\partial \sigma_x}{\partial x} + \frac{\partial \tau_{xy}}{\partial y} + \frac{\partial \tau_{xz}}{\partial z} &= 0 \\ \frac{\partial \tau_{xy}}{\partial x} + \frac{\partial \sigma_y}{\partial y} + \frac{\partial \tau_{yz}}{\partial z} &= 0 \\ \frac{\partial \tau_{xz}}{\partial x} + \frac{\partial \tau_{yz}}{\partial y} + \frac{\partial \sigma_z}{\partial z} &= 0\end{aligned}\quad (2.1)$$

### 2.2 Strain-Displacement Relations

$$\begin{aligned}\epsilon_x &= \frac{\partial u}{\partial x}, \quad \epsilon_y = \frac{\partial v}{\partial y}, \quad \epsilon_z = \frac{\partial w}{\partial z} \\ \gamma_{yz} &= \frac{\partial v}{\partial z} + \frac{\partial w}{\partial y}, \quad \gamma_{xz} = \frac{\partial w}{\partial x} + \frac{\partial u}{\partial z}, \quad \gamma_{xy} = \frac{\partial u}{\partial y} + \frac{\partial v}{\partial x}\end{aligned}\quad (2.2)$$

### 2.3 Stress-Strain Relations

$$\begin{bmatrix} \sigma_x \\ \sigma_y \\ \sigma_z \\ \tau_{yz} \\ \tau_{xz} \\ \tau_{xy} \end{bmatrix} = [A_{ij}] \begin{bmatrix} \epsilon_x \\ \epsilon_y \\ \epsilon_z \\ \gamma_{yz} \\ \gamma_{xz} \\ \gamma_{xy} \end{bmatrix}, \quad i, j = 1, 6, \quad A_{ij} = A_{ji} \quad (2.3)$$

## 2.4 The Field Equations for an Orthotropic Body

For an orthotropic solid the matrix  $[A_{ij}]$  is:

$$[A_{ij}] = \begin{bmatrix} A_{11} & A_{12} & A_{13} & 0 & 0 & 0 \\ A_{12} & A_{22} & A_{23} & 0 & 0 & 0 \\ A_{13} & A_{23} & A_{33} & 0 & 0 & 0 \\ 0 & 0 & 0 & A_{44} & 0 & 0 \\ 0 & 0 & 0 & 0 & A_{55} & 0 \\ 0 & 0 & 0 & 0 & 0 & A_{66} \end{bmatrix} \quad (2.4)$$

Defining the inverse of  $[A_{ij}]$  by

$$[a_{ij}] = [A_{ij}]^{-1}$$

for orthotropic materials we have:

$$[a_{ij}] = \begin{bmatrix} \frac{1}{E_x} & -\frac{v_{yx}}{E_y} & -\frac{v_{zx}}{E_z} & 0 & 0 & 0 \\ -\frac{v_{xy}}{E_x} & \frac{1}{E_y} & -\frac{v_{zy}}{E_z} & 0 & 0 & 0 \\ -\frac{v_{xz}}{E_x} & -\frac{v_{yz}}{E_y} & \frac{1}{E_z} & 0 & 0 & 0 \\ 0 & 0 & 0 & \frac{1}{G_{yz}} & 0 & 0 \\ 0 & 0 & 0 & 0 & \frac{1}{G_{xz}} & 0 \\ 0 & 0 & 0 & 0 & 0 & \frac{1}{G_{xy}} \end{bmatrix}$$

$$E_x v_{yx} = E_y v_{xy}, \quad E_y v_{zy} = E_z v_{yz}, \quad E_z v_{xz} = E_x v_{zx}$$

(2.5)

Substituting (2.2) into (2.3) and using (2.1) and (2.4), the stresses and the equilibrium equations can be expressed in terms of the displacements as follows:

$$\begin{aligned}
 \sigma_x &= A_{11} \frac{\partial u}{\partial x} + A_{12} \frac{\partial v}{\partial y} + A_{13} \frac{\partial w}{\partial z} \\
 \sigma_y &= A_{12} \frac{\partial u}{\partial x} + A_{22} \frac{\partial v}{\partial y} + A_{23} \frac{\partial w}{\partial z} \\
 \sigma_z &= A_{13} \frac{\partial u}{\partial x} + A_{23} \frac{\partial v}{\partial y} + A_{33} \frac{\partial w}{\partial z} \\
 \tau_{yz} &= A_{44} \left( \frac{\partial v}{\partial z} + \frac{\partial w}{\partial y} \right) \\
 \tau_{xz} &= A_{55} \left( \frac{\partial w}{\partial x} + \frac{\partial u}{\partial z} \right) \\
 \tau_{xy} &= A_{66} \left( \frac{\partial u}{\partial y} + \frac{\partial v}{\partial x} \right) \quad (2.6)
 \end{aligned}$$

$$\begin{aligned}
 A_{11} \frac{\partial^2 u}{\partial x^2} + A_{66} \frac{\partial^2 u}{\partial y^2} + A_{55} \frac{\partial^2 u}{\partial z^2} + (A_{12} + A_{66}) \frac{\partial^2 v}{\partial x \partial y} + (A_{13} + A_{55}) \frac{\partial^2 w}{\partial x \partial z} &= 0 \\
 A_{66} \frac{\partial^2 v}{\partial x^2} + A_{22} \frac{\partial^2 v}{\partial y^2} + A_{44} \frac{\partial^2 v}{\partial z^2} + (A_{12} + A_{66}) \frac{\partial^2 u}{\partial x \partial y} + (A_{23} + A_{44}) \frac{\partial^2 w}{\partial y \partial z} &= 0 \\
 A_{55} \frac{\partial^2 w}{\partial x^2} + A_{44} \frac{\partial^2 w}{\partial y^2} + A_{33} \frac{\partial^2 w}{\partial z^2} + (A_{13} + A_{55}) \frac{\partial^2 u}{\partial x \partial z} + (A_{23} + A_{44}) \frac{\partial^2 v}{\partial y \partial z} &= 0 \quad (2.7)
 \end{aligned}$$

#### 2.4.1 Case of Plane Strain

For the plane strain case we have:

$$u = u(x, y), \quad v = v(x, y), \quad w = 0 \quad (2.8)$$

and from (2.2),

$$\begin{aligned}\epsilon_x &= \frac{\partial u}{\partial x}, \quad \epsilon_y = \frac{\partial v}{\partial y}, \quad \gamma_{xy} = \frac{\partial u}{\partial y} + \frac{\partial v}{\partial x} \\ \epsilon_z &= 0, \quad \gamma_{yz} = 0, \quad \gamma_{xz} = 0\end{aligned}\quad (2.9)$$

Thus, the stress-displacement relations and the equilibrium equations become:

$$\begin{aligned}\sigma_x &= A_{11} \frac{\partial u}{\partial x} + A_{12} \frac{\partial v}{\partial y} \\ \sigma_y &= A_{12} \frac{\partial u}{\partial x} + A_{22} \frac{\partial v}{\partial y} \\ \sigma_z &= A_{13} \frac{\partial u}{\partial x} + A_{23} \frac{\partial v}{\partial y} \\ \tau_{xy} &= A_{66} \left( \frac{\partial u}{\partial y} + \frac{\partial v}{\partial x} \right) \\ \tau_{yz} &= \tau_{xz} = 0\end{aligned}\quad (2.10)$$

$$\begin{aligned}A_{11} \frac{\partial^2 u}{\partial x^2} + A_{66} \frac{\partial^2 u}{\partial y^2} + (A_{12} + A_{66}) \frac{\partial^2 v}{\partial x \partial y} &= 0 \\ A_{66} \frac{\partial^2 v}{\partial x^2} + A_{22} \frac{\partial^2 v}{\partial y^2} + (A_{12} + A_{66}) \frac{\partial^2 u}{\partial x \partial y} &= 0\end{aligned}\quad (2.11)$$

#### 2.4.2 Case of Generalized Plane Stress

In this case since  $\sigma_z = \tau_{xz} = \tau_{yz} = 0$  from (2.3), for the average stresses and strains we can write:

$$\begin{bmatrix} \epsilon_x \\ \epsilon_y \\ \gamma_{xy} \end{bmatrix} = [\bar{a}] \quad \begin{bmatrix} \sigma_x \\ \sigma_y \\ \tau_{xy} \end{bmatrix}$$

where

$$[\bar{A}] = \begin{bmatrix} \frac{1}{E_x} & -\frac{v_{yx}}{E_y} & 0 \\ -\frac{v_{xy}}{E_x} & \frac{1}{E_y} & 0 \\ 0 & 0 & \frac{1}{G_{xy}} \end{bmatrix}, \quad [\bar{A}] = \begin{bmatrix} \frac{1}{E_y \cdot \Delta} & \frac{v_{yx}}{E_y \cdot \Delta} & 0 \\ \frac{v_{xy}}{E_x \cdot \Delta} & \frac{1}{E_x \cdot \Delta} & 0 \\ 0 & 0 & G_{xy} \end{bmatrix},$$

$$\text{and } \Delta = \frac{1}{E_x \cdot E_y} (1 - v_{xy} v_{yx}). \quad (2.12)$$

The equilibrium equations reduce to:

$$\begin{aligned} \bar{A}_{11} \frac{\partial^2 u}{\partial x^2} + \bar{A}_{33} \frac{\partial^2 u}{\partial y^2} + (\bar{A}_{12} + \bar{A}_{33}) \frac{\partial^2 v}{\partial x \partial y} &= 0 \\ \bar{A}_{33} \frac{\partial^2 v}{\partial x^2} + \bar{A}_{22} \frac{\partial^2 v}{\partial y^2} + (\bar{A}_{12} + \bar{A}_{33}) \frac{\partial^2 u}{\partial x \partial y} &= 0 \end{aligned} \quad (2.13)$$

Considering the structure of equations (2.11) and (2.13), the equilibrium equations can be written for plane strain and generalized plane stress cases in the following form:

$$\begin{aligned} \beta_1 \frac{\partial^2 u}{\partial x^2} + \beta_2 \frac{\partial^2 u}{\partial y^2} + \beta_3 \frac{\partial^2 v}{\partial y \partial x} &= 0 \\ \frac{\partial^2 v}{\partial x^2} + \beta_2 \frac{\partial^2 v}{\partial y^2} + \beta_3 \frac{\partial^2 u}{\partial x \partial y} &= 0 \end{aligned} \quad (2.14)$$

where

$$\beta_1 = \frac{A_{11}}{A_{66}}, \quad \beta_2 = \frac{A_{22}}{A_{66}}, \quad \beta_3 = 1 + \frac{A_{12}}{A_{66}} \quad \text{for plane strain}$$

and

$$\beta_1 = \frac{\bar{A}_{11}}{\bar{A}_{33}}, \quad \beta_2 = \frac{\bar{A}_{22}}{\bar{A}_{33}}, \quad \beta_3 = 1 + \frac{\bar{A}_{12}}{\bar{A}_{33}} \quad \text{for plane stress.} \quad (2.15)$$

### 3. DISPLACEMENT AND STRESS FIELDS FOR STRIPS

The two-dimensional composite medium is formed of two sets of periodically arranged strips having widths  $2h_1$  and  $2h_2$  as shown in Figure 1. They are perfectly bonded along their straight boundaries, and contain symmetrically located cracks normal to the interfaces, of length  $2a$  and  $2b$  respectively. The load is applied away from the crack region, such that the crack plane is a plane of symmetry.

Using the usual superposition technique, the solution of the actual traction-free crack problem may be obtained by superposing the homogeneous uncracked strip solution to the solution of a cracked strip loaded with self-equilibrating crack surface tractions (see Figure 2). Since we are interested only in the computation of stress intensity factors and the singular behavior of the stresses around crack ends, we will consider only the singular part of the solution, where the self-equilibrating crack tractions are the only external forces.

First we will find solutions to (2.14) satisfying certain boundary conditions of the strips. The combination of these solutions will be forced to satisfy the remaining boundary and continuity conditions.

### 3.1 Solutions $u^{(a)}(x,y)$ , $v^{(a)}(x,y)$

Assume:

$$u^{(a)}(x,y) = \frac{2}{\pi} \int_0^{\infty} f(\alpha, x) \cos \alpha y \, d\alpha$$

$$v^{(a)}(x,y) = \frac{2}{\pi} \int_0^{\infty} g(\alpha, x) \sin \alpha y \, d\alpha \quad (3.1)$$

Substituting (3.1) into (2.14) we obtain:

$$\beta_1 \frac{d^2 f}{dx^2} - \alpha^2 f + \beta_3 \alpha \frac{df}{dx} = 0$$

$$\frac{d^2 g}{dx^2} - \beta_2 \alpha^2 g - \beta_3 \alpha \frac{dg}{dx} = 0 \quad (3.2)$$

The solution of (3.2) can be written as:

$$f(\alpha, x) = A(\alpha) e^{s_1 \alpha x} + B(\alpha) e^{-s_1 \alpha x} + C(\alpha) e^{s_2 \alpha x} + D(\alpha) e^{-s_2 \alpha x}$$

$$g(\alpha, x) = \beta_7 [A(\alpha) e^{s_1 \alpha x} - B(\alpha) e^{-s_1 \alpha x}] + \beta_8 [C(\alpha) e^{s_2 \alpha x} - D(\alpha) e^{-s_2 \alpha x}] \quad (3.3)$$

where  $s_1$  and  $s_2$  are the roots of

$$s^4 + \beta_4 s^2 + \beta_5 = 0 \quad (3.4)$$

and

$$\beta_4 = \frac{\beta_3^2 - \beta_1 \beta_2 - 1}{\beta_1}, \quad \beta_5 = \frac{\beta_2}{\beta_1}, \quad \beta_6 = \sqrt{\beta_4^2 - 4\beta_5},$$

$$\beta_7 = \frac{1 - \beta_1 s_1^2}{\beta_3 s_1}, \quad \beta_8 = \frac{1 - \beta_1 s_2^2}{\beta_3 s_2}$$

From (3.4) we can write :

$$s_1 = w_1 + iw_2 = \sqrt{(-\beta_4 + \beta_6)/2}$$

$$s_2 = w_3 + iw_4 = \sqrt{(-\beta_4 - \beta_6)/2}$$

$$s_3 = -s_1, \quad s_4 = -s_2 \quad (3.5)$$

$s_1$  and  $s_2$  are both real or complex conjugates.

### 3.2 Solutions $u^{(b)}(x,y)$ , $v^{(b)}(x,y)$

Assume:

$$u^{(b)}(x,y) = \frac{2}{\pi} \int_0^{\infty} h(\alpha, y) \sin \alpha x \, d\alpha$$

$$v^{(b)}(x,y) = \frac{2}{\pi} \int_0^{\infty} l(\alpha, y) \cos \alpha x \, d\alpha \quad (3.6)$$

Substituting (3.6) into (2.14) we have:

$$\begin{aligned} \frac{d^2h}{dy^2} - \beta_3 \alpha \frac{dh}{dy} - \beta_1 \alpha^2 h &= 0 \\ \beta_2 \frac{d^2l}{dy^2} + \beta_3 \alpha \frac{dl}{dy} - \alpha^2 l &= 0 \end{aligned} \quad (3.7)$$

Solving (3.7) we obtain:

$$h(\alpha, y) = E(\alpha) e^{s_1 \alpha y / \sqrt{\beta_5}} + F(\alpha) e^{-s_1 \alpha y / \sqrt{\beta_5}}$$

$$+ G(\alpha) e^{s_2 \alpha y / \sqrt{\beta_5}} + H(\alpha) e^{-s_2 \alpha y / \sqrt{\beta_5}},$$

and

$$l(\alpha, y) = \beta_9 \left[ E(\alpha) e^{s_1 \alpha y / \sqrt{\beta_5}} - F(\alpha) e^{-s_1 \alpha y / \sqrt{\beta_5}} \right]$$

$$+ \beta_{10} \left[ G(\alpha) e^{s_2 \alpha y / \sqrt{\beta_s}} - H(\alpha) e^{-s_2 \alpha y / \sqrt{\beta_s}} \right]$$

where

$$\begin{aligned}\beta_9 &= \frac{1}{\beta_3} \left\{ -\frac{\beta_1 \sqrt{\beta_s}}{s_1} + \frac{s_1}{\sqrt{\beta_s}} \right\} \\ \beta_{10} &= \frac{1}{\beta_3} \left\{ -\frac{\beta_1 \sqrt{\beta_s}}{s_2} + \frac{s_2}{\sqrt{\beta_s}} \right\}\end{aligned}\quad (3.8)$$

A superscript \* will be used for the material constants and unknown functions when the above expressions are used for the second strip.

If one examines the roots of equation (3.4), he will realize that there are two types of orthotropic materials. We will denote the material as type I when  $s_1$  and  $s_2$  are real, and as type II when they are complex conjugates. We will assume, in our analysis, that the materials of both strips are of type I. Similar analysis can be done for the remaining combinations.

### 3.3 The Displacements

For each strip, we can write:

$$u(x,y) = u^{(a)}(x,y) + u^{(b)}(x,y)$$

$$v(x,y) = v^{(a)}(x,y) + v^{(b)}(x,y)$$

Noting that:

$$u(x,y) = -u(-x,y) \text{ and } v(x,y) = -v(x,-y)$$

we will obtain:

$$B(\alpha) = -A'(\alpha), D(\alpha) = -C(\alpha), F(\alpha) = -E(\alpha), H(\alpha) = -G(\alpha).$$

For material type I;

$$s_1 = w_1, s_2 = w_3 \text{ and } \beta_6, \beta_7, \beta_8, \beta_9, \beta_{10} \text{ are real.}$$

Using the information given above, and keeping in mind that  $u$  and  $v$  vanish when  $y$  goes to infinity, for  $y > 0$ , the displacement expressions can be written as follows:

$$\begin{aligned} u(x,y) &= \frac{4}{\pi} \int_0^{\infty} [A(\alpha) \sinh(w_1 \alpha x) + C(\alpha) \sinh(w_3 \alpha x)] \cos \alpha y d\alpha \\ &\quad + \frac{2}{\pi} \int_0^{\infty} \left[ E(\alpha) e^{-|w_1| \alpha y / \sqrt{\beta_5}} + G(\alpha) e^{-|w_3| \alpha y / \sqrt{\beta_5}} \right] \sin \alpha x d\alpha \\ v(x,y) &= \frac{4}{\pi} \int_0^{\infty} [\beta_7 A(\alpha) \cosh(w_1 \alpha x) + \beta_8 C(\alpha) \cosh(w_3 \alpha x)] \sin \alpha y d\alpha \\ &\quad - \frac{2}{\pi} \int_0^{\infty} \left[ \text{sign}(w_1) \beta_9 E(\alpha) e^{-|w_1| \alpha y / \sqrt{\beta_5}} + \text{sign}(w_3) \beta_{10} G(\alpha) e^{-|w_3| \alpha y / \sqrt{\beta_5}} \right] \\ &\quad \cdot \cos \alpha x d\alpha \end{aligned} \quad (3.9)$$

### 3.4 The stresses

For generalized plane stress case:

$$\sigma_x = \bar{A}_{11} \epsilon_x + \bar{A}_{12} \epsilon_y$$

$$\sigma_y = \bar{A}_{12} \epsilon_x + \bar{A}_{22} \epsilon_y$$

$$\tau_{xy} = \bar{A}_{33} \gamma_{xy} \quad (3.10)$$

Differentiating (3.9) and using (2.12) and (3.10),  
the stress expressions can be written as:

$$\frac{\pi(1-v_{xy}v_{yx})}{2E_x} \sigma_x(x,y) = \int_0^\infty \left[ \gamma_1 E(\alpha) e^{-|w_1|\alpha y/\sqrt{\beta_5}} + \gamma_2 G(\alpha) \right. \\ \left. e^{-|w_3|\alpha y/\sqrt{\beta_5}} \right] \alpha \cos \alpha x d\alpha + \int_0^\infty [2\gamma_3 A(\alpha) \cosh(w_1 \alpha x) + 2\gamma_4 C(\alpha)]$$

$$[\cosh(w_3 \alpha x)] \alpha \cos \alpha y d\alpha$$

$$\frac{\pi(1-v_{xy}v_{yx})}{2E_y} \sigma_y(x,y) = \int_0^\infty \left[ \gamma_5 E(\alpha) e^{-|w_1|\alpha y/\sqrt{\beta_5}} + \gamma_6 G(\alpha) \right. \\ \left. e^{-|w_3|\alpha y/\sqrt{\beta_5}} \right] \alpha \cos \alpha x d\alpha + \int_0^\infty [2\gamma_7 A(\alpha) \cosh(w_1 \alpha x) + 2\gamma_8 C(\alpha)]$$

$$[\cosh(w_3 \alpha x)] \alpha \cos \alpha y d\alpha$$

$$\frac{\pi}{2G_{xy}} \tau_{xy}(x,y) = \int_0^\infty [2\gamma_9 A(\alpha) \sinh(w_1 \alpha x) + 2\gamma_{10} C(\alpha)] \\ \cdot [\sinh(w_3 \alpha x)] \alpha \sin \alpha y d\alpha + \int_0^\infty \left[ \gamma_{11} E(\alpha) e^{-|w_1|\alpha y/\sqrt{\beta_5}} \right. \\ \left. + \gamma_{12} G(\alpha) e^{-|w_3|\alpha y/\sqrt{\beta_5}} \right] \alpha \sin \alpha x d\alpha \quad (3.11)$$

These expressions are valid also for the plane strain case with the following substitutions:

$$v_{yx} = A_{12}/A_{11}, \quad v_{xy} = A_{12}/A_{22}, \quad (E_y \Delta) = 1/A_{11}, \\ (E_x \Delta) = 1/A_{22}$$

The elastic material constants  $\gamma_j$  are defined in Appendix A.

#### 4. FORMULATION OF THE PROBLEM

The solution of the problem may be obtained by determining the unknown functions which appear in the displacement and stress expressions, under the following boundary and continuity conditions:

$$u_1(h_1, y) = u_2(-h_2, y) ,$$

$$v_1(h_1, y) = v_2(-h_2, y) \quad (0 \leq y < \infty) \quad (4.1a, b)$$

$$\sigma_{1x}(h_1, y) = \sigma_{2x}(-h_2, y) ,$$

$$\tau_{1xy}(h_1, y) = \tau_{2xy}(-h_2, y) \quad (0 \leq y < \infty) \quad (4.2a, b)$$

$$u_1(0, y) = 0 , \quad \tau_{1xy}(0, y) = 0 \quad (0 \leq y < \infty) \quad (4.3a, b)$$

$$u_2(0, y) = 0 , \quad \tau_{2xy}(0, y) = 0 \quad (0 \leq y < \infty) \quad (4.4a, b)$$

$$\tau_{1xy}(x_1, 0) = 0 , \quad |x_1| < h_1 ,$$

$$\tau_{2xy}(x_2, 0) = 0 , \quad |x_2| < h_2 \quad (4.5a, b)$$

$$\sigma_{1y}(x_1, 0) = -p_1(x_1) , \quad |x_1| < a ,$$

$$v_1(x_1, 0) = 0 , \quad a < |x_1| < h_1 \quad (4.6a, b)$$

$$\sigma_{2y}(x_2, 0) = -p_2(x_2) , \quad |x_2| < b ,$$

$$v_2(x_2, 0) = 0 , \quad b < |x_2| < h_2 \quad (4.7a, b)$$

The conditions (4.3a,b) and (4.4a,b) are satisfied identically.

Using (4.5a,b) we obtain:

$$G(\alpha) = -\frac{\gamma_{11}}{\gamma_{12}} E(\alpha), \quad G^*(\alpha) = -\frac{\gamma_{11}^*}{\gamma_{12}^*} E^*(\alpha)$$

The mixed condition (4.6) gives:

$$\begin{aligned} & \lim_{y \rightarrow 0^+} \int_0^\infty E(\alpha) \left[ \gamma_5 e^{-|w_1|\alpha y/\sqrt{\beta_5}} - \gamma_6 \frac{\gamma_{11}}{\gamma_{12}} e^{-|w_3|\alpha y/\sqrt{\beta_5}} \right] \cos \alpha x_1 d\alpha \\ & + \int_0^\infty [2\gamma_7 A(\alpha) \cosh(w_1 \alpha x_1) + 2\gamma_8 C(\alpha) \cosh(w_3 \alpha x_1)] \cos \alpha y d\alpha \\ & = -\frac{\pi(1-v_{xy}v_{yx})}{2E_y} p_1(x_1), \quad |x_1| < a \end{aligned} \quad (4.8a)$$

and

$$v_1(x_1, 0) = -\frac{2}{\pi} \gamma_{13} \int_0^\infty E(\alpha) \cos \alpha x_1 d\alpha = 0 \quad a < |x_1| < h_1 \quad (4.8b)$$

Define,

$$\frac{\partial v_1(x_1, 0)}{\partial x_1} = \phi(x_1) \text{ such that } \phi(x_1) = 0 \text{ for } |x_1| > a. \quad (4.9)$$

Differentiating (4.8b) with respect to  $x_1$  and taking the inverse transform, we obtain:

$$\gamma_{13} \alpha E(\alpha) = \int_0^a \phi(x_1) \sin \alpha x_1 dx_1. \quad (4.10)$$

If we now substitute (4.10) into (4.8a) and evaluate some of the integrals in closed form (see Appendix C) we will end up with the following singular integral equation:

$$\begin{aligned} \gamma_{14} \int_{-a}^a \frac{\phi(t)}{t-x_1} dt + \int_0^\infty [2\gamma_7 A(\alpha) \cosh(w_1 \alpha x_1) + 2\gamma_8 C(\alpha) \cosh(w_3 \alpha x_1)] \alpha d\alpha \\ = - \frac{\pi(1-v_{xy} v_{yx})}{2E_y} p_1(x_1) \quad -a < x_1 < a \end{aligned} \quad (4.11)$$

where because of symmetry  $\phi(t) = -\phi(-t)$ .

Similarly defining,

$$\frac{\partial v_2(x_2, 0)}{\partial x_2} = \phi^*(x_2) \text{ such that } \phi^*(x_2) = 0 \text{ for } |x_2| > b \quad (4.12)$$

and using the mixed condition (4.7) by the same procedure we obtain:

$$\begin{aligned} \gamma_{14}^* \int_{-b}^b \frac{\phi^*(t)}{t-x_2} dt + \int_0^\infty [2\gamma_7^* A^*(\alpha) \cosh(w_1^* \alpha x_2) + 2\gamma_8^* C^*(\alpha) \cosh(w_3^* \alpha x_2)] \alpha d\alpha \\ = - \frac{\pi(1-v_{xy}^* v_{yx}^*)}{2E_y^*} p_2(x_2) \quad -b < x_2 < b \end{aligned} \quad (4.13)$$

The next step is to determine the unknown functions  $A(\alpha)$ ,  $C(\alpha)$ ,  $A^*(\alpha)$ ,  $C^*(\alpha)$ . This can be done by using the continuity conditions (4.1a,b) and (4.2a,b) and taking the inverse transforms. Then we obtain the following system of linear equations:

$$\begin{aligned} 2A(\alpha) \sinh(w_1 \alpha h_1) + 2C(\alpha) \sinh(w_3 \alpha h_1) \\ + 2A^*(\alpha) \sinh(w_1^* \alpha h_2) + 2C^*(\alpha) \sinh(w_3^* \alpha h_2) = R_1(\alpha) \\ 2\beta_7 A(\alpha) \cosh(w_1 \alpha h_1) + 2\beta_8 C(\alpha) \cosh(w_3 \alpha h_1) \\ - 2\beta_7^* A^*(\alpha) \cosh(w_1^* \alpha h_2) - 2\beta_8^* C^*(\alpha) \cosh(w_3^* \alpha h_2) = R_2(\alpha) \end{aligned}$$

$$\begin{aligned}
& 2\gamma_3 A(\alpha) \cosh(w_1 \alpha h_1) + 2\gamma_4 C(\alpha) \cosh(w_3 \alpha h_1) \\
& - 2\lambda_1 \gamma_3^* A^*(\alpha) \cosh(w_1^* \alpha h_2) - 2\lambda_1 \gamma_4^* C^*(\alpha) \cosh(w_3^* \alpha h_2) = R_3(\alpha) \\
& 2\gamma_9 A(\alpha) \sinh(w_1 \alpha h_1) + 2\gamma_{10} C(\alpha) \sinh(w_3 \alpha h_1) \\
& + 2\gamma_9^* \lambda_2 A^*(\alpha) \sinh(w_1^* \alpha h_2) + 2\gamma_{10}^* \lambda_2 C^*(\alpha) \sinh(w_3^* \alpha h_2) = R_4(\alpha)
\end{aligned} \tag{4.14}$$

$R_i(\alpha)$  and  $\lambda_j$  are defined in Appendices B and A respectively.  
Solving (4.14) we obtain:

$$\begin{aligned}
A(\alpha) &= \frac{1}{2\cosh(w_1 \alpha h_1)} \left[ \frac{R_1(\alpha)}{f(\alpha)} g_2(\alpha) + \frac{R_2(\alpha)}{f(\alpha)} h_2(\alpha) \right. \\
&\quad \left. + \frac{R_3(\alpha)}{f(\alpha)} m_2(\alpha) + \frac{R_4(\alpha)}{f(\alpha)} n_2(\alpha) \right] \\
C(\alpha) &= \frac{1}{2\cosh(w_3 \alpha h_1)} \left[ \frac{R_1(\alpha)}{f(\alpha)} g_1(\alpha) + \frac{R_2(\alpha)}{f(\alpha)} h_1(\alpha) \right. \\
&\quad \left. + \frac{R_3(\alpha)}{f(\alpha)} m_1(\alpha) + \frac{R_4(\alpha)}{f(\alpha)} n_1(\alpha) \right] \\
A^*(\alpha) &= \frac{1}{2\cosh(w_1^* \alpha h_2)} \left[ \frac{R_1(\alpha)}{f(\alpha)} g_0(\alpha) + \frac{R_2(\alpha)}{f(\alpha)} h_0(\alpha) \right. \\
&\quad \left. + \frac{R_3(\alpha)}{f(\alpha)} m_0(\alpha) + \frac{R_4(\alpha)}{f(\alpha)} n_0(\alpha) \right] \\
C^*(\alpha) &= \frac{1}{2\cosh(w_3^* \alpha h_2)} \left[ \frac{R_1(\alpha)}{f(\alpha)} g(\alpha) + \frac{R_2(\alpha)}{f(\alpha)} h(\alpha) \right. \\
&\quad \left. + \frac{R_3(\alpha)}{f(\alpha)} m(\alpha) + \frac{R_4(\alpha)}{f(\alpha)} n(\alpha) \right]
\end{aligned} \tag{4.15}$$

The functions  $f(\alpha)$ ,  $g(\alpha)$  etc., used above are given in Appendix B.

Substituting (4.15) into (4.11) and (4.13) we obtain the following system of singular integral equations:

$$\frac{1}{\pi} \int_{-a}^a \frac{\phi(t)}{t-x_1} dt + \int_{-a}^a k_{11}(x_1, t)\phi(t)dt \\ + \int_{-b}^b k_{12}(x_1, t)\phi^*(t)dt = - \frac{(1-v_{xy}v_{yx})}{2Y_{14}E_y} p_1(x_1)$$

$$-a < x_1 < a$$

$$\frac{1}{\pi} \int_{-b}^b \frac{\phi^*(t)}{t-x_2} dt + \int_{-a}^a k_{21}(x_2, t)\phi(t)dt \\ + \int_{-b}^b k_{22}(x_2, t)\phi^*(t)dt = - \frac{(1-v_{xy}^*v_{yx}^*)}{2Y_{14}^*E_y^*} p_2(x_2)$$

$$-b < x_2 < b$$

(4.16a,b)

where

$$k_{11}(x_1, t) = \frac{1}{\pi Y_{14}} \int_0^\infty \left[ k_1(x_1, \alpha) e^{-(h_1-t)\alpha\sqrt{\beta_5}/|w_1|} \right. \\ \left. + k_2(x_1, \alpha) e^{-(h_1-t)\alpha\sqrt{\beta_5}/|w_3|} \right] d\alpha$$

$$k_{12}(x_1, t) = \frac{1}{\pi Y_{14}} \int_0^\infty \left[ k_3(x_1, \alpha) e^{-(h_2-t)\alpha\sqrt{\beta_5^*}/|w_1^*|} \right. \\ \left. + k_4(x_1, \alpha) e^{-(h_2-t)\alpha\sqrt{\beta_5^*}/|w_3^*|} \right] d\alpha$$

$$\begin{aligned}
k_{21}(x_2, t) &= \frac{1}{\pi r^*} \frac{1}{14} \int_0^\infty \left[ k_5(x_2, \alpha) e^{-(h_1-t)\alpha/\beta_5^*/|w_1|} \right. \\
&\quad \left. + k_6(x_2, \alpha) e^{-(h_1-t)\alpha/\beta_5^*/|w_3|} \right] d\alpha \\
k_{22}(x_2, t) &= \frac{1}{\pi r^*} \frac{1}{14} \int_0^\infty \left[ k_7(x_2, \alpha) e^{-(h_2-t)\alpha/\beta_5^*/|w_1|} \right. \\
&\quad \left. + k_8(x_2, \alpha) e^{-(h_2-t)\alpha/\beta_5^*/|w_3|} \right] d\alpha
\end{aligned} \tag{4.17}$$

The expressions  $k_j$  ( $j = 1, 8$ ) are given in Appendix B.

By letting  $h_2 \rightarrow \infty$  (or  $h_1 \rightarrow \infty$ ) one can recover the special case studied in [5]. For  $a < h_1$  and  $b < h_2$ , the integrands of kernels  $k_{ij}$  ( $i, j = 1, 2$ ) vanish when  $\alpha \rightarrow \infty$  and are bounded for all values of  $\alpha$ , except when  $\alpha = 0$ .

Around  $\alpha = 0$  the asymptotic behavior of the integrands  $I_{ij}$  of the kernels  $k_{ij}$  is of the following form:

$$I_{ij}(\alpha) = \frac{c_{ij}}{\alpha} + O(1) \quad (i, j = 1, 2) \tag{4.18}$$

where the  $c_{ij}$ 's are known constants. In order to obtain a solution, one should show that the singularity due to  $1/\alpha$  is removable. Consider the following integral:

$$\begin{aligned}
\int_{-a}^a \phi(t) dt \int_0^\infty I_{11}(x_1, t, \alpha) d\alpha &= \int_{-a}^a \phi(t) dt \left[ \int_0^\epsilon I_{11}(x_1, t, \alpha) d\alpha \right. \\
&\quad \left. + \int_\epsilon^\infty I_{11}(x_1, t, \alpha) d\alpha \right]
\end{aligned}$$

where  $\epsilon$  is a positive small number. Using (4.18) for the first part of the integral, we obtain:

$$\int_{-a}^a k_{11}(x_1, t)\phi(t)dt = \int_{-a}^a \phi(t)dt \left[ \int_0^\epsilon \frac{c_{11}}{\alpha} d\alpha + \int_0^\epsilon O(1)d\alpha + \int_\epsilon^\infty I_{11}(x_1, t, \alpha)d\alpha \right].$$

Making use of the single-valuedness condition

$\int_{-a}^a \phi(t)dt = 0$ , the unbounded integral  $\int_0^\epsilon \frac{c_{11}}{\alpha} d\alpha$  drops out, leaving only bounded integrals which can be evaluated numerically. Using  $\int_{-b}^b \phi^*(t)dt = 0$  for the second crack, similarly one can show that the singularity due to  $1/\alpha$  cancels in all the integrands  $I_{ij}$ .

## 5. CASE OF BROKEN LAMINATES

This is the case when one of the cracks touches the interfaces (i.e.,  $a = h_1$ , or  $b = h_2$ ). The integral equations (4.16a,b) are still valid but some of the kernels  $k_{ij}$  are no longer bounded. For example for  $a = h_1$ ,  $k_{12}$ ,  $k_{21}$ ,  $k_{22}$  are bounded but  $k_{11}$  becomes unbounded as  $x_1$  and  $t$  approach the ends  $\pm h_1$  simultaneously. In this case the integrand  $I_{11}$  of  $k_{11}$  diverges as  $\alpha \rightarrow \infty$ .

In order to obtain the proper singularity at the crack tips and to compute  $k_{11}$  numerically, the singular part,  $k_{11s}$ , should be evaluated in closed form. In this case, the kernel  $k_{11}$  can be written as:

$$k_{11}(x_1, t) = k_{11s}(x_1, t) + k_{11f}(x_1, t)$$

where  $k_{11s}$  is the singular part and  $k_{11f}$  is the bounded part of  $k_{11}$ .

Following the procedure described in [4]  $k_{11s}(x_1, t)$  is obtained as follows:

$$\begin{aligned} \pi k_{11s}(x_1, t) &= \lambda_{85} \left\{ \frac{(h_1-t)\sqrt{\beta_5}/|w_1| + |w_1|h_1}{[(h_1-t)\sqrt{\beta_5}/|w_1| + |w_1|h_1]^2 - (w_1 x_1)^2} \right\} \\ &+ \lambda_{86} \left\{ \frac{(h_1-t)\sqrt{\beta_5}/|w_3| + |w_3|h_1}{[(h_1-t)\sqrt{\beta_5}/|w_1| + |w_3|h_1]^2 - (w_3 x_1)^2} \right\} \\ &+ \lambda_{87} \left\{ \frac{(h_1-t)\sqrt{\beta_5}/|w_3| + |w_1|h_1}{[(h_1-t)\sqrt{\beta_5}/|w_3| + |w_1|h_1]^2 - (w_1 x_1)^2} \right\} \end{aligned}$$

$$+ \lambda_{88} \left\{ \frac{(h_1-t)\sqrt{\beta_5}/|w_3| + |w_3|h_1}{[(h_1-t)\sqrt{\beta_5}/|w_3| + |w_3|h_1]^2 - (w_3x_1)^2} \right\}$$

$$- h_1 \leq x_1, t \leq h_1 \quad (5.1)$$

The governing singular integral equations become:

$$\frac{1}{\pi} \int_{-h_1}^{h_1} \left[ \frac{1}{t-x_1} + \pi k_{11s}(x_1, t) \right] \phi(t) dt + \int_{-h_1}^{h_1} [k_{11}(x_1, t) - k_{11s}(x_1, t)] \phi(t) dt \\ + \int_{-b}^b k_{12}(x_1, t) \phi^*(t) dt = - \frac{(1-v_{xy}v_{yx})}{2\gamma_{14}^* E_y} p_1(x_1) \quad - h_1 < x_1 < h_1$$

$$\frac{1}{\pi} \int_{-b}^b \frac{\phi^*(t)}{t-x_2} dt + \int_{-h_1}^{h_1} k_{21}(x_2, t) \phi(t) dt + \int_{-b}^b k_{22}(x_2, t) \phi^*(t) dt = \\ - \frac{(1-v_{xy}^* v_{yx}^*)}{2\gamma_{14}^* E_y} p_2(x_2) \quad - b < x_2 < b \quad (5.2a, b)$$

Since in the integral equation (5.2b) the only singular term is  $\frac{1}{t-x_2}$ , the power of singularity at the end of the internal crack in the second layer is still 1/2. But in (5.2a) we have further singular contribution from the kernel  $k_{11s}$  resulting in a power different than 1/2. To find this singularity power  $\gamma$ , we will again use the procedure described in reference [14]. Throwing all the bounded terms to the right hand side, the singular integral equation (5.2a) can be written as follows:

$$\frac{1}{\pi} \int_{-h_1}^{h_1} \left[ \frac{1}{t-x_1} + \pi k_{11s}(x_1, t) \right] \phi(t) dt = P(x_1) \quad - h_1 < x_1 < h_1 \quad (5.3)$$

where  $P(x_1)$  is a bounded function for all values of  $x_1$ .

The unknown function  $\phi(t)$  can be written as (see [14]):

$$\phi(t) = \frac{F(t)}{(h_1^2 - t^2)^\gamma} \quad (5.4)$$

where  $F(t)$  is bounded and Hölder-continuous in the interval  $|t| < h_1$ , and  $0 < \operatorname{Re}(\gamma) < 1$ .

Define the sectionally holomorphic function:

$$\psi(z) = \frac{1}{\pi} \int_{-h_1}^{h_1} \frac{\phi(t)}{t-z} dt = \frac{1}{\pi} \int_{-h_1}^{h_1} \frac{F(t)e^{i\pi\gamma}}{(t-h_1)^\gamma(t+h_1)^\gamma(t-z)} dt$$

Then,

$$\psi(z) = \frac{F(-h_1)e^{i\pi\gamma}}{(2h_1)^\gamma \sin\pi\gamma} \frac{1}{(z+h_1)^\gamma} - \frac{F(h_1)}{(2h_1)^\gamma \sin\pi\gamma} \frac{1}{(z-h_1)^\gamma} + \psi_0(z) \quad (5.5)$$

where

$$|\psi_0(z)| < \frac{C}{|z \pm h_1|^{\gamma_0}}, \quad \gamma_0 < \operatorname{Re}(\gamma)$$

$C$  and  $\gamma_0$  are real constants.

Using (5.5), equation (5.3) takes the form:

$$\frac{F(-h_1)\cot\pi\gamma}{(2h_1)^\gamma(h_1+x_1)^\gamma} - \frac{F(h_1)\cot\pi\gamma}{(2h_1)^\gamma(h_1-x_1)^\gamma} + \frac{\lambda_{85}}{2} \frac{|w_1|}{\sqrt{\beta_s}} \frac{F(h_1)}{(2h_1)^\gamma \sin\pi\gamma \left(\frac{|w_1|}{\sqrt{\beta_s}}\right)^\gamma}$$

$$\begin{aligned}
& \left[ \frac{1}{(h_1+x_1)^Y} + \frac{1}{(h_1-x_1)^Y} \right] + \frac{\lambda_{86} |w_1|}{2 \sqrt{\beta_s}} \frac{F(h_1)}{(2h_1)^Y \sin \pi Y} \left[ \frac{|w_1| |w_3|}{\sqrt{\beta_s}} \right]^Y \\
& \quad \left[ \frac{1}{(h_1+x_1)^Y} + \frac{1}{(h_1-x_1)^Y} \right] \\
& + \frac{\lambda_{87} |w_3|}{2 \sqrt{\beta_s}} \frac{F(h_1)}{(2h_1)^Y \sin \pi Y} \left[ \frac{|w_1| |w_3|}{\sqrt{\beta_s}} \right]^Y \left[ \frac{1}{(h_1+x_1)^Y} + \frac{1}{(h_1-x_1)^Y} \right] \\
& + \frac{\lambda_{88} |w_3|}{2 \sqrt{\beta_s}} \frac{F(h_1)}{(2h_1)^Y \sin \pi Y} \left[ \frac{w_3^2}{\sqrt{\beta_s}} \right]^Y \left[ \frac{1}{(h_1+x_1)^Y} + \frac{1}{(h_1-x_1)^Y} \right] = P_1(x_1) \tag{5.6}
\end{aligned}$$

where because of symmetry  $F(h_1) = -F(-h_1)$ .

Multiplying both sides of (5.6) by  $(h_1+x_1)^Y$  and letting  $x_1 = -h_1$  we obtain the following characteristic equation:

$$\begin{aligned}
& -2 \cos \pi Y + \lambda_{85} \frac{|w_1|}{\sqrt{\beta_s}} \frac{1}{\left[ \frac{w_1^2}{\sqrt{\beta_s}} \right]^Y} + \lambda_{86} \frac{|w_1|}{\sqrt{\beta_s}} \frac{1}{\left[ \frac{|w_1| |w_3|}{\sqrt{\beta_s}} \right]^Y} \\
& + \lambda_{87} \frac{|w_3|}{\sqrt{\beta_s}} \frac{1}{\left[ \frac{|w_1| |w_3|}{\sqrt{\beta_s}} \right]^Y} + \lambda_{88} \frac{|w_3|}{\sqrt{\beta_s}} \frac{1}{\left[ \frac{w_3^2}{\sqrt{\beta_s}} \right]^Y} = 0 \tag{5.7}
\end{aligned}$$

where  $\lambda_j$ 's are elastic constants defined in Appendix A.

This is the same equation found in [7]. Choosing the orthotropic elastic constants close to isotropic constants numerically we find the same singularity power computed in [6] and [9]. The characteristic equation (5.7) can be solved numerically to find  $\gamma$ . For practical orthotropic materials equation (5.7) has only one root between 0 and 1. To establish the dependence of  $\gamma$  on the material constants more accurately, a separate study of equation (5.7) is needed.

## 6. CASE OF A CRACK CROSSING THE INTERFACE

To formulate this problem we will start by using the crack configuration shown in Figure 3. In this case we have again an internal crack in the first layer, but two symmetrically located cracks in the second layer. Using the symmetry property of  $\phi^*(t)$ , we can write:

$$\int_{-b}^b k_{i2}(x_i, t) \phi^*(t) dt = \int_0^b [k_{i2}(x_i, t) - k_{i2}(x_i, -t)] \phi^*(t) dt \quad (i = 1, 2)$$

and

$$\int_{-b}^b \frac{\phi^*(t)}{t-x_2} dt = \int_0^b \left[ \frac{1}{t-x_2} + \frac{1}{t+x_2} \right] \phi^*(t) dt \quad (6.1)$$

Therefore we can write the governing singular integral equations, by simply changing the limits of the integrals from  $(0, b)$  to  $(c, d)$  in equations (4.16a,b). Thus, we obtain:

$$\begin{aligned} & \frac{1}{\pi} \int_{-a}^a \frac{\phi(t)}{t-x_1} dt + \int_{-a}^a k_{11}(x_1, t) \phi(t) dt \\ & + \int_c^d [k_{12}(x_1, t) - k_{12}(x_1, -t)] \phi^*(t) dt \\ & = - \frac{(1-v_{xy})v_{yx}}{2\gamma_{14}E_y} p_1(x_1) \quad -a < x_1 < a \\ & \frac{1}{\pi} \int_c^d \left[ \frac{1}{t-x_2} + \frac{1}{t+x_2} \right] \phi^*(t) dt + \int_{-a}^a k_{21}(x_2, t) \phi(t) dt \end{aligned}$$

$$+ \int_c^d [k_{22}(x_2, t) - k_{22}(x_2, -t)] \phi^*(t) dt = - \frac{(1 - v_{xy}^* v_{yx}^*)}{2\gamma_{14}^* E_y^*} p_2(x_2)$$

$$c < x_2 < d \quad (6.2a, b)$$

By letting  $a = h_1$  and  $d = h_2$  we obtain the case of a crack crossing the interface. As in the previous case for  $a = h_1$  and  $d = h_2$  all the kernels  $k_{ij}$  become unbounded when  $x_i, t$  approach the ends  $\pm h_1$  and  $x_2, t$  approach the end  $h_2$  simultaneously. Therefore to study the singular behavior at the interface and to make the kernels numerically integrable, the singular parts of the kernels  $k_{ij}$  must be separated.

The kernels can be written as:

$$k_{ij}(x_i, t) = k_{ijs}(x_i, t) + k_{ijf}(x_i, t)$$

where  $k_{ijs}(x_i, t)$  is the singular and  $k_{ijf}(x_i, t)$  is the bounded part. Following the same procedure used in the previous section the expressions of  $k_{ijs}$  are found as follows:

$$\begin{aligned} \pi k_{11s}(x_1, t) = & \lambda_{85} \left\{ \frac{(h_1 - t)\sqrt{\beta_5}/|w_1| + |w_1|h_1}{[(h_1 - t)\sqrt{\beta_5}/|w_1| + |w_1|h_1]^2 - (w_1 x_1)^2} \right\} \\ & + \lambda_{86} \left\{ \frac{(h_1 - t)\sqrt{\beta_5}/|w_1| + |w_3|h_1}{[(h_1 - t)\sqrt{\beta_5}/|w_1| + |w_3|h_1]^2 - (w_3 x_1)^2} \right\} \\ & + \lambda_{87} \left\{ \frac{(h_1 - t)\sqrt{\beta_5}/|w_3| + |w_1|h_1}{[(h_1 - t)\sqrt{\beta_5}/|w_3| + |w_1|h_1]^2 - (w_1 x_1)^2} \right\} \\ & + \lambda_{88} \left\{ \frac{(h_1 - t)\sqrt{\beta_5}/|w_3| + |w_3|h_1}{[(h_1 - t)\sqrt{\beta_5}/|w_3| + |w_3|h_1]^2 - (w_3 x_1)^2} \right\} \end{aligned}$$

$$\pi k_{12s}(x_1, t) = \lambda_{93} \left\{ \frac{(h_2-t)\sqrt{\beta_s^*}/|w_1^*| + |w_1|h_1}{[(h_2-t)\sqrt{\beta_s^*}/|w_1^*| + |w_1|h_1]^2 - (w_1 x_1)^2} \right\}$$

$$+ \lambda_{94} \left\{ \frac{(h_2-t)\sqrt{\beta_s^*}/|w_1^*| + |w_3|h_1}{[(h_2-t)\sqrt{\beta_s^*}/|w_1^*| + |w_3|h_1]^2 - (w_3 x_1)^2} \right\}$$

$$+ \lambda_{95} \left\{ \frac{(h_2-t)\sqrt{\beta_s^*}/|w_3^*| + |w_1|h_1}{[(h_2-t)\sqrt{\beta_s^*}/|w_3^*| + |w_1|h_1]^2 - (w_1 x_1)^2} \right\}$$

$$+ \lambda_{96} \left\{ \frac{(h_2-t)\sqrt{\beta_s^*}/|w_3^*| + |w_3|h_1}{[(h_2-t)\sqrt{\beta_s^*}/|w_3^*| + |w_3|h_1]^2 - (w_3 x_1)^2} \right\}$$

$$\pi k_{21s}(x_2, t) = \lambda_{101} \left\{ \frac{(h_1-t)\sqrt{\beta_s}/|w_1| + |w_1^*|h_2}{[(h_1-t)\sqrt{\beta_s}/|w_1| + |w_1^*|h_2]^2 - (w_1^* x_2)^2} \right\}$$

$$+ \lambda_{102} \left\{ \frac{(h_1-t)\sqrt{\beta_s}/|w_1| + |w_3^*|h_2}{[(h_1-t)\sqrt{\beta_s}/|w_1| + |w_3^*|h_2]^2 - (w_3^* x_2)^2} \right\}$$

$$+ \lambda_{103} \left\{ \frac{(h_1-t)\sqrt{\beta_s}/|w_3| + |w_1^*|h_2}{[(h_1-t)\sqrt{\beta_s}/|w_3| + |w_1^*|h_2]^2 - (w_1^* x_2)^2} \right\}$$

$$+ \lambda_{104} \left\{ \frac{(h_1-t)\sqrt{\beta_s}/|w_3| + |w_3^*|h_2}{[(h_1-t)\sqrt{\beta_s}/|w_3| + |w_3^*|h_2]^2 - (w_3^* x_2)^2} \right\}$$

$$\pi k_{22s}(x_2, t) = \lambda_{109} \left\{ \frac{(h_2-t)\sqrt{\beta_s^*}/|w_1^*| + |w_1^*|h_2}{[(h_2-t)\sqrt{\beta_s^*}/|w_1^*| + |w_1^*|h_2]^2 - (w_1^* x_2)^2} \right\}$$

$$+ \lambda_{110} \left\{ \frac{(h_2-t)\sqrt{\beta_s^*}/|w_1^*| + |w_3^*|h_2}{[(h_2-t)\sqrt{\beta_s^*}/|w_1^*| + |w_3^*|h_2]^2 - (w_3^* x_2)^2} \right\}$$

$$+ \lambda_{111} \left\{ \frac{(h_2-t)\sqrt{\beta_s^*}/|w_3^*| + |w_1^*|h_2}{[(h_2-t)\sqrt{\beta_s^*}/|w_3^*| + |w_1^*|h_2]^2 - (w_1^* x_2)^2} \right\}$$

$$+ \lambda_{112} \left\{ \frac{(h_2-t)\sqrt{\beta_s^*}/|w_3^*| + |w_3^*|h_2}{[(h_2-t)\sqrt{\beta_s^*}/|w_3^*| + |w_3^*|h_2]^2 - (w_3^* x_2)^2} \right\}$$

Separating the singular parts, equations (6.2a,b) take the form:

$$\begin{aligned}
 & \frac{1}{\pi} \int_{-h_1}^{h_1} \left[ \frac{1}{t-x_1} + \pi k_{11s}(x_1, t) \right] \phi(t) dt \\
 & + \int_{-h_1}^{h_1} [k_{11}(x_1, t) - k_{11s}(x_1, t)] \phi(t) dt \\
 & + \int_c^{h_2} k_{12s}^*(x_1, t) \phi^*(t) dt \\
 & + \int_c^{h_2} [k_{12}^*(x_1, t) - k_{12s}^*(x_1, t)] \phi^*(t) dt \\
 & = - \frac{(1-v_{xy}^* v_{yx}^*)}{2\gamma_{14}^* E_y} p_1(x_1) \quad -h_1 < x_1 < h_1
 \end{aligned}$$

and

$$\begin{aligned}
 & \frac{1}{\pi} \int_c^{h_2} \left[ \frac{1}{t-x_2} + \frac{1}{t+x_2} + \pi k_{22s}^*(x_2, t) \right] \phi^*(t) dt \\
 & + \int_{-h_1}^{h_1} [k_{21}(x_2, t) - k_{21s}(x_2, t)] \phi(t) dt \\
 & + \int_{-h_1}^{h_1} k_{21s}(x_2, t) \phi(t) dt \\
 & + \int_c^{h_2} [k_{22}^*(x_2, t) - k_{22s}^*(x_2, t)] \phi(t) dt \\
 & = - \frac{(1-v_{xy}^* v_{yx}^*)}{2\gamma_{14}^* E_y} p_2(x_2) \quad c < x_2 < h_2
 \end{aligned}$$

(6.3a,b)

where

$$k_{12}^*(x_i, t) = k_{12}(x_i, t) - k_{12}(x_i, -t) \quad (i=1,2).$$

To find the proper singularity power  $\beta$  at the interface, we will first throw all the bounded terms to the right hand sides of the equations as it has been done in the previous case. Then, we obtain the following system of equations:

$$\begin{aligned} & \frac{1}{\pi} \int_{-h_1}^{h_1} \left[ \frac{1}{t-x_1} + \pi k_{11s}(x_1, t) \right] \phi(t) dt \\ & + \int_c^{h_2} k_{12s}^*(x_1, t) \phi^*(t) dt = Q_1(x_1) \\ & -h_1 < x_1 < h_1 \\ & \frac{1}{\pi} \int_c^{h_2} \left[ \frac{1}{t-x_2} + \frac{1}{t+x_2} + \pi k_{22s}^*(x_2, t) \right] \phi^*(t) dt \\ & + \int_{-h_1}^{h_1} [k_{21s}(x_2, t) \phi(t) dt] = Q_2(x_2) \\ & c < x_2 < h_2 \end{aligned} \quad (6.4a,b)$$

where  $Q_1(x_1)$  and  $Q_2(x_2)$  are bounded functions of  $x_1$  and  $x_2$ . Considering the behavior of  $\phi(t)$  and  $\phi^*(t)$  at the end points, we can write:

$$\phi(t) = \frac{F(t)}{(h_1^2 - t^2)^\beta}, \quad \phi^*(t) = \frac{F^*(t)}{(h_2 - t)^\beta (t - c)^\delta} \quad (6.5a,b)$$

Define the following sectionally holomorphic functions:

$$\psi(z) = \frac{1}{\pi} \int_{-h_1}^{h_1} \frac{\phi(t)}{t-z} dt, \quad \psi^*(z) = \frac{1}{\pi} \int_c^{h_2} \frac{\phi^*(t)}{t-z} dt \quad (6.6a,b)$$

From [14] and using (6.5a,b) and (6.6a,b) we have:

$$\begin{aligned}\psi(z) &= \frac{F(-h_1)e^{i\pi\beta}}{(2h_1)^\beta \sin\beta} \frac{1}{(z+h_1)^\beta} - \frac{F(h_1)}{(2h_1)^\beta \sin\beta} \frac{1}{(z-h_1)^\beta} + \psi_0(z) \\ \psi^*(z) &= \frac{F^*(c)e^{i\pi\delta}}{(h_2-c)^\delta \sin\delta} \frac{1}{(z-c)^\delta} - \frac{F^*(h_2)}{(h_2-c)^\delta \sin\beta} \frac{1}{(z-h_2)^\beta} + \psi_0^*(z)\end{aligned}\quad (6.7a,b)$$

where  $\psi_0(z)$  and  $\psi_0^*(z)$  are bounded functions which around ends behave as follows:

$$|\psi_0(z)| \leq \frac{C_0}{|z+h_1|^{\beta_0}}, \quad \beta_0 < \operatorname{Re}(\beta)$$

and

$$|\psi_0^*(z)| \left\{ \begin{array}{l} \leq \frac{D_0}{|z-h_2|^{\beta_0^*}}, \quad \beta_0^* < \operatorname{Re}(\beta) \\ \text{and} \\ \leq \frac{E_0}{|z-c|^{\delta_0}}, \quad \delta_0 < \operatorname{Re}(\delta) \end{array} \right.$$

$C_0, D_0, E_0, \beta_0, \beta_0^*, \delta_0$  are real constants.

Using (6.7a,b) and following the procedure used in section 5, equations (6.4a,b) reduce to:

$$\cot\pi\delta = 0 \quad (6.8)$$

and

$$\frac{F(h_1)}{2(2h_1)^\beta \sin \pi \beta} \left\{ -2 \cos \pi \beta + \lambda_{85} \frac{|w_1|}{\sqrt{\beta_s}} \left[ \frac{1}{\frac{|w_1|^2}{\sqrt{\beta_s}}} \right]^\beta + \lambda_{86} \frac{|w_1|}{\sqrt{\beta_s}} \left[ \frac{1}{|w_1| |w_3|} \right]^\beta \right.$$

$$\left. + \lambda_{87} \frac{|w_3|}{\sqrt{\beta_s}} \left[ \frac{1}{|w_1| |w_3|} \right]^\beta + \lambda_{88} \frac{|w_3|}{\sqrt{\beta_s}} \left[ \frac{1}{\frac{|w_3|^2}{\sqrt{\beta_s}}} \right]^\beta \right\}$$

$$+ \frac{F^*(h_2)}{2(h_2 - c)^\delta \sin \pi \beta} \left\{ \lambda_{93} \frac{|w_1^*|}{\sqrt{\beta_s^*}} \left[ \frac{1}{\frac{|w_1| |w_1^*|}{\sqrt{\beta_s^*}}} \right]^\beta + \lambda_{94} \frac{|w_1^*|}{\sqrt{\beta_s^*}} \left[ \frac{1}{|w_3| |w_1^*|} \right]^\beta \right.$$

$$\left. + \lambda_{95} \frac{|w_3^*|}{\sqrt{\beta_s^*}} \left[ \frac{1}{|w_1| |w_3^*|} \right]^\beta + \lambda_{96} \frac{|w_3^*|}{\sqrt{\beta_s^*}} \left[ \frac{1}{\frac{|w_3|^2}{\sqrt{\beta_s^*}}} \right]^\beta \right\} = 0$$

$$\frac{F(h_1)}{2(2h_1)^\beta \sin \pi \beta} \left\{ \lambda_{101} \frac{|w_1|}{\sqrt{\beta_s}} \left[ \frac{1}{\frac{|w_1^*| |w_1|}{\sqrt{\beta_s}}} \right]^\beta + \lambda_{102} \frac{|w_1|}{\sqrt{\beta_s}} \left[ \frac{1}{\frac{|w_3^*| |w_1|}{\sqrt{\beta_s}}} \right]^\beta \right.$$

$$\left. + \lambda_{103} \frac{|w_3|}{\sqrt{\beta_s}} \left[ \frac{1}{\frac{|w_1^*| |w_3|}{\sqrt{\beta_s}}} \right]^\beta + \lambda_{104} \frac{|w_3|}{\sqrt{\beta_s}} \left[ \frac{1}{\frac{|w_3^*| |w_3|}{\sqrt{\beta_s}}} \right]^\beta \right\}$$

$$+ \frac{F^*(h_2)}{2(h_2 - c)^\delta \sin \pi \beta} \left\{ -2 \cos \pi \beta + \lambda_{109} \frac{|w_1^*|}{\sqrt{\beta_s^*}} \left[ \frac{1}{\frac{|w_1|^2}{\sqrt{\beta_s^*}}} \right]^\beta + \lambda_{110} \frac{|w_1^*|}{\sqrt{\beta_s^*}} \left[ \frac{1}{|w_3^*| |w_1^*|} \right]^\beta \right\}$$

$$+ \lambda_{111} \frac{|w_1^*|}{\sqrt{\beta_s^*}} \left[ \frac{1}{\frac{|w_1^*| |w_3^*|}{\sqrt{\beta_s^*}}} \right]^\beta + \lambda_{112} \frac{|w_3^*|}{\sqrt{\beta_s^*}} \left[ \frac{1}{\frac{|w_3|^2}{\sqrt{\beta_s^*}}} \right]^\beta \} = 0$$

(6.9a,b)

Equation (6.8) gives the expected  $\delta = 1/2$  singularity power at the crack tip.

(6.9a,b) is a system of homogeneous linear equations for  $F(h_1)$  and  $F^*(h_2)$ . Since  $F(h_1) \neq 0$ ,  $F^*(h_2) \neq 0$ ,  $\beta \neq 0, 1$  to solve the system one should equate the determinant of coefficients to zero. Thus,

$$\Delta(\beta) = 4\cos^2\pi\beta - 2\cos\pi\beta \left\{ \lambda_{109} \frac{|w_1^*|}{\sqrt{\beta_s^*}} \frac{1}{(r_{11}^*)^\beta} + \left[ \lambda_{110} \frac{|w_1^*|}{\sqrt{\beta_s^*}} \right. \right. \\ \left. \left. + \lambda_{111} \frac{|w_3^*|}{\sqrt{\beta_s^*}} \right] \frac{1}{(r_{13}^*)^\beta} + \lambda_{112} \frac{|w_3^*|}{\sqrt{\beta_s^*}} \frac{1}{(r_{33}^*)^\beta} \right. \\ \left. + \lambda_{85} \frac{|w_1|}{\sqrt{\beta_s}} \frac{1}{(r_{11})^\beta} \right. \\ \left. + \left[ \lambda_{86} \frac{|w_1|}{\sqrt{\beta_s}} + \lambda_{87} \frac{|w_3|}{\sqrt{\beta_s}} \right] \frac{1}{(r_{13})^\beta} \right. \\ \left. + \lambda_{88} \frac{|w_3|}{\sqrt{\beta_s}} \frac{1}{(r_{33})^\beta} \right\}$$

$$\begin{aligned}
& + \lambda_{113} \frac{1}{(r_{11} r_{11}^*)^\beta} + \lambda_{114} \frac{1}{(r_{11} r_{13})^\beta} \\
& + \lambda_{115} \frac{1}{(r_{11} r_{33})^\beta} + \lambda_{116} \frac{1}{(r_{13} r_{11})^\beta} \\
& + \lambda_{117} \frac{1}{(r_{13} r_{13})^\beta} + \lambda_{118} \frac{1}{(r_{13} r_{33})^\beta} + \lambda_{119} \frac{1}{(r_{33} r_{11})^\beta} \\
& + \lambda_{120} \frac{1}{(r_{33} r_{13})^\beta} + \lambda_{121} \frac{1}{(r_{33} r_{33})^\beta} = 0 \tag{6.10}
\end{aligned}$$

where

$$\begin{aligned}
r_{11} &= \frac{w_1^2}{\sqrt{\beta_5}} , \quad r_{13} = \frac{|w_1||w_3|}{\sqrt{\beta_5}} , \quad r_{33} = \frac{w_3^2}{\sqrt{\beta_5}} \\
r_{11}^* &= \frac{w_1^{*2}}{\sqrt{\beta_5^*}} , \quad r_{13}^* = \frac{|w_1^*||w_3^*|}{\sqrt{\beta_5^*}} , \quad r_{33}^* = \frac{w_3^{*2}}{\sqrt{\beta_5^*}}
\end{aligned}$$

From the characteristic equation (6.10) we can determine the singularity power  $\beta$ . Choosing the orthotropic elastic constants close to isotropic elastic constants, we recover the singularity power found in [8] and [9]. Equation (6.10) does not always have a root between 0 and 1. For some orthotropic material combinations, there is no power singularity at the interface. In this case one should investigate the possibility of pure imaginary or complex roots. Numerical computation shows that there are no pure imaginary roots or complex roots for which the real part

is between 0 and 1. On the other hand  $F(h_1)$  and  $F^*(h_2)$  are related through (6.9a) or (6.9b). This is a condition to be used while obtaining the solution. The absence of power singularity for some orthotropic material combinations may be very important from the view point of design applications. Therefore we will study in some detail the behavior of the crack surface displacement derivatives and the stresses at the interface.

Let's first investigate the possibility of a weaker i.e., logarithmic singularity in the crack surface displacement derivatives at the interface. Suppose that the power singularity  $\beta$  at the interface is zero. Define:

$$\psi(z) = \frac{1}{\pi} \int_{-h_1}^{h_1} \frac{\phi(t)}{t-z} dt, \quad \psi^*(z) = \frac{1}{\pi} \int_c^{h_2} \frac{\phi^*(t)}{t-z} dt$$

The behavior of  $\psi(z)$  around  $z = \pm h_1$ , and of  $\psi^*(z)$  around  $z = h_2$  can be expressed as:

$$\psi^*(z) = \frac{\phi(h_1)}{\pi} \log(z-h_1) + \phi_{01}(z) \text{ near } z = h_1$$

$$\psi(z) = -\frac{\phi(-h_1)}{\pi} \log(z+h_1) + \phi_{02}(z) \text{ near } z = -h_1$$

$$\psi^*(z) = \frac{\phi^*(h_2)}{\pi} \log(z-h_2) + \phi_{03}(z) \text{ near } z = +h_2$$

(6.11a,b,c)

where  $\phi_{01}(z)$ ,  $\phi_{02}(z)$ ,  $\phi_{03}(z)$  are bounded functions.

Using (6.11a,b,c), near  $x_1 = h_1$  and  $x_2 = h_2$  equations

(6.4a,b) take the following form:

$$\begin{aligned} \log(h_1 - x_1) & \left\{ \frac{\phi(h_1)}{\pi} \left[ 1 - \lambda_{85} \frac{|w_1|}{2\sqrt{\beta_s}} - \lambda_{86} \frac{|w_1|}{2\sqrt{\beta_s}} - \lambda_{87} \frac{|w_3|}{2\sqrt{\beta_s}} - \lambda_{88} \frac{|w_3|}{2\sqrt{\beta_s}} \right] \right. \\ & \left. + \frac{\phi^*(h_2)}{\pi} \left[ \frac{|w_1^*|}{2\sqrt{\beta_s^*}} \lambda_{93} - \frac{|w_1^*|}{2\sqrt{\beta_s^*}} \lambda_{94} - \frac{|w_3^*|}{2\sqrt{\beta_s^*}} \lambda_{95} - \frac{|w_3^*|}{2\sqrt{\beta_s^*}} \lambda_{96} \right] \right\} = F_1(x_1) \\ \log(h_2 - x_2) & \left\{ \frac{\phi(h_1)}{\pi} \left[ -\frac{|w_1|}{2\sqrt{\beta_s}} \lambda_{101} - \frac{|w_1|}{2\sqrt{\beta_s}} \lambda_{102} - \frac{|w_3|}{2\sqrt{\beta_s}} \lambda_{103} - \frac{|w_3|}{2\sqrt{\beta_s}} \lambda_{104} \right] \right. \\ & \left. + \frac{\phi^*(h_2)}{\pi} \left[ 1 - \frac{|w_1^*|}{2\sqrt{\beta_s^*}} \lambda_{109} - \frac{|w_1^*|}{2\sqrt{\beta_s^*}} \lambda_{110} - \frac{|w_3^*|}{2\sqrt{\beta_s^*}} \lambda_{111} - \frac{|w_3^*|}{2\sqrt{\beta_s^*}} \lambda_{112} \right] \right\} = F_2(x_2) \end{aligned} \quad (6.12a,b)$$

where  $F_1(x_1)$  and  $F_2(x_2)$  are bounded functions.

In order that equations (6.12 a,b) be bounded for  $x_1=h_1$ ,  $x_2=h_2$ , the coefficients of the logarithmic terms must be zero.

Thus:

$$\begin{aligned} \frac{\phi(h_1)}{\pi} & \left[ 1 - \lambda_{85} \frac{|w_1|}{2\sqrt{\beta_s}} - \lambda_{86} \frac{|w_1|}{2\sqrt{\beta_s}} - \lambda_{87} \frac{|w_3|}{2\sqrt{\beta_s}} - \lambda_{88} \frac{|w_3|}{2\sqrt{\beta_s}} \right] \\ & + \frac{\phi^*(h_2)}{\pi} \left[ -\frac{|w_1^*|}{2\sqrt{\beta_s^*}} \lambda_{93} - \frac{|w_1^*|}{2\sqrt{\beta_s^*}} \lambda_{94} - \frac{|w_3^*|}{2\sqrt{\beta_s^*}} \lambda_{95} - \frac{|w_3^*|}{2\sqrt{\beta_s^*}} \lambda_{96} \right] = 0 \end{aligned}$$

and

$$\frac{\phi(h_1)}{\pi} \left[ -\frac{|w_1|}{2\sqrt{\beta_s}} \lambda_{101} - \frac{|w_1|}{2\sqrt{\beta_s}} \lambda_{102} - \frac{|w_3|}{2\sqrt{\beta_s}} \lambda_{103} - \frac{|w_3|}{2\sqrt{\beta_s}} \lambda_{104} \right] + \frac{\phi^*(h_2)}{\pi} \left[ 1 - \frac{|w_1^*|}{2\sqrt{\beta_s^*}} \lambda_{109} - \frac{|w_1^*|}{2\sqrt{\beta_s^*}} \lambda_{110} - \frac{|w_3^*|}{2\sqrt{\beta_s^*}} \lambda_{111} - \frac{|w_3^*|}{2\sqrt{\beta_s^*}} \lambda_{112} \right] = 0$$

(6.13a,b)

(6.13a,b) is a system of linear equations for  $\phi(h_1)$  and  $\phi^*(h_2)$ . Since  $\phi(h_1)$  and  $\phi^*(h_2)$  are different than zero, in order to have a solution the determinant of coefficients,  $\Delta$ , must be zero. Numerical computation shows that  $\Delta = 0$  and either from equation (6.13a) or (6.13b) we have:

$$\frac{\phi(h_1)}{\phi^*(h_2)} = -1$$

and using the symmetry condition  $\phi^*(h_2) = -\phi^*(-h_2)$ , we obtain:

$$\frac{\phi(h_1)}{\phi^*(-h_2)} = 1$$

(6.14)

Relation (6.14) shows that the surface displacement derivative is continuous at the interface. This is an important result which makes the solution of the singular integral equations easier.

To study the behavior of the stresses, let's first write their expressions at the interface. By making use of

(3.11) and (4.15) we obtain:

$$\sigma(y) = \frac{\pi(1-\nu_{xy}\nu_{yx})}{2E_x} \sigma_{1x}(h_1, y) = \int_{-h_1}^{h_1} K_1(n, y)\phi(n)dn + \int_{-C}^{h_2} K_2(n, y)\phi^*(n)dn$$

$$\tau(y) = \frac{\pi}{2G_{xy}} \tau_{1xy}(h_1, y) = \int_{-h_1}^{h_1} K_3(n, y)\phi(n)dn + \int_{-C}^{h_2} K_4(n, y)\phi^*(n)dn$$

(6.15a,b)

where  $K_j(n, y)$  ( $j = 1, 4$ ) are given in Appendix B.

The kernels  $K_j$  become unbounded as  $y \rightarrow 0^+$  and  $n \rightarrow \pm h_1$  or  $n \rightarrow h_2$  respectively. When equation (6.10) has a root, i.e. when the functions  $\phi(t)$  and  $\phi^*(t)$  are singular, the stresses have the same singularity power as we will see in the derivation of the stress intensity factors. But it is necessary to know whether the stresses have a logarithmic singularity when the crack surface displacement derivatives are bounded. To do that let's first separate the singular parts of kernels  $K_j$ . Following the usual procedure, we obtain:

$$K_{1s}(n, y) = \frac{1}{4\gamma_{13}} \left[ \frac{\gamma_1(h_1+n)}{\frac{w_1^2 y^2}{\beta_s} + (h_1+n)^2} - \gamma_1 \frac{(h_1-n)}{\frac{w_1^2 y^2}{\beta_s} + (h_1-n)^2} \right.$$

$$\left. - \gamma_2 \frac{\gamma_{11}}{\gamma_{12}} \frac{h_1+n}{\frac{w_3^2 y^2}{\beta_s} + (h_1+n)^2} + \gamma_2 \frac{\gamma_{11}}{\gamma_{12}} \frac{h_1-n}{\frac{w_3^2 y^2}{\beta_s} + (h_1-n)^2} \right]$$

$$+ \frac{1}{2\gamma_{13}^*\lambda_{80}} (\gamma_3\lambda_{81} + \gamma_4\lambda_{82}) \frac{(h_1 - \eta)\sqrt{\beta_5}/|w_1|}{[(h_1 - \eta)\sqrt{\beta_5}/|w_1|]^2 + y^2}$$

$$+ \frac{1}{2\gamma_{13}^*\lambda_{80}} (\gamma_3\lambda_{83} + \gamma_4\lambda_{84}) \frac{(h_1 - \eta)\sqrt{\beta_5}/|w_3|}{[(h_1 - \eta)\sqrt{\beta_5}/|w_3|]^2 + y^2}$$

$$K_{2s}(n, y) = \frac{1}{2\gamma_{13}^*\lambda_{80}} (\gamma_3\lambda_{89} + \gamma_4\lambda_{90}) \frac{(h_2 - \eta)\sqrt{\beta_5^*}/|w_1^*|}{[(h_2 - \eta)\sqrt{\beta_5^*}/|w_1^*|]^2 + y^2}$$

$$+ \frac{1}{2\gamma_{13}^*\lambda_{80}} (\gamma_3\lambda_{91} + \gamma_4\lambda_{92}) \frac{(h_2 - \eta)\sqrt{\beta_5^*}/|w_3^*|}{[(h_2 - \eta)\sqrt{\beta_5^*}/|w_3^*|]^2 + y^2}$$

$$K_{3s}(n, y) = \frac{\gamma_{11}}{4\gamma_{13}} \left[ -\frac{|w_1|y/\sqrt{\beta_5}}{\frac{w_1^2y^2}{\beta_5} + (h_1 + \eta)} - \frac{|w_1|y/\sqrt{\beta_5}}{\frac{w_1^2y^2}{\beta_5} + (h_1 - \eta)} \right]$$

$$+ \frac{|w_3|y/\sqrt{\beta_5}}{\frac{w_3^2y^2}{\beta_5} + (h_1 + \eta)} - \frac{|w_3|y/\sqrt{\beta_5}}{\frac{w_3^2y^2}{\beta_5} + (h_1 - \eta)}$$

$$+ \frac{1}{2\gamma_{13}^*\lambda_{80}} (\gamma_9\lambda_{81} + \gamma_{10}\lambda_{82}) \frac{y}{[(h_1 - \eta)\sqrt{\beta_5}/|w_1|]^2 + y^2}$$

$$+ \frac{1}{2\gamma_{13}^*\lambda_{80}} (\gamma_9\lambda_{83} + \gamma_{10}\lambda_{84}) \frac{y}{[(h_1 - \eta)\sqrt{\beta_5}/|w_3|]^2 + y^2}$$

$$K_{4s}(n, y) = \frac{1}{2\gamma_{13}^*\lambda_{80}} (\gamma_9\lambda_{89} + \gamma_{10}\lambda_{90}) \frac{y}{[(h_2 - \eta)\sqrt{\beta_5^*}/|w_1^*|]^2 + y^2}$$

$$+ \frac{1}{2\gamma_{13}^*\lambda_{80}} (\gamma_9\lambda_{91} + \gamma_{10}\lambda_{92}) \frac{y}{[(h_2 - \eta)\sqrt{\beta_5^*}/|w_3^*|]^2 + y^2}$$

(6.16a, b, c, d)

Keeping only the singular terms, equations (6.15a,b) can be written as follows:

$$\sigma(y) = \int_{-h_1}^{h_1} K_{1s}(n,y)\phi(n)dn + \int_c^{h_2} K_{2s}(n,y)\phi^*(n)dn + A(y)$$

$$\tau(y) = \int_{-h_1}^{h_1} K_{3s}(n,y)\phi(n)dn + \int_c^{h_2} K_{2s}(n,y)\phi^*(n)dn + B(y) \quad (6.17a,b)$$

where  $A(y)$  and  $B(y)$  are bounded functions.

Define:

$$\psi(z) = \int_{-h_1}^{h_1} \frac{\phi(t)}{t-z} dt \quad \text{and} \quad \psi^*(z) = \int_c^{h_2} \frac{\phi^*(t)}{t-z} dt$$

Considering their behavior around ends, we can write:

$$\psi(z) = \phi(h_1)\log(z-h_1) - \phi(-h_1)\log(z+h_1) + \psi_0(z)$$

and

$$\psi^*(z) = \phi^*(h_2)\log(z-h_2) + \psi_0^*(z) \quad (6.18a,b)$$

where  $\psi_0(z)$  and  $\psi_0^*(z)$  are bounded functions. Making use of (6.18a,b) and (6.14), equations (6.17a,b) become:

$$\begin{aligned} \sigma(y) = & \log y \phi(h_1) \left[ \frac{1}{2\gamma_{13}} (\gamma_1 - \gamma_2 \frac{\gamma_{11}}{\gamma_{12}}) - \frac{1}{2\gamma_{13}\lambda_{80}} \frac{|w_1|}{\sqrt{\beta_5}} (\gamma_3\lambda_{81} + \gamma_4\lambda_{82}) \right. \\ & - \frac{1}{2\gamma_{13}\lambda_{80}} \frac{|w_3|}{\sqrt{\beta_5}} (\gamma_3\lambda_{83} + \gamma_4\lambda_{84}) + \frac{1}{2\gamma_{13}^*\lambda_{80}} \frac{|w_1^*|}{\sqrt{\beta_5^*}} (\gamma_3\lambda_{89} + \gamma_4\lambda_{90}) \\ & \left. + \frac{1}{2\gamma_{13}^*\lambda_{80}} \frac{|w_3^*|}{\sqrt{\beta_5^*}} (\gamma_3\lambda_{91} + \gamma_4\lambda_{92}) \right] + C(y) \end{aligned}$$

and

$$\tau(y) = D(y) \quad (6.19a,b)$$

where  $C(y)$  and  $D(y)$  are bounded functions. Equation (6.19b) indicates that the shear stress at the interface is bounded. Since  $C(y)$  is bounded and  $\phi(h_1) \neq 0$ , if  $\sigma(y)$  is bounded the coefficient of  $\log y$  in equation (6.19a) should be zero. Numerical computation shows that the above mentioned coefficient is identically zero. Therefore the normal stress  $\sigma_x$  at the interface is also bounded. These are important results for orthotropic materials and may have practical implications in designing with composite materials.

## 7. THE SOLUTION AND THE RESULTS

Since we have mainly three different problems, the solution will be discussed in three sections.

### 7.1 Case of Internal Cracks

In this case we have to solve the system of singular equations (4.16a,b). Defining  $x_1 = a\kappa_1$ ,  $t = a\tau$  for  $-a < x_1, t < a$  and  $x_2 = a\kappa_2$ ,  $t = b\tau$  for  $-b < x_2, t < b$  after normalization, equations (4.16a,b) take the form:

$$\frac{1}{\pi} \int_{-1}^1 \frac{\phi_0(\tau)}{\tau - \kappa_1} d\tau + a \int_{-1}^1 k_{11}^0(\kappa_1, \tau) \phi_0(\tau) d\tau + b \int_{-1}^1 k_{12}^0(\kappa_1, \tau) \phi_0^*(\tau) d\tau$$

$$= -\frac{(1-v_{xy}v_{yx})}{2\gamma_{14}^* E_y} p_1^0(\kappa_1) \quad -1 < \kappa_1 < 1$$

$$\frac{1}{\pi} \int_{-1}^1 \frac{\phi_0^*(\tau)}{\tau - \kappa_2} d\tau + a \int_{-1}^1 k_{21}^0(\kappa_2, \tau) \phi_0(\tau) d\tau + b \int_{-1}^1 k_{22}^0(\kappa_2, \tau) \phi_0^*(\tau) d\tau$$

$$= -\frac{(1-v_{xy}^* v_{yx}^*)}{2\gamma_{14}^* E_y^*} p_2^0(\kappa_2) \quad -1 < \kappa_2 < 1 \quad (7.1a,b)$$

where the index "o" denotes the normalized quantities. To get the complete solution we need also the single-valuedness conditions:

$$\int_{-1}^1 \phi_0(\tau) d\tau = 0 \quad \text{and} \quad \int_{-1}^1 \phi_0^*(\tau) d\tau = 0 \quad (7.2a,b)$$

Since  $\phi_0(\tau)$  and  $\phi_0^*(\tau)$  have a power singularity  $-1/2$  at

the ends, the solution will be sought in the form:

$$\phi_0(\tau) = \frac{F_0(\tau)}{\sqrt{1-\tau^2}} \quad \text{and} \quad \phi_0^*(\tau) = \frac{F_0^*(\tau)}{\sqrt{1-\tau^2}}$$

where  $F_0(\tau)$  and  $F_0^*(\tau)$  are Hölder continuous in the interval  $-1 \leq \tau \leq 1$ .

Using the method described in [11] we obtain:

$$\sum_{j=1}^N F_0(\tau_j) \left[ \frac{1}{\tau_j - \kappa_i} + a\pi k_{11}^0(\kappa_i, \tau_j) \right] + \sum_{j=1}^N b\pi k_{12}^0(\kappa_i, \tau_j) F_0^*(\tau_j) \\ = -N \frac{(1-v_{xy}v_{yx})}{2\gamma_{14}E_y} p_1^0(\kappa_i) \quad i = 1, \dots, N-1$$

$$\sum_{j=1}^N a\pi k_{21}^0(\kappa_i, \tau_j) F_0(\tau_j) + \sum_{j=1}^N F_0^*(\tau_j) \left[ \frac{1}{\tau_j - \kappa_i} + b\pi k_{22}^0(\kappa_i, \tau_j) \right] \\ = -N \frac{(1-v_{xy}^*v_{yx}^*)}{2\gamma_{14}^*E_y^*} p_2^0(\kappa_i) \quad i = 1, \dots, N-1$$

$$\sum_{j=1}^N \frac{\pi}{N} F_0(\tau_j) = 0 \quad \text{and} \quad \sum_{j=1}^N \frac{\pi}{N} F_0^*(\tau_j) = 0 \quad (7.3a, b, c, d)$$

where

$$\tau_j = \cos \frac{\pi}{2N} (2j-1) \quad j = 1, \dots, N$$

$$\kappa_i = \cos \frac{\pi i}{N} \quad i = 1, \dots, N-1$$

The  $2N$  unknowns  $F_0(\tau_j)$  and  $F_0^*(\tau_j)$  can be found by solving equations (7.3a, b, c, d). In this problem we are mostly interested in the computation of the stress intensity factors.

The stress intensity factors may be expressed in terms of the density functions  $F_0(\tau)$  and  $F_0^*(\tau)$  as follows:

$$\text{For } a < h_1: k_a = \lim_{x_1 \rightarrow a} \sqrt{2(x_1 - a)} \sigma_{1y}(x_1, 0)$$

$$\text{and for } b < h_2: k_b = \lim_{x_2 \rightarrow b} \sqrt{2(x_2 - b)} \sigma_{2y}(x_2, 0) \quad (7.4a, b)$$

Making use of equations (4.16a,b) and definitions (7.4a,b), after lengthy algebra (see Appendix D) we obtain:

$$k_a = - \frac{2\gamma_{14} E_y \sqrt{a}}{(1-\nu_{xy} \nu_{yx})} F_0(1)$$

and

$$k_b = - \frac{2\gamma_{14}^* E_y^* \sqrt{b}}{(1-\nu_{xy}^* \nu_{yx}^*)} F_0^*(1) \quad (7.5a, b)$$

The computation is done for generalized plane stress case only. Results can easily be obtained for plane strain case by redefining the elastic material constants. In the perturbation problem considered  $p_1$  and  $p_2$  are constant. Assuming that there is no constraint in  $x$ - direction,  $p_1$  and  $p_2$  satisfy the following condition:

$$\frac{p_1}{p_2} = \frac{E_y}{E_y^*}$$

where  $E_y$  and  $E_y^*$  are the Young's moduli in  $y$  direction.

Two material combinations are formed among the following three materials.

$$1. E_x = 55.16 \times 10^9 \text{ N/m}^2, E_y = 170.8 \times 10^9 \text{ N/m}^2,$$

$$G_{xy} = 4.82 \times 10^9 \text{ N/m}^2, v_{xy} = 0.036$$

$$(E_x = 8 \times 10^6 \text{ psi}), (E_y = 24.75 \times 10^6 \text{ psi}),$$

$$(G_{xy} = 0.7 \times 10^6 \text{ psi}), (v_{xy} = 0.036)$$

$$2. E_x = 134.4 \times 10^9 \text{ N/m}^2, E_y = 31.0 \times 10^9 \text{ N/m}^2,$$

$$G_{xy} = 24.12 \times 10^9 \text{ N/m}^2, v_{xy} = 0.650$$

$$(E_x = 19.5 \times 10^6 \text{ psi}), (E_y = 4.5 \times 10^6 \text{ psi}),$$

$$(G_{xy} = 3.5 \times 10^6 \text{ psi}), (v_{xy} = 0.650)$$

$$3. E_x = 154.7 \times 10^9 \text{ N/m}^2, E_y = 155.8 \times 10^9 \text{ N/m}^2,$$

$$G_{xy} = 59.65 \times 10^9 \text{ N/m}^2, v_{xy} = 0.300$$

$$(E_x = 22.447 \times 10^6 \text{ psi}), (E_y = 22.6 \times 10^6 \text{ psi}),$$

$$(G_{xy} = 8.655 \times 10^6 \text{ psi}), (v_{xy} = 0.300)$$

As it is seen from the values given above the first two materials are orthotropic, while the third is isotropic.

The following pairs of materials are used:

Combination I: The first layer is of material 1,  
the second of material 2.

Combination II: The first layer is of material 3, the second of material 2.

Choosing the same materials and letting  $a$ ,  $b$ ,  $h_1$  or  $h_2$  go to proper limits we recover all the special cases done in [5], [6], and [10].

Figures 4-12 show some of the calculated results. In Figures 4 and 5 the stress intensity factors  $k_a$  are plotted versus  $h_2/h_1$  for  $b=0$  (there is no crack in the second material) and for the two material combinations. For  $h_2=0$ , we recover the solution of colinear cracks imbedded in a homogeneous material (see [10]). It is important to note that in the colinear crack problem the material doesn't have to be isotropic. As  $h_2 \rightarrow \infty$ ,  $k_a$  reaches an asymptotic value which can be found in [5]. For a fixed  $h_2/h_1$  ratio,  $k_a$  increases as  $a/h_1$  becomes larger.

Figures 6 and 7 show the stress intensity factors  $k_b$  for the case  $a=0$ . In this case also, for  $h_1=0$ , we obtain the solution of colinear cracks. There is a critical value of  $(h_1/h_2)$  for which the stress intensity factor starts to decrease as the ratio  $b/h_2$  increases. For the examples done this critical ratio is between 0 and 0.5. For  $h_1 \rightarrow \infty$  the stress intensity factor  $k_b$  reaches an asymptotic value which also can be found in [5].

The stress intensity factors  $k_a$  and  $k_b$  when both layers contain cracks, are given in Figures 8-11.  $k_b \rightarrow 0$  as  $(b/h_2) \rightarrow 1$ , since the power singularity  $\gamma$  is less than 0.5 when the crack in the second material touches the interface. Another interesting result is obtained from the comparison of isotropic and orthotropic materials. As it is seen in Figure 12, for the same  $E_y$  and  $E_y^*$  the stress intensity factor  $k_a$  for orthotropic materials can be larger or smaller than the stress intensity factor  $k_a$  for isotropic materials depending on the other elastic constants. One can significantly reduce  $k_a$  by a convenient choice of the elastic constants. The materials used in the comparison are given in Table 1. The dependence of  $k_a$  on the materials constants is given in Table 2.  $G_{xy}$  and  $G_{xy}^*$  are the most important constants, while keeping  $E_y$  and  $E_y^*$  constant. To reduce the stress intensity factor  $k_a$ , it is sufficient to increase  $E_x^*$ ,  $G_{xy}^*$ ,  $v_{xy}^*$  or decrease  $E_x$ ,  $G_{xy}$ ,  $v_{xy}$ .

## 7.2 Case of Broken Laminates

The solution will be obtained by solving equations (5.2a,b), with the single-valuedness conditions,

$$\int_{-h_1}^{h_1} \phi(t) dt = 0 \quad \text{and} \quad \int_{-b}^b \phi^*(t) dt = 0 \quad (7.6a,b)$$

Defining again,

$$t = h_1 \tau_1, \quad x_1 = h_1 \kappa_1 \quad \text{for } -h_1 < x_1, t < h_1$$

and

$$t = b \tau_2, \quad x_2 = b \kappa_2 \quad \text{for } -b < x_2, t < b$$

the normalized form of equations (5.2a,b) and (7.6a,b) can be written as follows:

$$\begin{aligned} & \frac{1}{\pi} \int_{-1}^1 \left[ \frac{1}{\tau_1 - \kappa_1} + \pi h_1 k_{11}^0 s(\kappa_1, \tau_1) \right] \phi_0(\tau_1) d\tau_1 + h_1 \int_{-1}^1 k_{11}^0 f(\kappa_1, \tau_1) \phi_0(\tau_1) d\tau_1 \\ & + b \int_{-1}^1 k_{12}^0(\kappa_1, \tau_2) \phi_0^*(\tau_2) d\tau_2 = - \frac{(1-v_{xy}v_{yx})}{2\gamma_{14}E_y} p_1^0(\kappa_1) \quad -1 < \kappa_1 < 1 \\ & \frac{1}{\pi} \int_{-1}^1 \frac{\phi_0^*(\tau_2)}{\tau_2 - \kappa_2} d\tau_2 + h_1 \int_{-1}^1 k_{21}^0(\kappa_2, \tau_1) \phi_0(\tau_1) d\tau_1 + b \int_{-1}^1 k_{22}^0(\kappa_2, \tau_2) \phi_0^*(\tau_2) d\tau_2 = \\ & - \frac{(1-v_{xy}^*v_{yx}^*)}{2\gamma_{14}^*E_y^*} p_2^0(\kappa_2), \quad -1 < \kappa_2 < 1 \\ & \int_{-1}^1 \phi_0(\tau_1) d\tau_1 = 0 \quad \text{and} \quad \int_{-1}^1 \phi_0^*(\tau_2) d\tau_2 = 0 \quad (7.7a,b,c,d) \end{aligned}$$

To obtain the solution, we will use the numerical method described in [11]. Hence, we obtain:

$$\frac{1}{\pi} \sum_{j=1}^N w_j(\tau_{1j}) \left[ \frac{1}{\tau_{1j} - \kappa_{1i}} + \pi h_1 k_{11s}^0(\kappa_{1i}, \tau_{1j}) \right] F_0(\tau_{1j})$$

$$+ h_1 \sum_{j=1}^N w_j(\tau_{1j}) k_{11f}^0(\kappa_{1i}, \tau_{1j}) F_0(\tau_{1j})$$

$$+ b \frac{\pi}{N} \sum_{j=1}^N k_{12}^0(\kappa_{1i}, \tau_{2j}) F_0^*(\tau_{2j})$$

$$= - \frac{(1-v_{xy} v_{yx})}{2\gamma_{14} E_y} p_1^0(\kappa_{1i}) , \quad i = 1, \dots, N-1$$

$$\frac{1}{\pi} \sum_{j=1}^N \left[ \frac{1}{\tau_{2j} - \kappa_{2i}} + \pi b k_{22}^0(\kappa_{2i}, \tau_{2j}) \right] F_0^*(\tau_{2j})$$

$$+ \sum_{j=1}^N w_j(\tau_{1j}) k_{21}^0(\kappa_{2i}, \tau_{1j}) F_0(\tau_{1j})$$

$$= - \frac{(1-v_{xy} v_{yx})}{2\gamma_{14} E_y} p_2^0(\kappa_{2i}) , \quad i = 1, \dots, N-1$$

$$\sum_{j=1}^N w_j(\tau_{1j}) F_0(\tau_{1j}) = 0 \quad \text{and} \quad \frac{\pi}{N} \sum_{j=1}^N F_0^*(\tau_{2j}) = 0$$

(7.8a,b,c,d)

where

$$\phi_0(\tau_1) = \frac{F_0(\tau_1)}{(1-\tau_1^2)^{\gamma}} , \quad \phi_0^*(\tau_2) = \frac{F_0^*(\tau_2)}{\sqrt{1-\tau_2^2}}$$

$$p_N^{(-\gamma, -\gamma)}(\tau_{1j}) = 0 , \quad j = 1, \dots, N$$

$$p_{N-1}^{(1-\gamma, 1-\gamma)}(\kappa_{1i}) = 0 , \quad i = 1, \dots, N-1$$

$$\tau_{2j} = \cos \frac{\pi}{2N} (2j-1), \quad j = 1, \dots, N$$

$$\kappa_{2i} = \cos \frac{\pi i}{N}, \quad i = 1, \dots, N-1$$

and  $w_j(\tau_{1j})$  are the weights of  $P_N^{(-\gamma, -\gamma)}(\tau_{1j})$ .

Solving the  $2N \times 2N$  system of linear equations one can find the  $2N$  unknowns  $F_0(\tau_{1j})$  and  $F_0^*(\tau_{2j})$ . But again we are interested in the stress intensity factors, which can be calculated in terms of  $F_0(\tau_{1j})$  and  $F_0^*(\tau_{2j})$ .

Define:

$$k_a = \lim_{x_2 \rightarrow -h_2} 2^Y (x_2 + h_2)^Y \sigma_{2y}(x_2, 0)$$

and

$$k_b = \lim_{x_2 \rightarrow b} \sqrt{2(x_2 - b)} \sigma_{2y}(x_2, 0)$$

After some calculation shown in Appendix D, we have:

$$k_a = \frac{(h_1) Y Y_1^* E Y_0^*(1)}{(1 - v_{xy}^* v_{yx}^*) \sin \pi Y} \left[ \lambda_{101} \begin{bmatrix} |w_1| \sqrt{B_5} \\ \frac{|w_1^*| |w_1|}{\sqrt{B_5}} Y \end{bmatrix} + \lambda_{102} \begin{bmatrix} |w_1| \sqrt{B_5} \\ \frac{|w_3^*| |w_1|}{\sqrt{B_5}} Y \end{bmatrix} \right. \\ \left. + \lambda_{103} \begin{bmatrix} |w_3| \sqrt{B_5} \\ \frac{|w_3| |w_1^*|}{\sqrt{B_5}} Y \end{bmatrix} + \lambda_{104} \begin{bmatrix} |w_3| \sqrt{B_5} \\ \frac{|w_3| |w_3^*|}{\sqrt{B_5}} Y \end{bmatrix} \right]$$

and

$$k_b = -\frac{2Y_{14}^* E_y^* \sqrt{b}}{(1-v_{xy}^* v_{yx}^*)} F_0^*(1) \quad (7.9a,b)$$

The results for the case of broken laminates are shown in Figures 13-16. Again the same material combinations are used. In Figure 13 the stress intensity factor  $k_a$  is plotted versus the ratio  $h_2/h_1$  for  $b = 0$ . When  $h_2 \rightarrow \infty$   $k_a$  has an asymptotic value which can be recovered in [7]. Figure 14 shows the variation of  $k_b$  with  $h_1/h_2$ , for the case  $a = 0$ . For  $b \neq 0$ , the variation of  $k_a$  and  $k_b$  with  $b/h_2$  are given in Figures 15 and 16.

### 7.3 Case of a Crack Crossing the Interface

In this case the governing singular integral equations are (6.3a,b). As it was pointed out in Section 6, the characteristic equation (6.10) does not always give a singularity power at the bimaterial interface. Therefore, the numerical solution needs care, and we should solve equations (6.3a,b) considering the singular and non-singular cases at the bimaterial interfaces.

#### 7.3.1 Singular Behavior at the Interface

For the material combination II (isotropic-orthotropic) equation (6.10) has a root between 0 and 1. Using Newton-Raphson method to solve equation (6.10), we have:

$$\beta = 0.04248 .$$

We will make use of the following definitions to normalize equations (6.3a,b), the single-valuedness condition, and the relation (6.9a):

$$t = h_1 \tau_1 , \quad x_1 = h_1 \kappa_1 \quad \text{for } -h_1 < t, \quad x_1 < h_1$$

$$t = \frac{h_2 - c}{2} \tau_2 + \frac{h_2 + c}{2} , \quad x_2 = \frac{h_2 - c}{2} \kappa_2 + \frac{h_2 + c}{2} \quad \text{for } c < t, \quad x_2 < h_2.$$

Then, we obtain:

$$\frac{1}{\pi} \int_{-1}^1 \frac{\phi_0(\tau_1)}{\tau_1 - \kappa_1} d\tau_1 + h_1 \int_{-1}^1 k_{11s}^0(\kappa_1, \tau_1) \phi_0(\tau_1) d\tau_1 + h_1 \int_{-1}^1 k_{11f}^0(\kappa_1, \tau_1) \phi_0(\tau_1) d\tau_1$$

$$+ \frac{h_2 - c}{2} \int_{-1}^1 k_{12s}^{*0}(\kappa_1, \tau_2) \phi_0^*(\tau_2) d\tau_2 + \frac{h_2 - c}{2} \int_{-1}^1 k_{12f}^{*0}(\kappa_1, \tau_2) \phi_0^*(\tau_2) d\tau_2$$

$$= - \frac{(1 - v_{xy}^* v_{yx}^*)}{2 \gamma_{14} E_y} p_1^0(\kappa_1) \quad -1 < \kappa_1 < 1$$

$$\frac{1}{\pi} \int_{-1}^1 \left[ \frac{1}{\tau_2 - \kappa_2} + \frac{1}{\tau_2 + \kappa_2 + \frac{2(h_2 + c)}{h_2 - c}} \right] \phi_0^*(\tau_2) d\tau_2$$

$$+ h_1 \int_{-1}^1 k_{21s}^0(\kappa_2, \tau_1) \phi_0(\tau_1) d\tau_1 + h_1 \int_{-1}^1 k_{21f}^0(\kappa_2, \tau_1) \phi_0(\tau_1) d\tau_1$$

$$+ \frac{h_2 - c}{2} \int_{-1}^1 k_{22s}^{*0}(\kappa_2, \tau_2) \phi_0^*(\tau_2) d\tau_2 + \frac{h_2 - c}{2} \int_{-1}^1 k_{22f}^{*0}(\kappa_2, \tau_2) \phi_0^*(\tau_2) d\tau_2$$

$$= - \frac{(1 - v_{xy}^* v_{yx}^*)}{2 \gamma_{14} E_y} p_2^0(\kappa_2) \quad -1 < \kappa_2 < 1$$

$$h_1 \int_{-1}^1 \phi_0(\tau_1) d\tau_1 = 0 \quad \text{and} \quad F_0^*(1) = - \sqrt{2} \left[ \frac{h_1}{h_2 - c} \right]^{\beta} \frac{a_1}{a_2} F_0(1)$$

(7.10a,b,c,d)

where

$$a_1 = -2\cos\pi\beta + \lambda_{85} \frac{|w_1|}{\sqrt{\beta_s}} \left[ \frac{1}{\frac{|w_1|}{\sqrt{\beta_s}} \beta} + \lambda_{86} \frac{|w_1|}{\sqrt{\beta_s}} \frac{1}{\left( \frac{|w_1| |w_3|}{\sqrt{\beta_s}} \right) \beta} \right]$$

$$+ \lambda_{87} \frac{|w_3|}{\sqrt{\beta_s}} \left[ \frac{1}{\frac{|w_1| |w_3|}{\sqrt{\beta_s}} \beta} + \lambda_{88} \frac{|w_3|}{\sqrt{\beta_s}} \frac{1}{\left( \frac{|w_3|^2}{\sqrt{\beta_s}} \right) \beta} \right]$$

$$a_2 = \lambda_{93} \frac{|w_1^*|}{\sqrt{\beta_s^*}} \left[ \frac{1}{\frac{|w_1| |w_1^*|}{\sqrt{\beta_s^*}} \beta} + \lambda_{94} \frac{|w_1^*|}{\sqrt{\beta_s^*}} \frac{1}{\left( \frac{|w_3| |w_1^*|}{\sqrt{\beta_s^*}} \right) \beta} \right]$$

$$+ \lambda_{95} \frac{|w_3^*|}{\sqrt{\beta_s^*}} \left[ \frac{1}{\frac{|w_1| |w_3^*|}{\sqrt{\beta_s^*}} \beta} + \lambda_{96} \frac{|w_3^*|}{\sqrt{\beta_s^*}} \frac{1}{\left( \frac{|w_3| |w_3^*|}{\sqrt{\beta_s^*}} \right) \beta} \right]$$

Using the numerical method given in [11] equations (7.10  
a,b,c,d) further reduce to:

$$\begin{aligned} & \frac{1}{\pi} \sum_{j=1}^N w_j(\tau_{1j}) \left[ \frac{1}{\tau_{1j} - \kappa_{1i}} + \pi h_1 k_{11s}^0(\kappa_{1i}, \tau_{1j}) + \pi h_1 k_{11f}^0(\kappa_{1i}, \tau_{1j}) \right] F_0(\tau_{1j}) \\ & + \sum_{j=1}^N w_j^*(\tau_{2j}) \left[ \frac{h_2 - c}{2} k_{12s}^{*0}(\kappa_{1i}, \tau_{2j}) + \frac{h_2 - c}{2} k_{12f}^{*0}(\kappa_{1i}, \tau_{2j}) \right] F_0^*(\tau_{2j}) \\ & = - \frac{(1 - v_{xy} v_{yx})}{2 Y_{14} E_y} p_i^0(\kappa_{1i}) \quad i = 1, \dots, N-1 \end{aligned}$$

$$\begin{aligned}
& \frac{1}{\pi} \sum_{j=1}^N w_j^*(\tau_{2j}) \left[ \frac{1}{\tau_{2j}-\kappa_{2i}} + \frac{1}{\tau_{2j}+\kappa_{2i} + \frac{2(h_2+c)}{h_2-c}} + \pi \frac{h_2-c}{2} k_{22s}^{*0}(\kappa_{2i}, \tau_{2j}) \right. \\
& \left. + \pi \frac{h_2-c}{2} k_{22f}^{*0}(\kappa_{2i}, \tau_{2j}) \right] F_0^*(\tau_{2j}) + \sum_{j=1}^N w_j(\tau_{1j}) [h_1 k_{12s}^0(\kappa_{2i}, \tau_{1j}) \\
& + h_1 k_{21f}^0(\kappa_{2i}, \tau_{1j})] F_0(\tau_{1j}) = - \frac{(1-v_{xy}^* v_{yx}^*)}{2\gamma_{14}^* E_y} p_2^0(\kappa_{2i}) \quad i = 1, \dots, N-1
\end{aligned}$$

$$h_1 \sum_{j=1}^N w_j(\tau_{1j}) F_0(\tau_{1j}) = 0$$

and

$$F_0^*(1) = -\sqrt{\epsilon} \left( \frac{h_1}{h_2-c} \right)^\beta \frac{a_1}{a_2} F_0(1) \quad (7.11a, b, c, d)$$

where

$$\phi_0(\tau) = \frac{F_0(\tau)}{(1-\tau^2)^\beta}, \quad \phi_0^*(\tau) = \frac{F_0^*(\tau)}{(1-\tau)^\beta (1+\tau)^{\frac{1}{2}}}$$

$$p_N^{(-\beta, -\beta)}(\tau_{1j}) = 0 \quad j = 1, \dots, N$$

$$p_{N-1}^{(1-\beta, 1-\beta)}(\kappa_{1i}) = 0 \quad i = 1, \dots, N-1$$

$$p_N^{(-\beta, -\frac{1}{2})}(\tau_{2j}) = 0 \quad j = 1, \dots, N$$

$$p_{N-1}^{(1-\beta, \frac{1}{2})}(\kappa_{2i}) = 0 \quad i = 1, \dots, N-1$$

$w_j(\tau_{1j})$  and  $w_j^*(\tau_{2j})$  are the corresponding weights of

$P_N^{(-\beta, -\beta)}(\tau_{1j})$  and  $P_N^{(-\beta, -\frac{1}{2})}(\tau_{2j})$ .

Solving the  $2N \times 2N$  system of linear equations, we obtain the  $2N$  unknowns  $F_0(\tau_{1j})$  and  $F_0^*(\tau_{2j})$ . The stress intensity factors can be defined as follows:

$$k_b = \lim_{x_2 \rightarrow c} \sqrt{2(c-x_2)} \sigma_{2y}(x_2, 0)$$

and at the bimaterial interfaces

$$k_{xx} = \lim_{y \rightarrow 0^+} y^\beta \sigma_{1x}(h_1, y)$$

$$k_{xy} = \lim_{y \rightarrow 0^+} y^\beta \tau_{1xy}(h_1, y) \quad (7.12a, b, c)$$

By making use of definitions (7.12a, b, c) and after lengthy calculation shown in Appendix D we obtain:

$$k_b = \frac{2^{1-\beta} Y_{14}^* E_y^*}{(1-v_{xy}^* v_{yx}^*) \sqrt{h_2 - c}} F_0^*(-1)$$

$$k_{xx} = \frac{E_x}{(1-v_{xy}^* v_{yx}^*)} \frac{1}{2^{\beta+1} \sin \frac{\pi \beta}{2}} \left\{ \frac{h_1^\beta}{Y_{13}} \left[ \frac{Y_1}{\frac{|w_1|}{\sqrt{\beta_5}}^\beta} - Y_2 \frac{Y_{11}}{Y_{12}} \frac{1}{\frac{|w_3|}{\sqrt{\beta_5}}^\beta} \right] \right.$$

$$\left. - \frac{|w_1|}{\sqrt{\beta_5}} \frac{(Y_3 \lambda_{81} + Y_4 \lambda_{82})}{\frac{|w_1|}{\sqrt{\beta_5}}^\beta \lambda_{80}} - \frac{|w_3|}{\sqrt{\beta_5}} \frac{(Y_3 \lambda_{83} + Y_4 \lambda_{84})}{\frac{|w_3|}{\sqrt{\beta_5}}^\beta \lambda_{80}} \right] F_0(-1)$$

$$+ \frac{(h_2 - c)^\beta}{\sqrt{2} Y^* \lambda} \left[ \frac{|w_1^*|}{\frac{|w_1^*|}{\sqrt{\beta_5^*}}^\beta} \frac{(Y_3 \lambda_{89} + Y_4 \lambda_{90})}{\frac{|w_1^*|}{\sqrt{\beta_5^*}}^\beta} + \frac{|w_3^*|}{\frac{|w_3^*|}{\sqrt{\beta_5^*}}^\beta} \frac{(Y_3 \lambda_{91} + Y_4 \lambda_{92})}{\frac{|w_3^*|}{\sqrt{\beta_5^*}}^\beta} \right] F_0^*(1) \}$$

$$\begin{aligned}
 k_{xy} = & \frac{G_{xy}}{2^{\beta+1} \cos \frac{\pi \beta}{2}} \left\{ \frac{h_1^\beta}{Y_{13}} \left[ - \frac{\gamma_{11}}{\left( \frac{|w_1|}{\sqrt{\beta_s}} \right)^\beta} + \frac{\gamma_{11}}{\left( \frac{|w_3|}{\sqrt{\beta_s}} \right)^\beta} \right. \right. \\
 & - \frac{(\gamma_9 \lambda_{81} + \gamma_{10} \lambda_{82})}{\lambda_{80} \left( \frac{|w_1|}{\sqrt{\beta_s}} \right)^\beta} \frac{|w_1|}{\sqrt{\beta_s}} - \frac{(\gamma_9 \lambda_{83} + \gamma_{10} \lambda_{84})}{\lambda_{80} \left( \frac{|w_3|}{\sqrt{\beta_s}} \right)^\beta} \frac{|w_3|}{\sqrt{\beta_s}} \Big] F_0(-1) \\
 & \left. + \frac{(h_2 - c)^\beta}{\sqrt{2} Y_{13}^* \lambda_{80}} \left[ \frac{(\gamma_9 \lambda_{89} + \gamma_{10} \lambda_{90})}{\left( \frac{|w_1^*|}{\sqrt{\beta_s^*}} \right)^\beta} \frac{|w_1^*|}{\sqrt{\beta_s^*}} + \frac{(\gamma_9 \lambda_{91} + \gamma_{10} \lambda_{92})}{\left( \frac{|w_3^*|}{\sqrt{\beta_s^*}} \right)^\beta} \frac{|w_3^*|}{\sqrt{\beta_s^*}} \right] F_0^*(1) \right\}
 \end{aligned}$$

(7.13a,b,c)

Extrapolating the results found from equations (7.11a,b,c, d) the stress intensity factors  $k_b$ ,  $k_{xx}$ ,  $k_{xy}$  can be computed in a straight-forward manner. The results are shown in Figures 17-19. Figure 17 shows the variation of  $k_b$  with  $c/h_2$ , for different values of  $(h_1/h_2)$  ratio.  $k_b$  increases as  $(h_1/h_2)$  increases. Figures 18 and 19 show the variation of  $k_{xx}$  and  $k_{xy}$  with respect to  $c/h_2$ .

### 7.3.2 Non-Singular Behavior at the Interface

In this case, the characteristic equation (6.10) has no root and therefore the surface displacement derivatives are bounded. Since, as it was shown in Section 6, the displacement derivatives are continuous at the bimaterial interfaces, using the single-valuedness condition to write

the integrals from 0 to  $\ell$ , equations (6.2a,b) take the form  
 (see [12]):

$$\frac{1}{\pi} \int_0^\ell \left[ \frac{1}{r-s} + \frac{1}{r+s} \right] G(r) dr + \int_0^\ell k(r,s) G(r) dr = p(s) \quad 0 < s < \ell \quad (7.10)$$

where

$$G(r) = \begin{cases} \phi(t) & (0 < t < h_1, 0 < r < h_1) \\ \phi^*(t) & (-h_2 < t < -c, h_1 < r < \ell) \end{cases}$$

$$p(s) = \begin{cases} -\frac{(1-v_{xy}v_{yx})}{2\gamma_{14}E_y} p_1(x_1) & (0 < x_1 < h_1, 0 < s < h_1) \\ -\frac{(1-v_{xy}^*v_{yx}^*)}{2\gamma_{14}^*E_y^*} p_2(x_2) & (-h_2 < x_2 < -c, h_1 < s < \ell) \end{cases}$$

$$k(r,s) = \begin{cases} k_{11}^*(x_1, t) & (0 < x_1, t < h_1, 0 < r, s < h_1) \\ k_{12}^*(x_1, t) - \frac{1}{\pi} \left( \frac{1}{t-x_1} + \frac{1}{t+x_1} \right) & \begin{cases} 0 < x_1 < h_1, 0 < s < h_1 \\ -h_2 < t < -c, h_1 < r < \ell \end{cases} \\ k_{21}^*(x_2, t) - \frac{1}{\pi} \left( \frac{1}{t-x_2} + \frac{1}{t+x_2} \right) & \begin{cases} -h_2 < x_2 < -c, h_1 < s < \ell \\ 0 < t < h_1, 0 < r < h_1 \end{cases} \\ k_{22}^*(x_2, t) & (-h_2 < x_2, t < -c, h_1 < r, s < \ell) \end{cases}$$

with

$$k_{ij}^*(x_i, t) = k_{ij}(x_i, t) - k_{ij}(x_i, -t) \quad (i, j = 1, 2)$$

Now we have the governing equations for a crack imbedded in a non-homogeneous material, obviously with power singularity - $\frac{1}{2}$  at the crack tip. Normalizing equation (7.10) by means of:

$$r = \ell\tau \quad \text{and} \quad s = \ell\kappa$$

we obtain:

$$\frac{1}{\pi} \int_0^1 \left[ \frac{1}{\tau-\kappa} + \frac{1}{\tau+\kappa} \right] G_0(\tau) d\tau + \ell \int_0^1 k_0(\tau, \kappa) G_0(\tau) d\tau = p_0(\kappa) \quad 0 < \kappa < 1 \quad (7.11)$$

Equation (7.11) reduces to a set of linear equations by using the method of collocations (see [11]):

$$\sum_{j=1}^N \left[ \frac{1}{\tau_j - \kappa_i} + \frac{1}{\tau_j + \kappa_i} + \pi \ell k_0(\tau_j, \kappa_i) \right] H_0(\tau_j) = 2N p_0(\kappa_i) \quad i = 1, \dots, N \quad (7.12)$$

where

$$\tau_j = \cos \left[ \frac{2j-1}{4N} \pi \right], \quad j = 1, \dots, N$$

$$\kappa_i = \cos \left[ \frac{i}{2N} \pi \right], \quad i = 1, \dots, N$$

$$G_0(\tau) = \frac{H_0(\tau)}{\sqrt{1-\tau^2}}$$

The N unknowns  $H_0(\tau_j)$  can be found from equation (7.12) in a straight-forward manner.

Defining the stress intensity factor at the crack tip as:

$$k_b = \lim_{x_2 \rightarrow l} \sqrt{2(x_2 - l)} \sigma_{2y}(x_2, 0)$$

we obtain:

$$k_b = -\frac{2\gamma_{14}^* E_y^* \sqrt{l}}{(1-\nu_{xy}^* \nu_{yx}^*)} H_0(1) \quad (7.13)$$

Using the same material for both strips, we recover again the results of colinear cracks in homogeneous medium. Figure 20 shows the variation of  $k_b$  with  $c/h_2$ .  $k_b$  increases as  $h_1/h_2$  increases.

## 8. CONCLUSIONS

The fracture problem of layered orthotropic composite plates has been studied. The following results have been obtained:

- 1) Depending on the elastic constants, orthotropic materials can be classified in two groups: materials of type I and materials of type II. (A different formulation is needed for each combination.)
- 2) The colinear crack solution is the same for homogeneous isotropic and orthotropic materials.
- 3) In the case of an internal crack in the first layer, the stress intensity factor  $k_a$  can be reduced significantly by a proper selection of the elastic constants.
- 4) For the case of broken laminates there is a singularity power which can be found from equation (5.7). The singularity power  $\gamma$  varies between 0 and 1 for different material combinations.
- 5) For a crack crossing the interface, the singular behavior at the interface disappears for some material combinations. In this case the crack surface displacement derivatives are bounded and continuous, and all stresses are bounded at the bimaterial interfaces.

## **9. RECOMMENDATIONS**

In the present work, a general formulation of the fracture problem of layered orthotropic composites with periodic cracks is given. The formulation is done only for the case where both materials are of type I. Following the same procedure, the problem can also be studied for orthotropic materials of type II, or for the combination of type I and type II. The dependence of the singular behavior at the interface on the elastic constants can also be investigated.

In our formulation the thickness of the adhesive bonding the layers has been neglected. The study of the effect of the adhesive also can be recommended.

A more realistic approach also would be to study the problem of finite number of strips. But this problem requires lengthy algebra.

There are many other problems to be studied in the fracture of composites. We hope that our work will have a small contribution in the study of these problems.

**TABLES**

**1-2**

Mat. No.	$E_x$	$E_y$	$G_{xy}$	$v_{xy}$
1	$55.16 \times 10^9 \text{ N/m}^2$ ( $8 \times 10^6 \text{ psi}$ )	$170.8 \times 10^9 \text{ N/m}^2$ ( $24.75 \times 10^6 \text{ psi}$ )	$4.82 \times 10^9 \text{ N/m}^2$ ( $0.7 \times 10^6 \text{ psi}$ )	0.036 (0.036)
2	$134.42 \times 10^9 \text{ N/m}^2$ ( $19.5 \times 10^6 \text{ psi}$ )	$31.01 \times 10^9 \text{ N/m}^2$ ( $4.5 \times 10^6 \text{ psi}$ )	$24.12 \times 10^9 \text{ N/m}^2$ ( $3.5 \times 10^6 \text{ psi}$ )	0.650 (0.650)
3	$154.7 \times 10^9 \text{ N/m}^2$ ( $22.447 \times 10^6 \text{ psi}$ )	$155.8 \times 10^9 \text{ N/m}^2$ ( $22.6 \times 10^6 \text{ psi}$ )	$59.65 \times 10^9 \text{ N/m}^2$ ( $8.655 \times 10^6 \text{ psi}$ )	0.300 (0.300)
4	$167.5 \times 10^9 \text{ N/m}^2$ ( $24.3 \times 10^6 \text{ psi}$ )	$170.8 \times 10^9 \text{ N/m}^2$ ( $24.75 \times 10^6 \text{ psi}$ )	$62.4 \times 10^9 \text{ N/m}^2$ ( $9.05 \times 10^6 \text{ psi}$ )	0.300 (0.300)
5	$10.06 \times 10^9 \text{ N/m}^2$ ( $1.46 \times 10^6 \text{ psi}$ )	$31.01 \times 10^9 \text{ N/m}^2$ ( $4.5 \times 10^6 \text{ psi}$ )	$8.825 \times 10^8 \text{ N/m}^2$ ( $0.128 \times 10^6 \text{ psi}$ )	0.036 (0.036)
6	$30.34 \times 10^9 \text{ N/m}^2$ ( $4.4 \times 10^6 \text{ psi}$ )	$31.01 \times 10^9 \text{ N/m}^2$ ( $4.5 \times 10^6 \text{ psi}$ )	$10.82 \times 10^9 \text{ N/m}^2$ ( $1.57 \times 10^6 \text{ psi}$ )	0.400 (0.400)
Comb. No.	Material of the first strip	Material of the second strip		
I	1	2		
II	3	2		
III	4	2		
IV	4	6		
V	4	5		

Table 1.  
The elastic constants  
and material  
combinations

Material constant		$k_a$
$E_x$	increases	increases
$E_x^*$	increases	decreases
$G_{xy}$	increases	increases
$G_{xy}^*$	increases	decreases
$v_{xy}$	increases	increases
$v_{xy}^*$	increases	decreases
$E_y$ and $E_y^*$ are kept constant		

Table 2. Dependence of  $k_a$  on the elastic constants

**FIGURES**

**1-20**

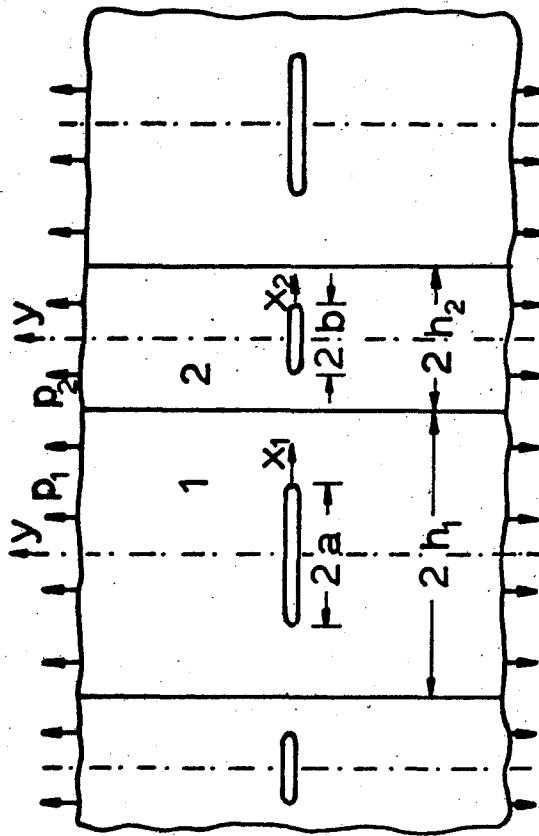


Figure 1. Geometry of the composite plate.

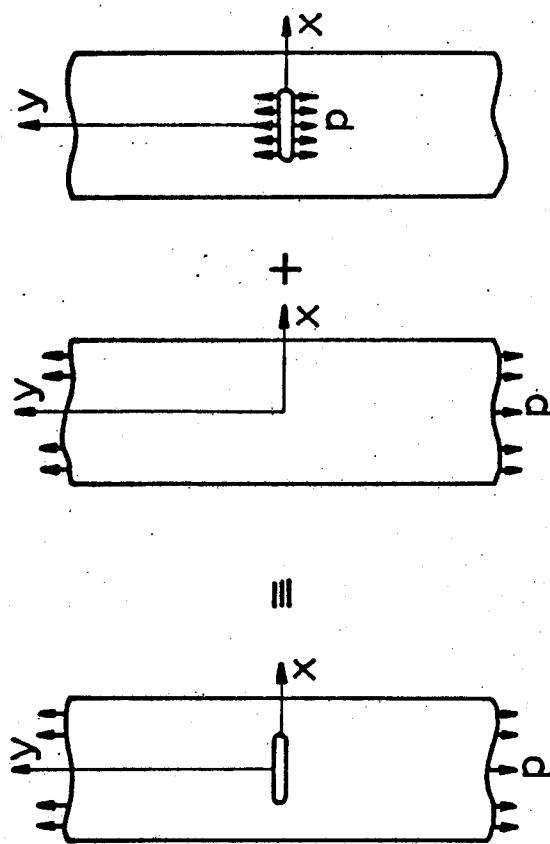


Figure 2. The superposition.

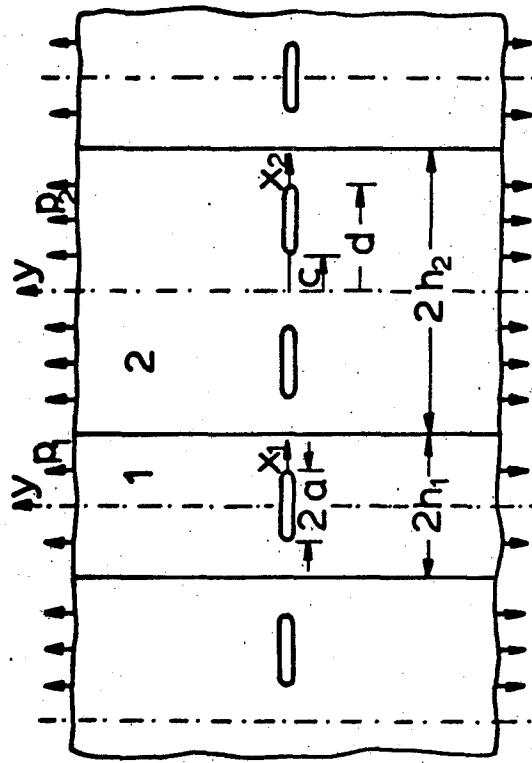


Figure 3. Initial geometry for the case of a crack crossing the interface.

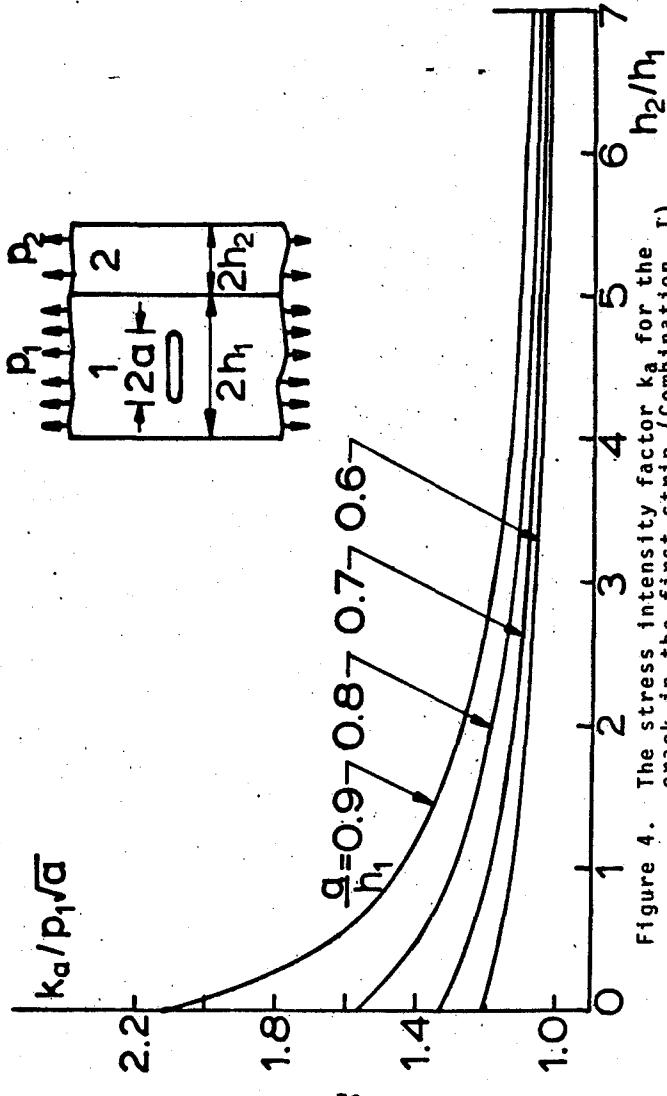


Figure 4. The stress intensity factor  $k_a$  for the crack in the first strip (Combination I).

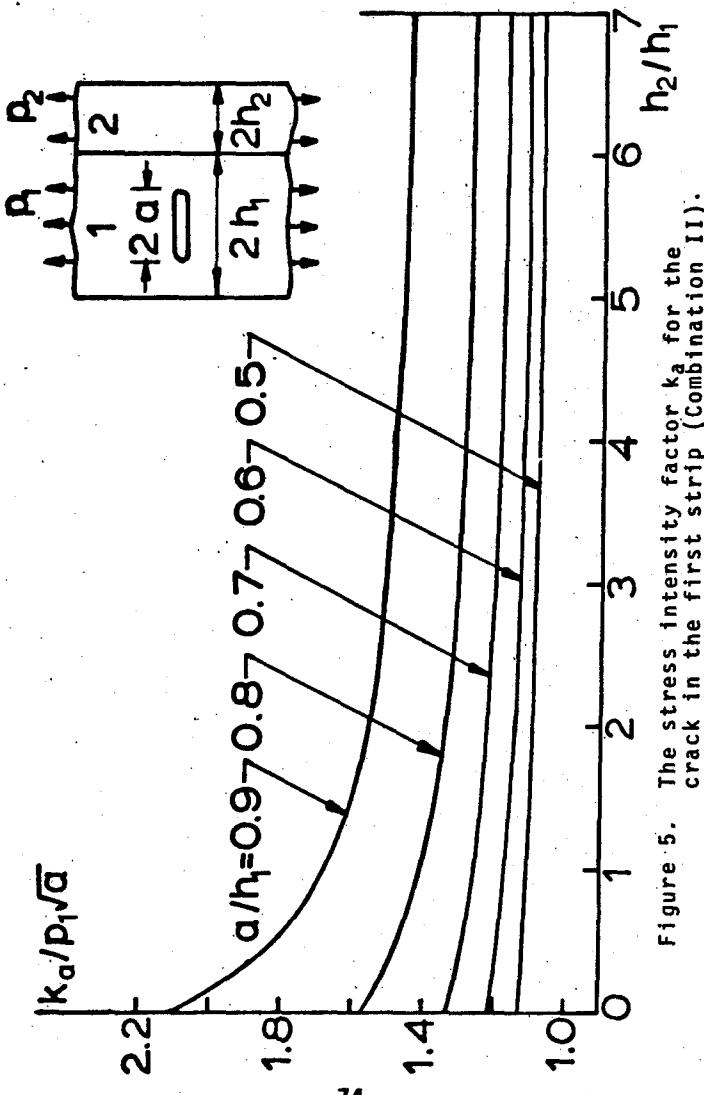


Figure 5. The stress intensity factor  $k_a$  for the crack in the first strip (Combination III).

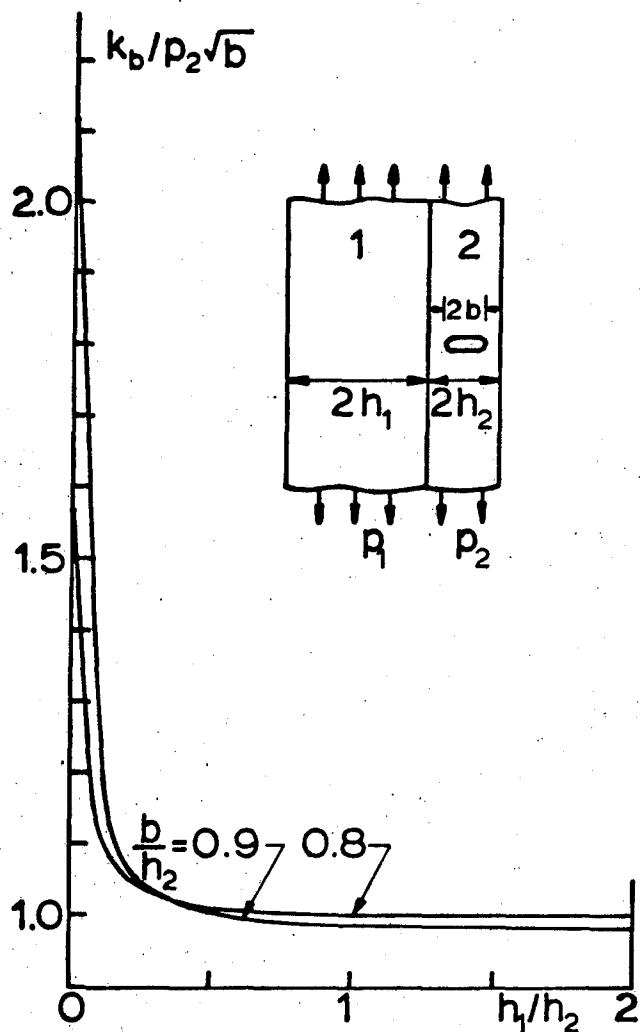


Figure 6. The stress intensity factor  $k_b$  for the crack in buffer strip (Combination I).

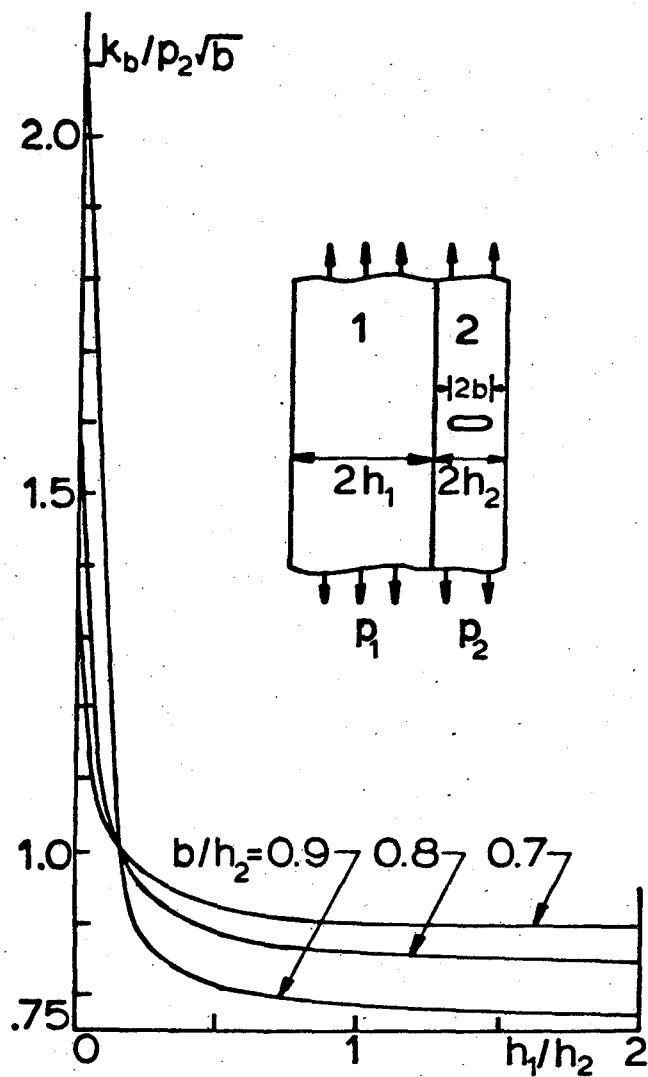


Figure 7. The stress intensity factor  $k_b$  for the crack in buffer strip (Combination II).

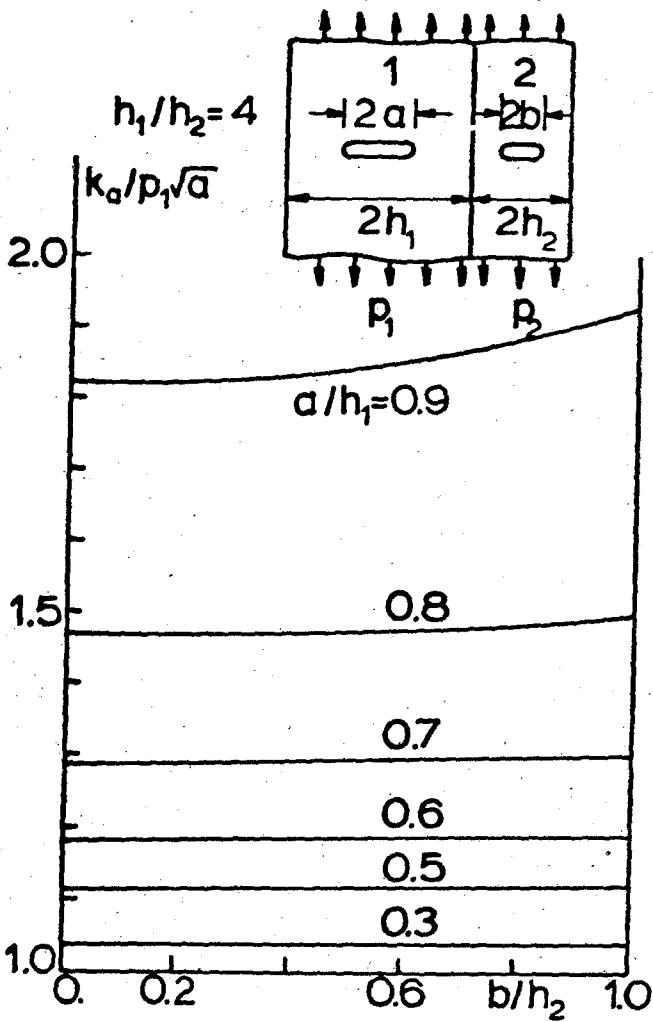
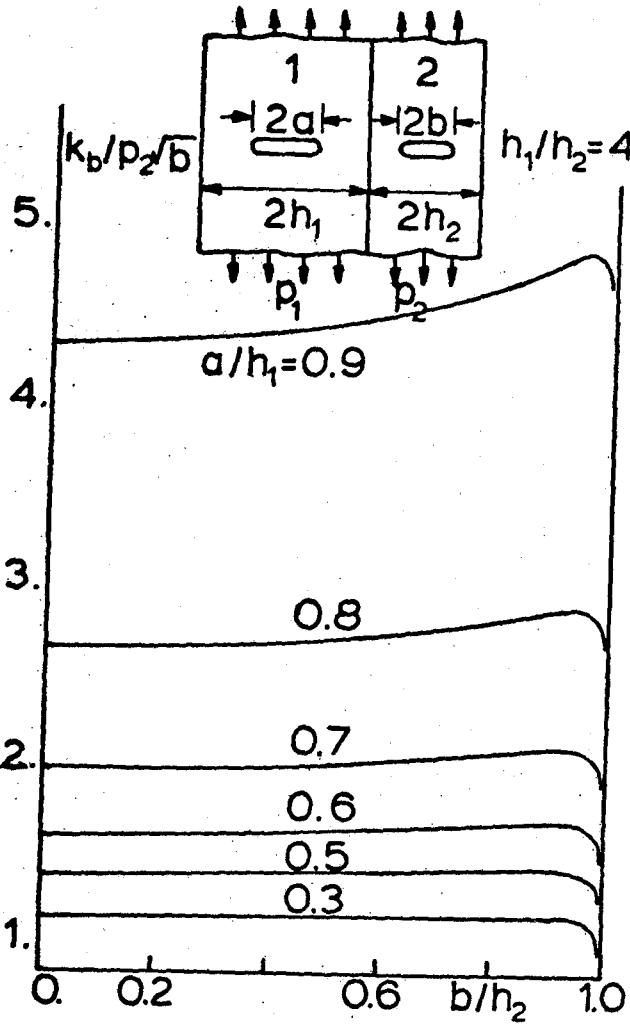


Figure 8. The stress intensity factor  $k_a$  for the case in which both strips contain cracks (Combination I).



**Figure 9.** The stress intensity factor  $k_b$  for the case in which both strips contain cracks (Combination I).

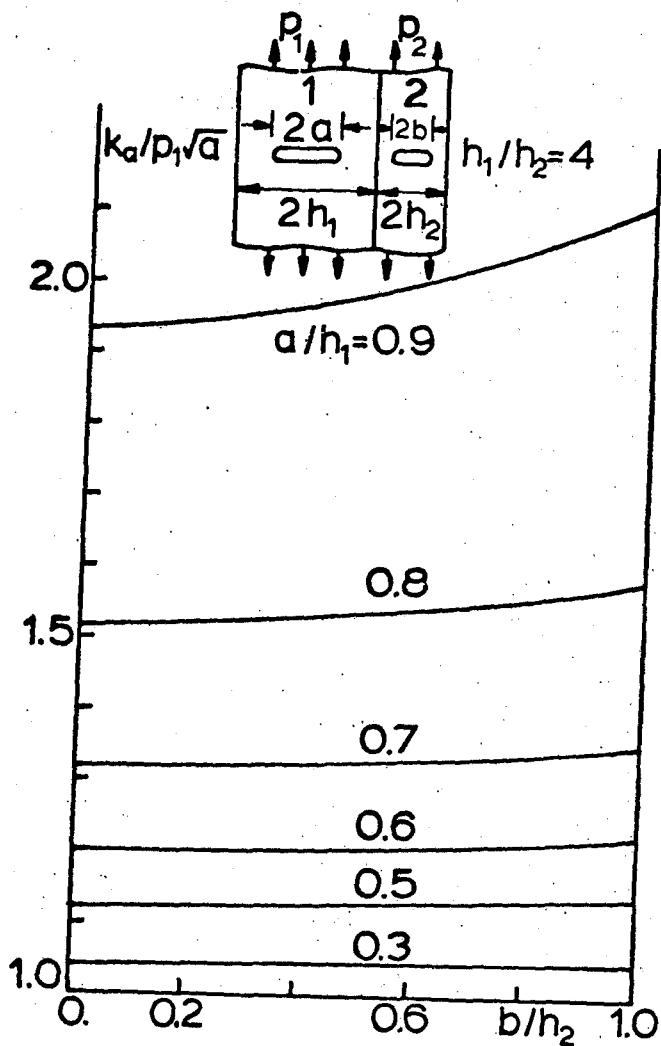


Figure 10. The stress intensity factor  $k_a$  for the case in which both strips contain cracks (Combination II).

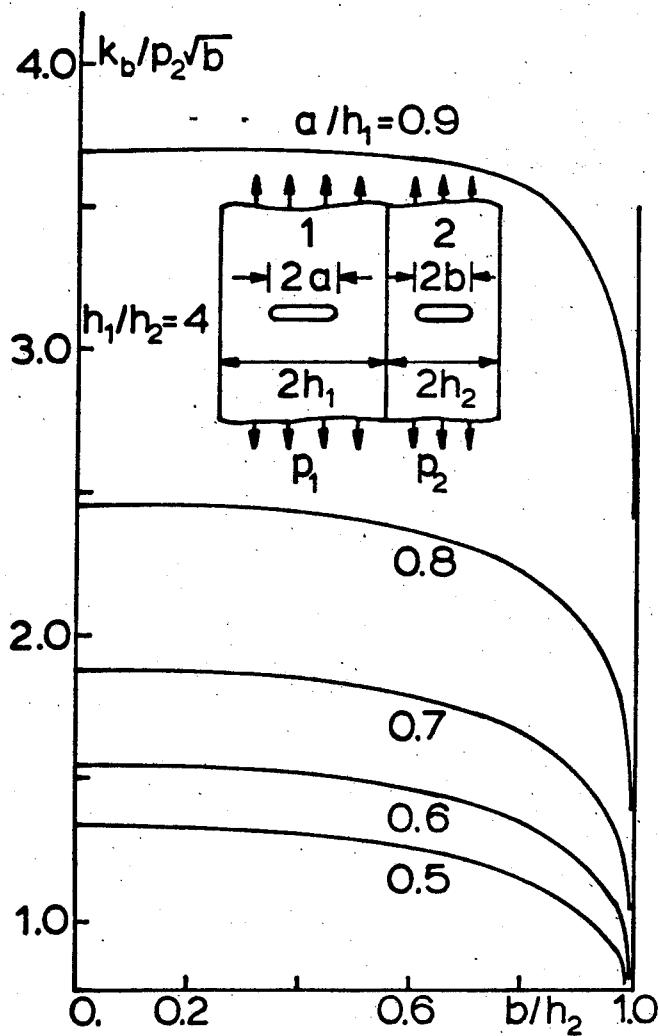


Figure 11. The stress intensity factor  $k_b$  for the case in which both strips contain cracks (Combination II).

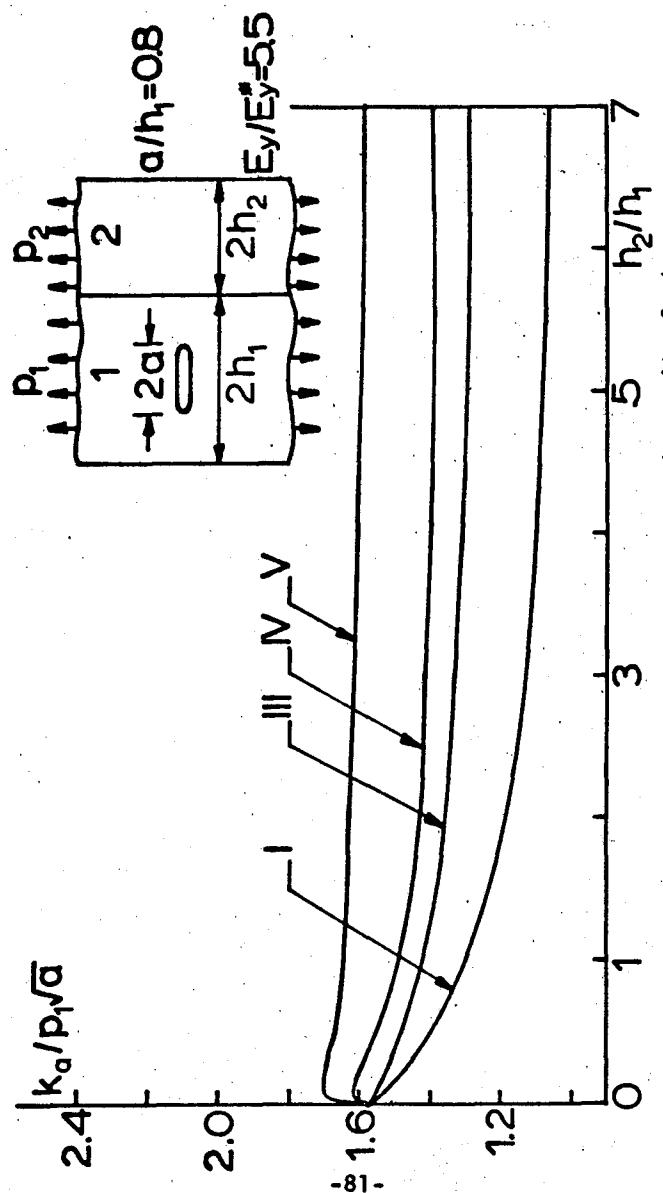


Figure 12. Comparison of the stress intensity factor  $k_a$  for isotropic and orthotropic materials.

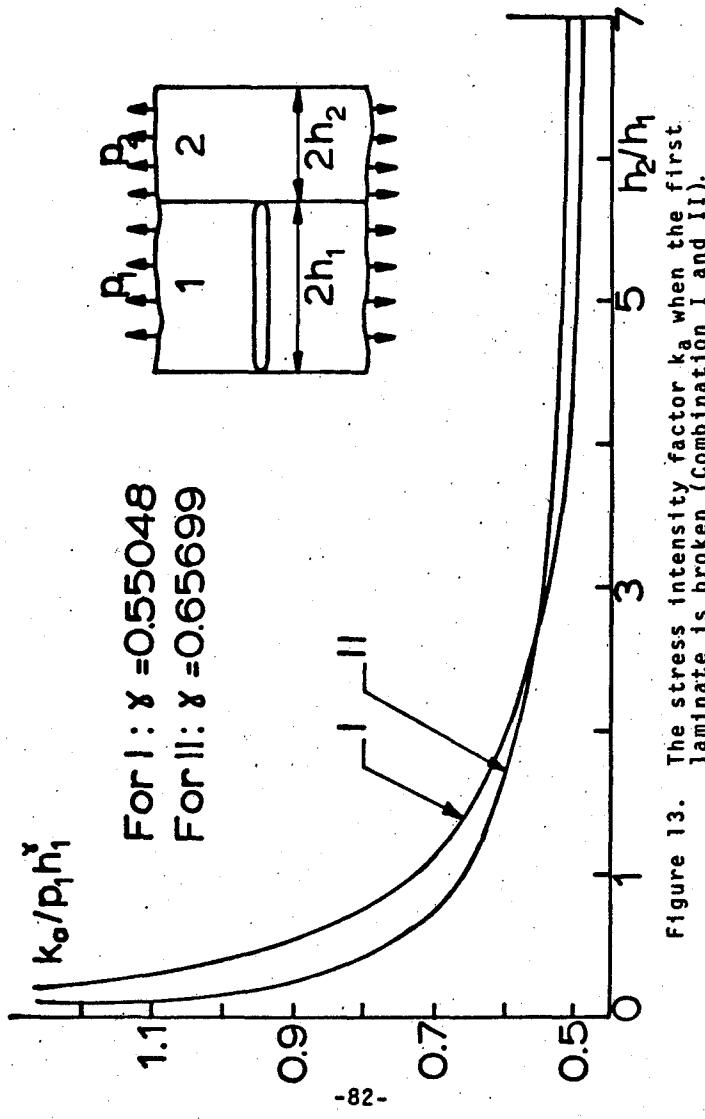


Figure 13. The stress intensity factor  $k_a$  when the first laminate is broken (Combination I and II).

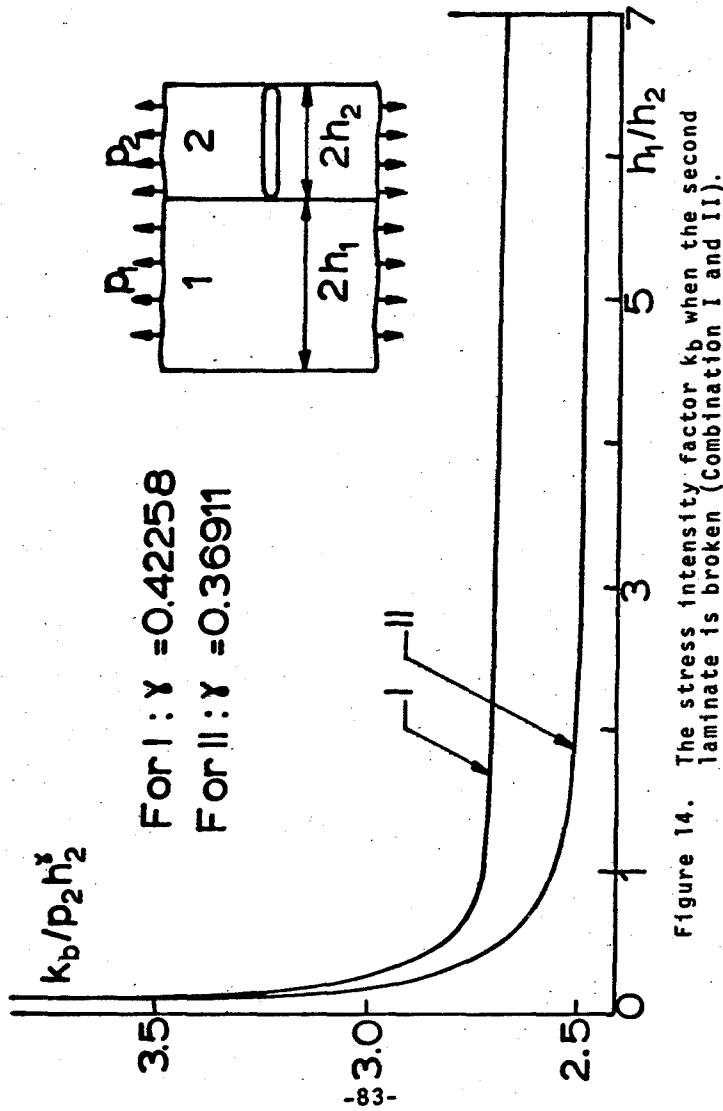


Figure 14. The stress intensity factor  $k_b$  when the second laminate is broken (Combination I and II).

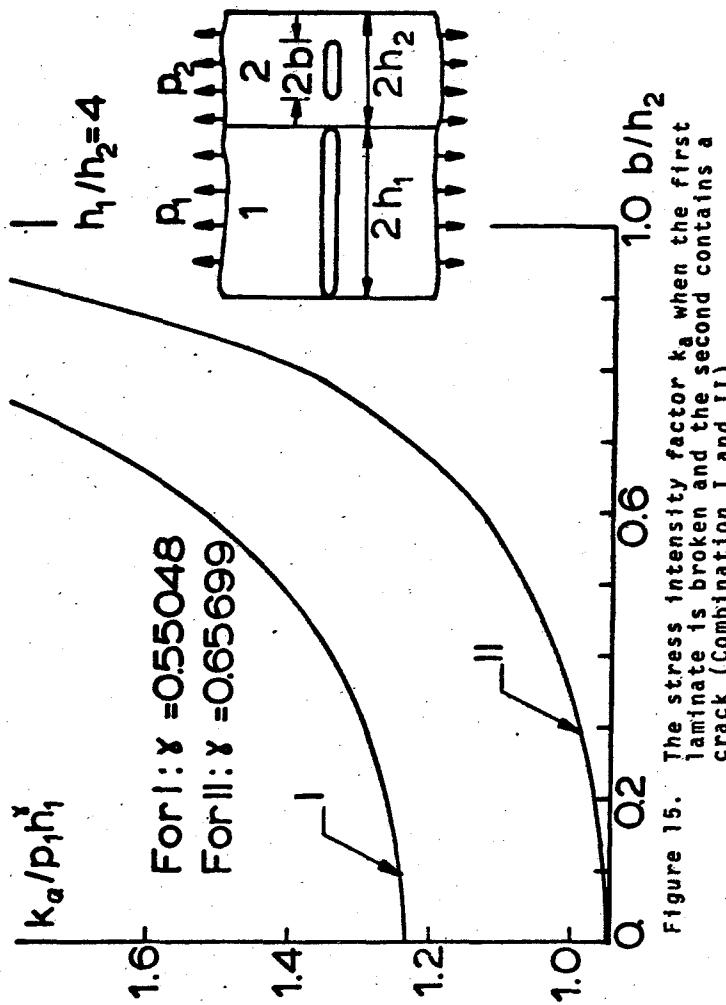


Figure 15. The stress intensity factor  $k_a$  when the first laminate is broken and the second contains a crack (Combination I and II).

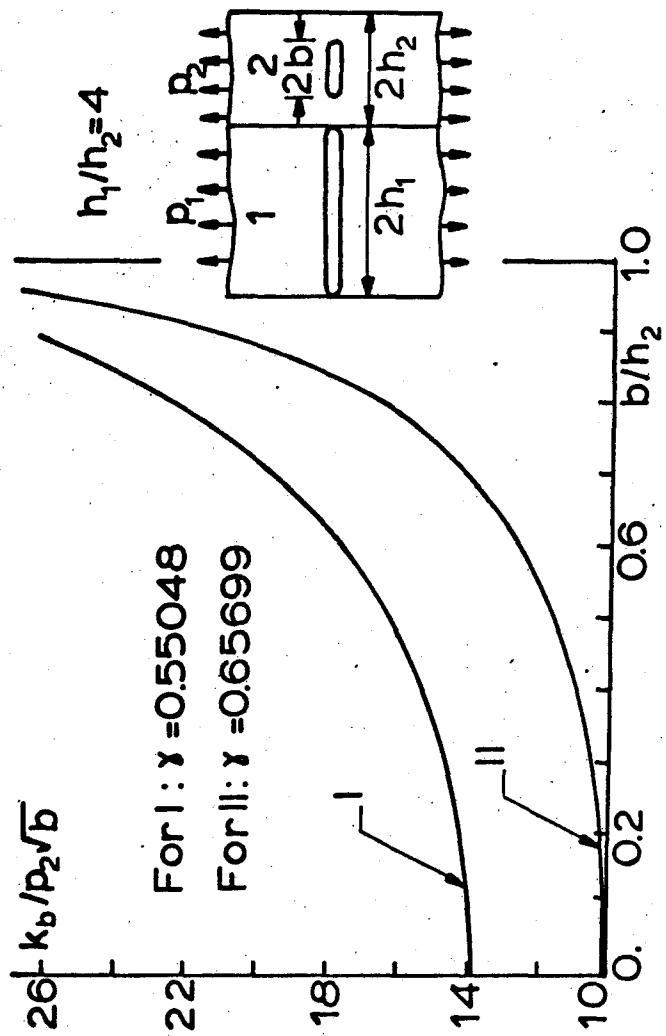


Figure 16. The stress intensity factor  $k_b$  when the first laminate is broken and the second contains a crack (Combination I and II).

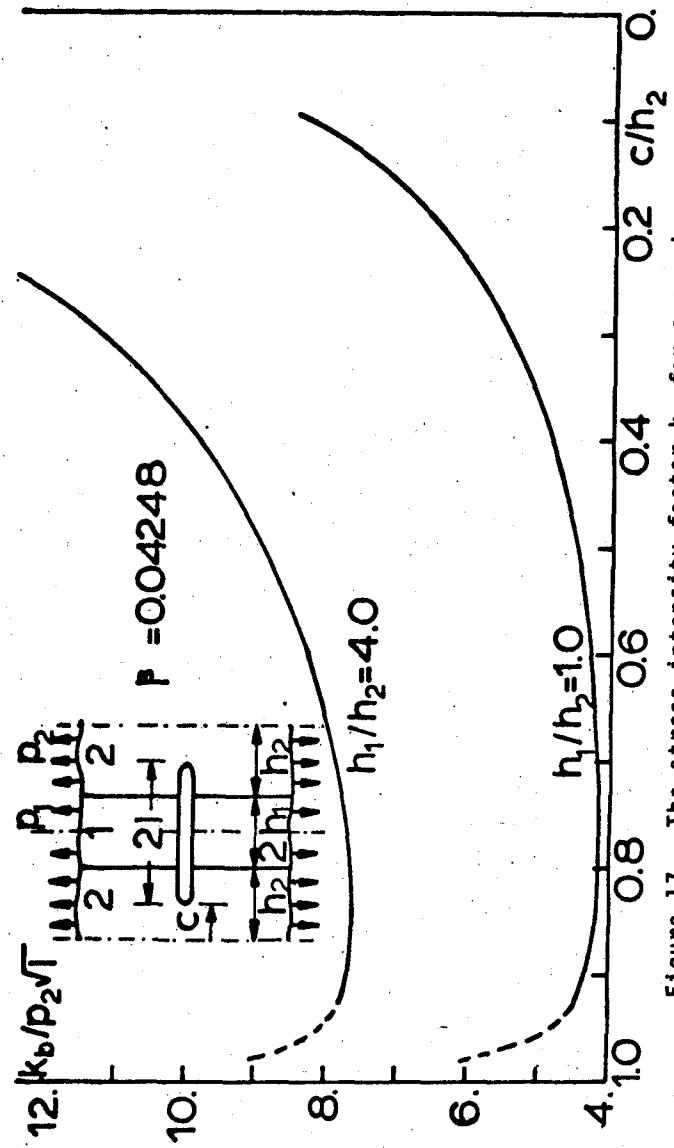


Figure 17. The stress intensity factor  $k_b$  for a crack crossing the interface (Singular behavior at the interface).

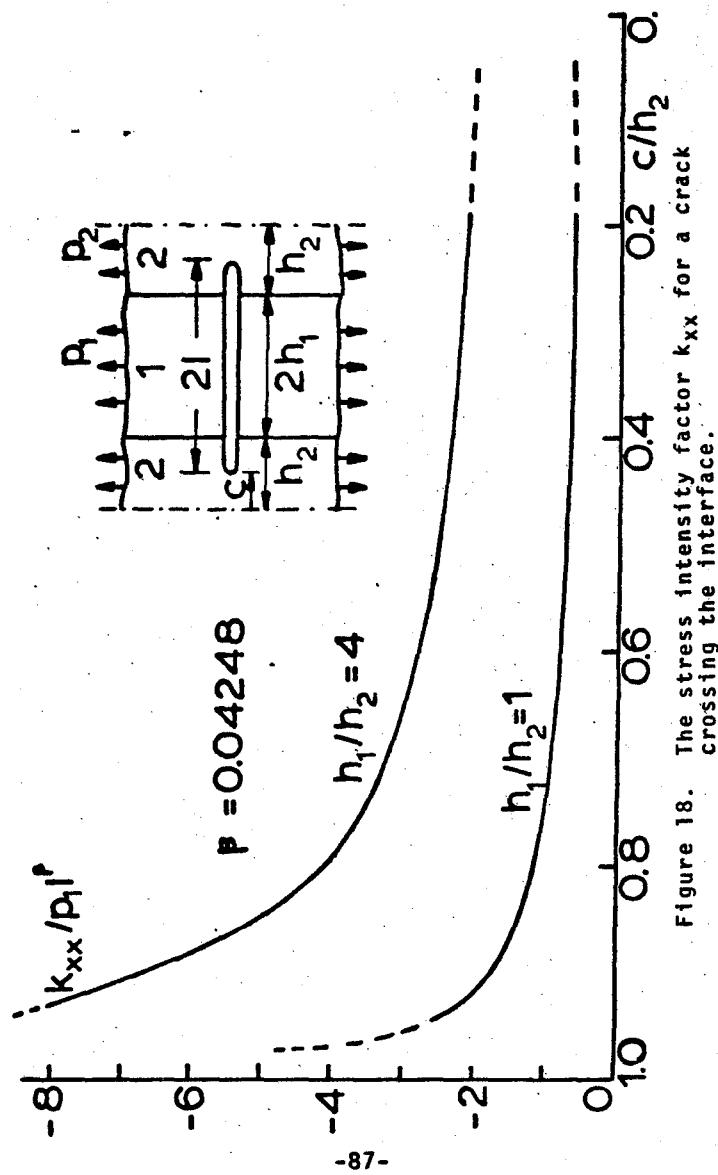


Figure 18. The stress intensity factor  $k_{xx}$  for a crack crossing the interface.

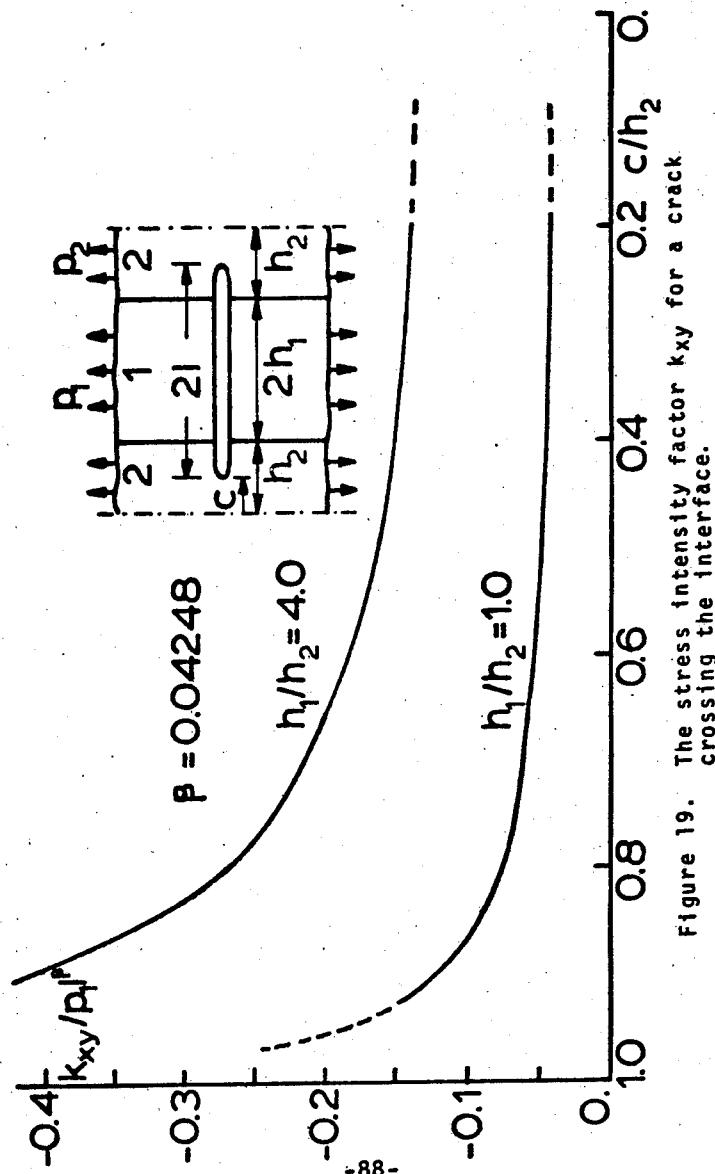


Figure 19. The stress intensity factor  $K_{xy}$  for a crack crossing the interface.

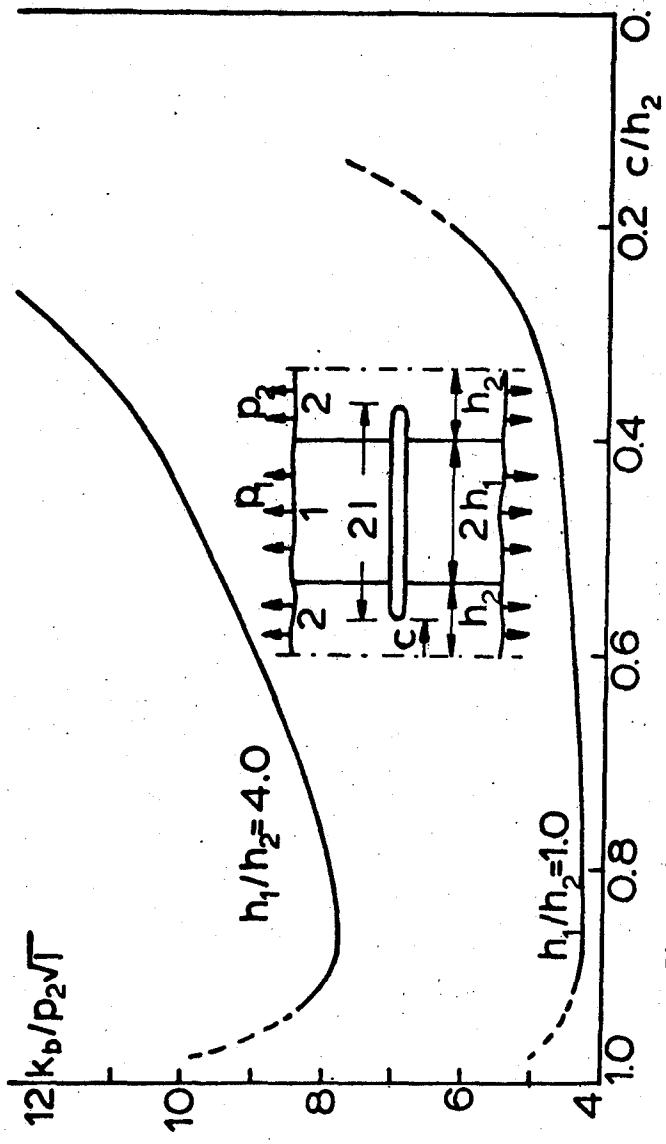


Figure 20. The stress intensity factor  $k_b$  for a crack crossing the interface (Non-singular behavior at the interface).

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## APPENDIX A

**Definitions of the material constants:**

A superscript \* will be used for the material in the second strip. The constants  $\beta_i$  ( $i = 1, \dots, 10$ ) and  $w_j$  ( $j = 1, \dots, 4$ ) are given by equations (2.15), (3.5), and (3.8).

$$\gamma_1 = 1 + \frac{v_{yx} w_1 \beta_9}{\sqrt{\beta_5}}$$

$$\gamma_{12} = - \frac{|w_3|}{\sqrt{\beta_5}} + \text{sign}(w_3) \beta_{10}$$

$$\gamma_2 = 1 + \frac{v_{yx} w_3 \beta_{10}}{\sqrt{\beta_5}}$$

$$\gamma_{13} = \text{sign}(w_1) \beta_9 - \frac{\gamma_{11}}{\gamma_{12}} \text{sign}(w_3) \beta_{10}$$

$$\gamma_3 = w_1 + v_{yx} \beta_7$$

$$\gamma_{14} = \frac{1}{2} \left[ \frac{\gamma_5}{\gamma_{13}} - \frac{\gamma_6}{\gamma_{13}} \frac{\gamma_{11}}{\gamma_{12}} \right]$$

$$\gamma_4 = w_3 + v_{yx} \beta_8$$

$$\lambda_1 = \frac{E_x^*}{E_x} \frac{1 - v_{xy}^* v_{yx}^*}{1 - v_{xy}^* v_{yx}^*}$$

$$\gamma_5 = v_{xy} + \frac{\beta_9 w_1}{\sqrt{\beta_5}}$$

$$\lambda_2 = \frac{G_{xy}^*}{G_{xy}}$$

$$\gamma_6 = v_{xy} + \frac{\beta_{10} w_3}{\sqrt{\beta_5}}$$

$$\lambda_3 = \beta_8 \gamma_3 - \beta_7 \gamma_4$$

$$\gamma_7 = v_{xy} w_1 + \beta_7$$

$$\lambda_4 = \lambda_1 \gamma_3^* \beta_7^* - \beta_7^* \gamma_3^*$$

$$\gamma_8 = v_{xy} w_3 + \beta_8$$

$$\lambda_5 = \lambda_1 \gamma_4^* \beta_7^* - \beta_8^* \gamma_3^*$$

$$\gamma_9 = -1 + \beta_7 w_1$$

$$\lambda_6 = \gamma_9 - \gamma_{10}$$

$$\gamma_{10} = -1 + \beta_8 w_3$$

$$\lambda_7 = \gamma_9 - \gamma_9 \lambda_2$$

$$\gamma_{11} = - \frac{|w_1|}{\sqrt{\beta_5}} + \text{sign}(w_1) \beta_9$$

$$\lambda_8 = \gamma_9 - \gamma_{10} \lambda_2$$

$$\lambda_9 = \lambda_3 \lambda_7$$

$$\lambda_{24} = \lambda_{10} \lambda_{16} - \lambda_9 \lambda_{16}$$

$$\lambda_{10} = \lambda_3 \lambda_8$$

$$\lambda_{25} = \lambda_{10} \lambda_{17}$$

$$\lambda_{11} = \lambda_3 \gamma_9$$

$$\lambda_{26} = \lambda_{12} \lambda_{16} - \lambda_{10} \lambda_{18}$$

$$\lambda_{12} = \lambda_6 \lambda_4$$

$$\lambda_{27} = \lambda_9 \lambda_{20} - \lambda_3 \lambda_{16}$$

$$\lambda_{13} = \lambda_5 \lambda_6$$

$$\lambda_{28} = \lambda_9 \lambda_{19}$$

$$\lambda_{14} = \lambda_6 \gamma_3$$

$$\lambda_{29} = \lambda_{12} \lambda_{19} - \lambda_{13} \lambda_{17}$$

$$\lambda_{15} = \lambda_6 \beta_7$$

$$\lambda_{30} = \lambda_{13} \lambda_{18} - \lambda_{12} \lambda_{20} = 0$$

$$\lambda_{16} = \lambda_3 \beta_7$$

$$\lambda_{31} = \lambda_{11} \lambda_{16} - \lambda_9 \lambda_{16}$$

$$\lambda_{17} = \lambda_3 \beta_7^* + \lambda_4 \beta_8$$

$$\lambda_{32} = \lambda_{11} \lambda_{17}$$

$$\lambda_{18} = \lambda_4 \beta_7$$

$$\lambda_{33} = \lambda_{16} \lambda_{12} - \lambda_{11} \lambda_{18}$$

$$\lambda_{19} = \lambda_3 \beta_8^* + \lambda_5 \beta_8$$

$$\lambda_{34} = \lambda_9 \lambda_{22} - \lambda_{14} \lambda_{16}$$

$$\lambda_{20} = \lambda_5 \beta_7$$

$$\lambda_{35} = \lambda_9 \lambda_{21}$$

$$\lambda_{21} = \lambda_3 - \gamma_3 \beta_8$$

$$\lambda_{36} = -\lambda_{14} \lambda_{17} - \lambda_{21} \lambda_{12}$$

$$\lambda_{22} = \gamma_3 \beta_7$$

$$\lambda_{37} = \lambda_{14} \lambda_{18} - \lambda_{22} \lambda_{12} = 0$$

$$\lambda_{23} = \beta_7 \beta_8$$

$$\lambda_{38} = \lambda_{15} \lambda_{16} - \beta_7^2 \lambda_9$$

$$\lambda_{39} = \lambda_9 \lambda_{23}$$

$$\lambda_{54} = -\lambda_{10} \lambda_{23}$$

$$\lambda_{40} = \lambda_{15} \lambda_{17} - \lambda_{12} \lambda_{23}$$

$$\lambda_{55} = \lambda_{13} \lambda_{23} - \lambda_{15} \lambda_{19}$$

$$\lambda_{41} = \beta_7^2 \lambda_{12} - \lambda_{15} \lambda_{18} = 0$$

$$\lambda_{56} = -\lambda_3 \lambda_{20}$$

$$\lambda_{42} = \lambda_3 \lambda_{16}$$

$$\lambda_{57} = \lambda_3 \lambda_{19}$$

$$\lambda_{43} = \lambda_3 \lambda_{17}$$

$$\lambda_{58} = -\lambda_4 \lambda_{16} (\lambda_8 - \gamma_9)$$

$$\lambda_{44} = \lambda_3 \lambda_{18}$$

$$\lambda_{59} = -\lambda_5 \lambda_{16} (\gamma_9 - \lambda_7)$$

$$\lambda_{45} = \lambda_{10} \lambda_{16} - \lambda_{11} \lambda_{16}$$

$$\lambda_{60} = \gamma_9 (\lambda_4 \lambda_{19} - \lambda_5 \lambda_{17})$$

$$\lambda_{46} = \lambda_{11} \lambda_{20} - \lambda_{13} \lambda_{16}$$

$$\lambda_{61} = \gamma_3 \beta_7 (\lambda_{10} - \lambda_9)$$

$$\lambda_{47} = -\lambda_{11} \lambda_{19}$$

$$\lambda_{62} = \lambda_8 (\gamma_3 \lambda_{17} + \lambda_4 \lambda_{21})$$

$$\lambda_{48} = \lambda_{14} \lambda_{16} - \lambda_{10} \lambda_{22}$$

$$\lambda_{63} = -\lambda_9 (\gamma_3 \beta_8^* + \lambda_5)$$

$$\lambda_{49} = \lambda_{13} \lambda_{22} - \lambda_{14} \lambda_{20} = 0$$

$$\lambda_{64} = -\beta_7^2 (\lambda_{10} - \lambda_9)$$

$$\lambda_{50} = -\lambda_{10} \lambda_{21}$$

$$\lambda_{65} = \lambda_8 (\lambda_4 \lambda_{23} - \beta_7 \lambda_{17})$$

$$\lambda_{51} = \lambda_{14} \lambda_{19} + \lambda_{13} \lambda_{21}$$

$$\lambda_{66} = \lambda_7 (\beta_7 \lambda_{19} - \lambda_5 \lambda_{23})$$

$$\lambda_{52} = \lambda_{10} \beta_7^2 - \lambda_{15} \lambda_{16}$$

$$\lambda_{67} = -\lambda_4 \lambda_{16}$$

$$\lambda_{53} = \lambda_{15} \lambda_{20} - \lambda_{13} \beta_7^2 = 0$$

$$\lambda_{68} = \lambda_5 \lambda_{16}$$

$$\lambda_{69} = \lambda_5 \lambda_{17} - \lambda_4 \lambda_{19}$$

$$\lambda_{70} = \beta_3^* [\beta_7^* (\lambda_{10} - \lambda_{11}) + \beta_8 \lambda_4 (\lambda_8 - \gamma_9)]$$

$$\lambda_{71} = \beta_3^* [\beta_8^* (\lambda_{11} - \lambda_9) + \beta_8 \lambda_5 (\gamma_9 - \lambda_7)]$$

$$\lambda_{72} = \beta_7^* (\lambda_{11} \lambda_5 - \lambda_3 \lambda_{13}) + \beta_8^* (\lambda_3 \lambda_{12} - \lambda_{11} \lambda_4)$$

$$\lambda_{73} = \lambda_{21} (\lambda_{10} - \lambda_9)$$

$$\lambda_{74} = \lambda_{12} \lambda_3 - \lambda_{10} \lambda_4 + \beta_7^* (\lambda_{14} \lambda_3 - \lambda_{10} \gamma_3)$$

$$\lambda_{75} = \lambda_9 \lambda_5 - \lambda_{13} \lambda_3 + \beta_8^* (\lambda_9 \gamma_3 - \lambda_{14} \lambda_3)$$

$$\lambda_{76} = \lambda_{23} (\lambda_{10} - \lambda_9)$$

$$\lambda_{77} = \beta_7^* (\lambda_{10} \beta_7 - \lambda_{15} \lambda_3)$$

$$\lambda_{78} = \beta_8^* (\lambda_{15} \lambda_3 - \beta_7 \lambda_9)$$

$$\lambda_{79} = \lambda_3 (\lambda_4 \beta_8^* - \lambda_5 \beta_7^*)$$

$$\lambda_{80} = \lambda_{24} + \lambda_{25} + \lambda_{26} + \lambda_{27} - \lambda_{28} + \lambda_{29}$$

$$\lambda_{81} = -\lambda_{70} - \lambda_{71} - \lambda_{72} + \text{sign}(w_1) \beta_9 (\lambda_{73} + \lambda_{74} + \lambda_{75})$$

$$+ \frac{\sqrt{\beta_s} \gamma_1}{|w_1|} (\lambda_{76} + \lambda_{77} + \lambda_{78}) - \frac{\gamma_{11} \sqrt{\beta_s}}{|w_1|} (\lambda_{43} - \lambda_{57} + \lambda_{79})$$

$$\lambda_{82} = -\lambda_{58} - \lambda_{59} - \lambda_{60} + \text{sign}(w_1) \beta_9 (\lambda_{61} + \lambda_{62} + \lambda_{63})$$

$$+ \frac{\sqrt{\beta_5} \gamma_1}{|w_1|} (\lambda_{64} + \lambda_{65} + \lambda_{66}) - \frac{\gamma_{11} \sqrt{\beta_5}}{|w_1|} (\lambda_{67} + \lambda_{68} + \lambda_{69})$$

$$\lambda_{83} = \frac{\gamma_{11}}{\gamma_{12}} (\lambda_{70} + \lambda_{71} + \lambda_{72}) - \frac{\gamma_{11}}{\gamma_{12}} \text{sign}(w_3) \beta_{10} (\lambda_{73} + \lambda_{74} + \lambda_{75})$$

$$- \sqrt{\beta_5} \frac{\gamma_{11}}{\gamma_{12} |w_3|} (\lambda_{76} + \lambda_{77} + \lambda_{78}) + \frac{\gamma_{11} \sqrt{\beta_5}}{|w_3|} (\lambda_{43} - \lambda_{57} + \lambda_{79})$$

$$\lambda_{84} = \frac{\gamma_{11}}{\gamma_{12}} (\lambda_{58} + \lambda_{59} + \lambda_{60}) - \frac{\gamma_{11}}{\gamma_{12}} \text{sign}(w_3) \beta_{10} (\lambda_{61} + \lambda_{62} + \lambda_{63})$$

$$- \sqrt{\beta_5} \frac{\gamma_{11}}{\gamma_{12} |w_3|} (\lambda_{64} + \lambda_{65} + \lambda_{66}) + \frac{\gamma_{11} \sqrt{\beta_5}}{|w_3|} (\lambda_{67} + \lambda_{68} + \lambda_{69})$$

$$\lambda_{85} = \frac{\gamma_7 \lambda_{81}}{\gamma_{13} \gamma_{14} \lambda_{80}}$$

$$\lambda_{86} = \frac{\gamma_8 \lambda_{82}}{\gamma_{13} \gamma_{14} \lambda_{80}}$$

$$\lambda_{87} = \frac{\gamma_7 \lambda_{83}}{\gamma_{13} \gamma_{14} \lambda_{80}}$$

$$\lambda_{88} = \frac{\gamma_8 \lambda_{84}}{\gamma_{13} \gamma_{14} \lambda_{80}}$$

$$\lambda_{89} = -\lambda_{70} - \lambda_{71} - \lambda_{72} - \text{sign}(w_1^*) \beta_9^* (\lambda_{73} + \lambda_{74} + \lambda_{75})$$

$$- \lambda_1 \sqrt{\beta_5^*} \frac{\gamma_1^*}{|w_1^*|} (\lambda_{76} + \lambda_{77} + \lambda_{78}) - \lambda_2 \frac{\gamma_{11}^* \sqrt{\beta_5^*}}{|w_1^*|} (\lambda_{43} - \lambda_{57} + \lambda_{79})$$

$$\lambda_{90} = -\lambda_{58} - \lambda_{59} - \lambda_{60} - \text{sign}(w_1^*) \beta_9^* (\lambda_{61} + \lambda_{62} + \lambda_{63})$$

$$- \lambda_1 \sqrt{\beta_5^*} \frac{\gamma_1^*}{|w_1^*|} (\lambda_{64} + \lambda_{65} + \lambda_{66}) - \lambda_2 \frac{\gamma_{11}^* \sqrt{\beta_5^*}}{|w_1^*|} (\lambda_{67} + \lambda_{68} + \lambda_{69})$$

$$\lambda_{91} = \frac{\gamma_{11}^*}{\gamma_{12}^*} (\lambda_{70} + \lambda_{71} + \lambda_{72}) + \frac{\gamma_{11}^*}{\gamma_{12}^*} \text{sign}(w_3^*) \beta_{10}^* (\lambda_{73} + \lambda_{74} + \lambda_{75}) \\ + \sqrt{\beta_s^*} \lambda_1 \frac{\gamma_{11}^*}{\gamma_{12}^*} \frac{\gamma_2^*}{|w_3^*|} (\lambda_{76} + \lambda_{77} + \lambda_{78}) + \frac{\gamma_{11}^* \lambda_2 \sqrt{\beta_s^*}}{|w_3^*|} (\lambda_{43} - \lambda_{57} + \lambda_{79})$$

$$\lambda_{92} = \frac{\gamma_{11}^*}{\gamma_{12}^*} (\lambda_{58} + \lambda_{59} + \lambda_{60}) + \frac{\gamma_{11}^*}{\gamma_{12}^*} \text{sign}(w_3^*) \beta_{10}^* (\lambda_{61} + \lambda_{62} + \lambda_{63}) \\ + \lambda_1 \sqrt{\beta_s^*} \frac{\gamma_{11}^*}{\gamma_{12}^*} \frac{\gamma_2^*}{|w_3^*|} (\lambda_{64} + \lambda_{65} + \lambda_{66}) + \frac{\gamma_{11}^* \lambda_2 \sqrt{\beta_s^*}}{|w_3^*|} (\lambda_{67} + \lambda_{68} + \lambda_{69})$$

$$\lambda_{93} = \frac{\gamma_7 \lambda_{89}}{\gamma_{13}^* \gamma_{14} \lambda_{80}}$$

$$\lambda_{94} = \frac{\gamma_8 \lambda_{90}}{\gamma_{13}^* \gamma_{14} \lambda_{80}}$$

$$\lambda_{95} = \frac{\gamma_7 \lambda_{91}}{\gamma_{13}^* \gamma_{14} \lambda_{80}}$$

$$\lambda_{96} = \frac{\gamma_8 \lambda_{92}}{\gamma_{13}^* \gamma_{14} \lambda_{80}}$$

$$\lambda_{97} = -\lambda_{45} - \lambda_{46} - \lambda_{47} + \text{sign}(w_1) \beta_9 (\lambda_{48} + \lambda_{50} + \lambda_{51})$$

$$+ \sqrt{\beta_s^*} \frac{\gamma_1}{|w_1|} (\lambda_{52} + \lambda_{54} + \lambda_{55}) - \frac{\gamma_{11} \sqrt{\beta_s^*}}{|w_1|} (\lambda_{42} + \lambda_{56} + \lambda_{57})$$

$$\lambda_{98} = -\lambda_{31} - \lambda_{32} - \lambda_{33} + \text{sign}(w_1) \beta_9 (\lambda_{34} + \lambda_{35} + \lambda_{36}) \\ + \sqrt{\beta_s^*} \frac{\gamma_1}{|w_1|} (\lambda_{38} + \lambda_{39} + \lambda_{40}) - \frac{\gamma_{11} \sqrt{\beta_s^*}}{|w_1|} (-\lambda_{42} - \lambda_{43} + \lambda_{44})$$

$$\lambda_{99} = \frac{\gamma_{11}}{\gamma_{12}} (\lambda_{45} + \lambda_{46} + \lambda_{47}) - \frac{\gamma_{11}}{\gamma_{12}} \text{sign}(w_3) \beta_{10} (\lambda_{48} + \lambda_{50} + \lambda_{51}) \\ - \sqrt{\beta_s^*} \frac{\gamma_{11}}{\gamma_{12}} \frac{\gamma_2}{|w_3|} (\lambda_{52} + \lambda_{54} + \lambda_{55}) + \frac{\gamma_{11} \sqrt{\beta_s^*}}{|w_3|} (\lambda_{42} + \lambda_{56} + \lambda_{57})$$

$$\lambda_{100} = \frac{\gamma_{11}}{\gamma_{12}} (\lambda_{31} + \lambda_{32} + \lambda_{33}) - \frac{\gamma_{11}}{\gamma_{12}} \text{sign}(w_3) \beta_{10} (\lambda_{34} + \lambda_{35} + \lambda_{36})$$

$$- \sqrt{\beta_5} \frac{\gamma_{11}}{\gamma_{12}} \frac{\gamma_2}{|w_3|} (\lambda_{38} + \lambda_{39} + \lambda_{40}) + \frac{\gamma_{11} \sqrt{\beta_5}}{|w_3|} (-\lambda_{42} - \lambda_{43} + \lambda_{44})$$

$$\lambda_{101} = \frac{\lambda_{97} \gamma_7^*}{\gamma_{13} \gamma_{14}^* \lambda_{80}}$$

$$\lambda_{102} = \frac{\lambda_{98} \gamma_8^*}{\gamma_{13} \gamma_{14}^* \lambda_{80}}$$

$$\lambda_{103} = \frac{\lambda_{99} \gamma_7^*}{\gamma_{13} \gamma_{14}^* \lambda_{80}}$$

$$\lambda_{104} = \frac{\lambda_{100} \gamma_8^*}{\gamma_{13} \gamma_{14}^* \lambda_{80}}$$

$$\lambda_{105} = -\lambda_{45} - \lambda_{46} - \lambda_{47} - \text{sign}(w_1^*) \beta_9^* (\lambda_{48} + \lambda_{50} + \lambda_{51})$$

$$- \lambda_1 \sqrt{\beta_5^*} \frac{\gamma_1^*}{|w_1^*|} (\lambda_{52} + \lambda_{54} + \lambda_{55}) - \lambda_2 \frac{\gamma_{11}^* \sqrt{\beta_5^*}}{|w_1^*|} (\lambda_{42} + \lambda_{56} + \lambda_{57})$$

$$\lambda_{106} = -\lambda_{31} - \lambda_{32} - \lambda_{33} - \text{sign}(w_1^*) \beta_9^* (\lambda_{34} + \lambda_{35} + \lambda_{36})$$

$$- \lambda_1 \sqrt{\beta_5^*} \frac{\gamma_1^*}{|w_1^*|} (\lambda_{38} + \lambda_{39} + \lambda_{40}) - \lambda_2 \frac{\gamma_{11}^*}{|w_1^*|} \sqrt{\beta_5^*} (-\lambda_{42} - \lambda_{43} + \lambda_{44})$$

$$\lambda_{107} = \frac{\gamma_{11}^*}{\gamma_{12}^*} (\lambda_{45} + \lambda_{46} + \lambda_{47}) + \frac{\gamma_{11}^*}{\gamma_{12}^*} \text{sign}(w_3^*) \beta_{10}^* (\lambda_{48} + \lambda_{50} + \lambda_{51})$$

$$+ \sqrt{\beta_5^*} \lambda_1 \frac{\gamma_{11}^*}{\gamma_{12}^*} \frac{\gamma_2^*}{|w_3^*|} (\lambda_{52} + \lambda_{54} + \lambda_{55}) + \frac{\gamma_{11}^* \lambda_2 \sqrt{\beta_5^*}}{|w_3^*|} (\lambda_{42} + \lambda_{56} + \lambda_{57})$$

$$\lambda_{108} = \frac{\gamma_{11}^*}{\gamma_{12}^*} (\lambda_{31} + \lambda_{32} + \lambda_{33}) + \frac{\gamma_{11}^*}{\gamma_{12}^*} \text{sign}(w_3^*) \beta_{10}^* (\lambda_{34} + \lambda_{35} + \lambda_{36})$$

$$+ \lambda_1 \sqrt{\beta_5^*} \frac{\gamma_{11}^*}{\gamma_{12}^*} \frac{\gamma_2^*}{|w_3^*|} (\lambda_{38} + \lambda_{39} + \lambda_{40}) + \frac{\gamma_{11}^* \lambda_2 \sqrt{\beta_5^*}}{|w_3^*|} (-\lambda_{42} - \lambda_{43} + \lambda_{44})$$

$$\lambda_{109} = \frac{\lambda_{105} \gamma_7^*}{\gamma_{13}^* \gamma_{14}^* \lambda_{80}}$$

$$\lambda_{110} = \frac{\lambda_{106} \gamma_8^*}{\gamma_{13}^* \gamma_{14}^* \lambda_{80}}$$

$$\lambda_{111} = \frac{\lambda_{107} \gamma_7^*}{\gamma_{13}^* \gamma_{14}^* \lambda_{80}}$$

$$\lambda_{112} = \frac{\lambda_{108} \gamma_8^*}{\gamma_{13}^* \gamma_{14}^* \lambda_{80}}$$

$$\lambda_{113} = \frac{|w_1|}{\sqrt{\beta_5}} \frac{|w_1^*|}{\sqrt{\beta_5^*}} \lambda_{85} \lambda_{109} - \frac{|w_1|}{\sqrt{\beta_5}} \frac{|w_1^*|}{\sqrt{\beta_5^*}} \lambda_{93} \lambda_{101}$$

$$\lambda_{114} = \frac{|w_1|}{\sqrt{\beta_5}} \frac{|w_1^*|}{\sqrt{\beta_5^*}} \lambda_{85} \lambda_{110} + \frac{|w_1|}{\sqrt{\beta_5}} \frac{|w_3|}{\sqrt{\beta_5^*}} \lambda_{85} \lambda_{111}$$

$$- \frac{|w_1|}{\sqrt{\beta_5}} \frac{|w_1^*|}{\sqrt{\beta_5^*}} \lambda_{93} \lambda_{102} - \frac{|w_1|}{\sqrt{\beta_5}} \frac{|w_3|}{\sqrt{\beta_5^*}} \lambda_{95} \lambda_{101}$$

$$\lambda_{115} = \frac{|w_1|}{\sqrt{\beta_5}} \frac{|w_3|}{\sqrt{\beta_5^*}} \lambda_{85} \lambda_{112} - \frac{|w_1|}{\sqrt{\beta_5}} \frac{|w_3|}{\sqrt{\beta_5^*}} \lambda_{95} \lambda_{102}$$

$$\lambda_{116} = \frac{|w_1|}{\sqrt{\beta_5}} \frac{|w_1^*|}{\sqrt{\beta_5^*}} \lambda_{86} \lambda_{109} + \frac{|w_3|}{\sqrt{\beta_5}} \frac{|w_1^*|}{\sqrt{\beta_5^*}} \lambda_{87} \lambda_{109}$$

$$- \frac{|w_3|}{\sqrt{\beta_5}} \frac{|w_1^*|}{\sqrt{\beta_5^*}} \lambda_{93} \lambda_{103} - \frac{|w_1|}{\sqrt{\beta_5}} \frac{|w_1^*|}{\sqrt{\beta_5^*}} \lambda_{94} \lambda_{101}$$

$$\lambda_{117} = \frac{|w_1|}{\sqrt{\beta_5}} \frac{|w_1^*|}{\sqrt{\beta_5^*}} \lambda_{86} \lambda_{110} + \frac{|w_1|}{\sqrt{\beta_5}} \frac{|w_3|}{\sqrt{\beta_5^*}} \lambda_{86} \lambda_{111}$$

$$+ \frac{|w_3|}{\sqrt{\beta_5}} \frac{|w_1^*|}{\sqrt{\beta_5^*}} \lambda_{87} \lambda_{110} + \frac{|w_3|}{\sqrt{\beta_5}} \frac{|w_3|}{\sqrt{\beta_5^*}} \lambda_{87} \lambda_{111}$$

$$- \frac{|w_3|}{\sqrt{\beta_5}} \frac{|w_1^*|}{\sqrt{\beta_5^*}} \lambda_{93} \lambda_{104} - \frac{|w_1|}{\sqrt{\beta_5}} \frac{|w_1^*|}{\sqrt{\beta_5^*}} \lambda_{94} \lambda_{102}$$

$$- \frac{|w_3|}{\sqrt{\beta_5}} \frac{|w_3|}{\sqrt{\beta_5^*}} \lambda_{95} \lambda_{103} - \frac{|w_1|}{\sqrt{\beta_5}} \frac{|w_3|}{\sqrt{\beta_5^*}} \lambda_{96} \lambda_{101}$$

$$\begin{aligned}
 \lambda_{118} &= \frac{|W_3|}{\sqrt{\beta_5}} \frac{|W_3^*|}{\sqrt{\beta_5^*}} \lambda_{87} \lambda_{112} + \frac{|W_1|}{\sqrt{\beta_5}} \frac{|W_3^*|}{\sqrt{\beta_5^*}} \lambda_{86} \lambda_{112} \\
 &\quad - \frac{|W_3|}{\sqrt{\beta_5}} \frac{|W_3^*|}{\sqrt{\beta_5^*}} \lambda_{95} \lambda_{104} - \frac{|W_1|}{\sqrt{\beta_5}} \frac{|W_3^*|}{\sqrt{\beta_5^*}} \lambda_{96} \lambda_{102} \\
 \lambda_{119} &= \frac{|W_3|}{\sqrt{\beta_5}} \frac{|W_1^*|}{\sqrt{\beta_5^*}} \lambda_{88} \lambda_{109} - \frac{|W_3|}{\sqrt{\beta_5}} \frac{|W_1^*|}{\sqrt{\beta_5^*}} \lambda_{94} \lambda_{103} \\
 \lambda_{120} &= \frac{|W_3|}{\sqrt{\beta_5}} \frac{|W_1^*|}{\sqrt{\beta_5^*}} \lambda_{88} \lambda_{110} + \frac{|W_3|}{\sqrt{\beta_5}} \frac{|W_1^*|}{\sqrt{\beta_5^*}} \lambda_{88} \lambda_{111} \\
 &\quad - \frac{|W_3|}{\sqrt{\beta_5}} \frac{|W_1^*|}{\sqrt{\beta_5^*}} \lambda_{94} \lambda_{104} - \frac{|W_3|}{\sqrt{\beta_5}} \frac{|W_1^*|}{\sqrt{\beta_5^*}} \lambda_{96} \lambda_{103} \\
 \lambda_{121} &= \frac{|W_3|}{\sqrt{\beta_5}} \frac{|W_3^*|}{\sqrt{\beta_5^*}} \lambda_{88} \lambda_{112} - \frac{|W_3|}{\sqrt{\beta_5}} \frac{|W_3^*|}{\sqrt{\beta_5^*}} \lambda_{96} \lambda_{104}
 \end{aligned}$$

APPENDIX B

Expressions of the functions used in equations (4.14) and (4.15):

$$R_1(\alpha) = -\frac{1}{2Y_{13}\alpha} \int_a^b \left[ e^{-(h_1-\eta)\alpha\sqrt{\beta_s}/|w_1|} - \frac{Y_{11}}{Y_{12}} e^{-(h_1-\eta)\alpha\sqrt{\beta_s}/|w_3|} \right] \phi(\eta) d\eta$$

$$- \frac{1}{2Y_{13}^*\alpha} \int_b^a \left[ e^{-(h_2-\eta)\alpha\sqrt{\beta_s^*}/|w_1^*|} - \frac{Y_{11}^*}{Y_{12}^*} e^{-(h_2-\eta)\alpha\sqrt{\beta_s^*}/|w_3^*|} \right] \phi^*(\eta) d\eta$$

$$R_2(\alpha) = \frac{1}{2Y_{13}\alpha} \int_a^b \left[ \text{sign}(w_1)\beta_9 e^{-(h_1-\eta)\alpha\sqrt{\beta_s}/|w_1|} \right.$$

$$\left. - \frac{Y_{11}}{Y_{12}} \text{sign}(w_3)\beta_{10} e^{-(h_1-\eta)\alpha\sqrt{\beta_s}/|w_3|} \right] \phi(\eta) d\eta$$

$$- \frac{1}{2Y_{13}^*\alpha} \int_b^a \left[ \text{sign}(w_1^*)\beta_9^* e^{-(h_2-\eta)\alpha\sqrt{\beta_s^*}/|w_1^*|} \right.$$

$$\left. - \frac{Y_{11}^*}{Y_{12}^*} \text{sign}(w_3^*)\beta_{10}^* e^{-(h_2-\eta)\alpha\sqrt{\beta_s^*}/|w_3^*|} \right] \phi^*(\eta) d\eta$$

$$R_3(\alpha) = \frac{\sqrt{\beta_s}}{2Y_{13}\alpha} \int_a^b \left[ \frac{Y_1}{|w_1|} e^{-(h_1-\eta)\alpha\sqrt{\beta_s}/|w_1|} \right.$$

$$\left. - \frac{Y_{11}}{Y_{12}} \frac{Y_2}{|w_3|} e^{-(h_1-\eta)\alpha\sqrt{\beta_s}/|w_3|} \right] \phi(\eta) d\eta$$

$$- \frac{\sqrt{\beta_s^*}}{2Y_{13}^*\alpha} \int_b^a \left[ \frac{Y_{11}^*}{|w_1^*|} e^{-(h_2-\eta)\alpha\sqrt{\beta_s^*}/|w_1^*|} \right.$$

$$\left. - \frac{Y_{11}^*}{Y_{12}^*} \frac{Y_{21}^*}{|w_3^*|} e^{-(h_2-\eta)\alpha\sqrt{\beta_s^*}/|w_3^*|} \right] \phi^*(\eta) d\eta$$

$$R_4(\alpha) = -\frac{\gamma_{11} \sqrt{\beta_s}}{2\gamma_{13}\alpha} \int_a^b \left[ \frac{1}{|w_1|} e^{-(h_1-\eta)\alpha\sqrt{\beta_s}/|w_1|} \right. \\ \left. - \frac{1}{|w_3|} e^{-(h_1-\eta)\alpha\sqrt{\beta_s}/|w_3|} \right] \phi(\eta) d\eta \\ - \frac{\gamma_{11}^* \sqrt{\beta_s^*} \lambda_2}{2\gamma_{13}^* \alpha} \int_{-b}^b \left[ \frac{1}{|w_1^*|} e^{-(h_2-\eta)\alpha\sqrt{\beta_s^*}/|w_1^*|} \right. \\ \left. - \frac{1}{|w_3^*|} e^{-(h_2-\eta)\alpha\sqrt{\beta_s^*}/|w_3^*|} \right] \phi^*(\eta) d\eta$$

$$f(\alpha) = \lambda_{24} f_1(\alpha) + \lambda_{25} f_2(\alpha) + \lambda_{26} f_3(\alpha) + \lambda_{27} f_4(\alpha) \\ - \lambda_{28} f_5(\alpha) + \lambda_{29} f_6(\alpha)$$

$$f_1(\alpha) = \tanh(w_1^* \alpha h_2) \tanh(w_3^* \alpha h_2)$$

$$f_2(\alpha) = \tanh(w_3^* \alpha h_2) \tanh(w_1 \alpha h_1)$$

$$f_3(\alpha) = \tanh(w_3^* \alpha h_2) \tanh(w_3 \alpha h_1)$$

$$f_4(\alpha) = \tanh(w_1^* \alpha h_2) \tanh(w_3 \alpha h_1)$$

$$f_5(\alpha) = \tanh(w_1^* \alpha h_2) \tanh(w_1 \alpha h_1)$$

$$f_6(\alpha) = \tanh(w_1 \alpha h_1) \tanh(w_3 \alpha h_1)$$

$$g(\alpha) = \lambda_{31} \tanh(w_1^* \alpha h_2) + \lambda_{32} \tanh(w_1 \alpha h_1) + \lambda_{33} \tanh(w_3 \alpha h_1)$$

$$h(\alpha) = \lambda_{34} f_4(\alpha) + \lambda_{35} f_5(\alpha) + \lambda_{36} f_6(\alpha)$$

$$m(\alpha) = \lambda_{38} f_4(\alpha) + \lambda_{39} f_5(\alpha) + \lambda_{40} f_6(\alpha)$$

$$n(\alpha) = -\lambda_{42} \tanh(w_1^* \alpha h_2) - \lambda_{43} \tanh(w_1 \alpha h_1) + \lambda_{44} \tanh(w_3 \alpha h_1)$$

$$g_0(\alpha) = \lambda_{45} \tanh(w_3^* \alpha h_2) + \lambda_{46} \tanh(w_3 \alpha h_1) + \lambda_{47} \tanh(w_1 \alpha h_1)$$

$$h_0(\alpha) = \lambda_{48} f_3(\alpha) + \lambda_{50} f_2(\alpha) + \lambda_{51} f_6(\alpha)$$

$$m_0(\alpha) = \lambda_{52} f_3(\alpha) + \lambda_{54} f_2(\alpha) + \lambda_{55} f_6(\alpha)$$

$$n_0(\alpha) = \lambda_{42} \tanh(w_3^* \alpha h_2) + \lambda_{56} \tanh(w_3 \alpha h_1) + \lambda_{57} \tanh(w_1 \alpha h_1)$$

$$g_1(\alpha) = \lambda_{58} \tanh(w_3^* \alpha h_2) + \lambda_{59} \tanh(w_1^* \alpha h_2) + \lambda_{60} \tanh(w_1 \alpha h_1)$$

$$h_1(\alpha) = \lambda_{61} f_1(\alpha) + \lambda_{62} f_2(\alpha) + \lambda_{63} f_5(\alpha)$$

$$m_1(\alpha) = \lambda_{64} f_1(\alpha) + \lambda_{65} f_2(\alpha) + \lambda_{66} f_5(\alpha)$$

$$n_1(\alpha) = \lambda_{67} \tanh(w_3^* \alpha h_2) + \lambda_{68} \tanh(w_1^* \alpha h_2) + \lambda_{69} \tanh(w_1 \alpha h_1)$$

$$g_2(\alpha) = \lambda_{70} \tanh(w_3^* \alpha h_2) + \lambda_{71} \tanh(w_1^* \alpha h_2) + \lambda_{72} \tanh(w_3 \alpha h_1)$$

$$h_2(\alpha) = \lambda_{73} f_1(\alpha) + \lambda_{74} f_3(\alpha) + \lambda_{75} f_4(\alpha)$$

$$m_2(\alpha) = \lambda_{76} f_1(\alpha) + \lambda_{77} f_3(\alpha) + \lambda_{78} f_4(\alpha)$$

$$n_2(\alpha) = \lambda_{43} \tanh(w_3^* \alpha h_2) - \lambda_{57} \tanh(w_1^* \alpha h_2) + \lambda_{79} \tanh(w_3 \alpha h_1)$$

Expressions of the functions  $k_j$  ( $j = 1, 8$ ) used in Eqs. (4.17):

$$k_1(x_1, \alpha) = \frac{1}{2\gamma_{13} f(\alpha)} \left[ \frac{\cosh(w_1 \alpha x_1)}{\cosh(w_1 \alpha h_1)} \gamma_7 \left[ -g_2(\alpha) + \text{sign}(w_1) \beta_9 h_2(\alpha) \right. \right. \\ \left. \left. + \sqrt{\beta_5} \frac{\gamma_1}{|w_1|} m_2(\alpha) - \frac{\gamma_{11} \sqrt{\beta_5}}{|w_1|} n_2(\alpha) \right] + \frac{\cosh(w_3 \alpha x_1)}{\cosh(w_3 \alpha h_1)} \gamma_8 \left[ -g_1(\alpha) \right. \right. \\ \left. \left. + \text{sign}(w_1) \beta_9 h_1(\alpha) + \sqrt{\beta_5} \frac{\gamma_1}{|w_1|} m_1(\alpha) - \frac{\gamma_{11} \sqrt{\beta_5}}{|w_1|} n_1(\alpha) \right] \right]$$

$$k_2(x_1, \alpha) = \frac{1}{2\gamma_{13}^* f(\alpha)} \left[ \frac{\cosh(w_1 \alpha x_1)}{\cosh(w_1 \alpha h_1)} \gamma_7 \left\{ g_2(\alpha) \frac{\gamma_{11}}{\gamma_{12}} - \frac{\gamma_{11}}{\gamma_{12}} \text{sign}(w_3) \beta_{10} h_2(\alpha) \right. \right. \\ - \sqrt{\beta_5} \frac{\gamma_{11}}{\gamma_{12}} \frac{\gamma_2}{|w_3|} m_2(\alpha) + \frac{\gamma_{11} \sqrt{\beta_5}}{|w_3|} n_2(\alpha) \Bigg\} + \frac{\cosh(w_3 \alpha x_1)}{\cosh(w_3 \alpha h_1)} \gamma_8 \left\{ \frac{\gamma_{11}}{\gamma_{12}} g_1(\alpha) \right. \\ \left. \left. - \frac{\gamma_{11}}{\gamma_{12}} \text{sign}(w_3) \beta_{10} h_1(\alpha) - \sqrt{\beta_5} \frac{\gamma_{11}}{\gamma_{12}} \frac{\gamma_2}{|w_3|} m_1(\alpha) + \frac{\gamma_{11} \sqrt{\beta_5}}{|w_3|} n_1(\alpha) \right\} \right]$$

$$k_3(x_1, \alpha) = \frac{1}{2\gamma_{13}^* f(\alpha)} \left[ \frac{\cosh(w_1 \alpha x_1)}{\cosh(w_1 \alpha h_1)} \gamma_7 \left\{ -g_2(\alpha) - \text{sign}(w_1^*) \beta_9^* h_2(\alpha) \right. \right. \\ - \lambda_1 \sqrt{\beta_5^*} \frac{\gamma_1^*}{|w_1^*|} m_2(\alpha) - \lambda_2 \frac{\gamma_{11}^* \sqrt{\beta_5^*}}{|w_1^*|} n_2(\alpha) \Bigg\} + \frac{\cosh(w_3 \alpha x_1)}{\cosh(w_3 \alpha h_1)} \gamma_8 \\ \left. \left. \left\{ -g_1(\alpha) - \text{sign}(w_1^*) \beta_9^* h_1(\alpha) - \lambda_1 \sqrt{\beta_5^*} \frac{\gamma_1^*}{|w_1^*|} m_1(\alpha) - \lambda_2 \frac{\gamma_{11}^* \sqrt{\beta_5^*}}{|w_1^*|} n_1(\alpha) \right\} \right] \right]$$

$$k_4(x_1, \alpha) = \frac{1}{2\gamma_{13}^* f(\alpha)} \left[ \frac{\cosh(w_1 \alpha x_1)}{\cosh(w_1 \alpha h_1)} \gamma_7 \left\{ g_2(\alpha) \frac{\gamma_{11}^*}{\gamma_{12}^*} + \frac{\gamma_{11}^*}{\gamma_{12}^*} \text{sign}(w_3^*) \beta_{10}^* h_2(\alpha) \right. \right. \\ + \sqrt{\beta_5^*} \lambda_1 \frac{\gamma_{11}^*}{\gamma_{12}^*} \frac{\gamma_2^*}{|w_3^*|} m_2(\alpha) + \frac{\gamma_{11}^* \lambda_2 \sqrt{\beta_5^*}}{|w_3^*|} n_2(\alpha) \Bigg\} + \frac{\cosh(w_3 \alpha x_1)}{\cosh(w_3 \alpha h_1)} \gamma_8 \\ \left. \left. \left\{ \frac{\gamma_{11}^*}{\gamma_{12}^*} g_1(\alpha) + \frac{\gamma_{11}^*}{\gamma_{12}^*} \text{sign}(w_3^*) \beta_{10}^* h_1(\alpha) + \lambda_1 \sqrt{\beta_5^*} \frac{\gamma_{11}^*}{\gamma_{12}^*} \frac{\gamma_2^*}{|w_3^*|} m_1(\alpha) \right. \right. \right. \\ \left. \left. \left. + \frac{\gamma_{11}^* \lambda_2 \sqrt{\beta_5^*}}{|w_3^*|} n_1(\alpha) \right\} \right] \right]$$

$$k_5(x_2, \alpha) = \frac{1}{2\gamma_{13}^* f(\alpha)} \left[ \frac{\cosh(w_1^* \alpha x_2)}{\cosh(w_1^* \alpha h_2)} \gamma_7^* \left\{ -g_0(\alpha) + \text{sign}(w_1) \beta_9 h_0(\alpha) \right. \right. \\ + \sqrt{\beta_5} \frac{\gamma_1}{|w_1|} m_0(\alpha) - \frac{\gamma_{11} \sqrt{\beta_5}}{|w_1|} n_0(\alpha) \Bigg\} + \frac{\cosh(w_3^* \alpha x_2)}{\cosh(w_3^* \alpha h_2)} \gamma_8^* \left\{ -g(\alpha) \right. \\ \left. \left. \left. \right\} \right]$$

$$+ \text{sign}(w_1) \beta_9 h(\alpha) + \sqrt{\beta_5} \frac{\gamma_1}{|w_1|} m(\alpha) - \frac{\gamma_{11} \sqrt{\beta_5}}{|w_1|} n(\alpha) \Bigg]$$

$$\begin{aligned} k_6(x_2, \alpha) = & \frac{1}{2\gamma_{13}^* f(\alpha)} \left[ \frac{\cosh(w_1^* \alpha x_2)}{\cosh(w_1^* \alpha h_2)} \gamma_7^* \left\{ g_0(\alpha) \frac{\gamma_{11}}{\gamma_{12}} - \frac{\gamma_{11}}{\gamma_{12}} \text{sign}(w_3) \beta_{10} h_0(\alpha) \right. \right. \\ & - \sqrt{\beta_5} \frac{\gamma_{11}}{\gamma_{12}} \frac{\gamma_2}{|w_3|} m_0(\alpha) + \frac{\gamma_{11} \sqrt{\beta_5}}{|w_3|} n_0(\alpha) \Bigg] + \frac{\cosh(w_3^* \alpha x_2)}{\cosh(w_3^* \alpha h_2)} \gamma_8^* \left\{ \frac{\gamma_{11}}{\gamma_{12}} g(\alpha) \right. \\ & \left. \left. - \frac{\gamma_{11}}{\gamma_{12}} \text{sign}(w_3) \beta_{10} h(\alpha) - \sqrt{\beta_5} \frac{\gamma_{11}}{\gamma_{12}} \frac{\gamma_2}{|w_3|} m(\alpha) + \frac{\gamma_{11} \sqrt{\beta_5}}{|w_3|} n(\alpha) \right\} \right] \end{aligned}$$

$$\begin{aligned} k_7(x_2, \alpha) = & \frac{1}{2\gamma_{13}^* f(\alpha)} \left[ \frac{\cosh(w_1^* \alpha x_2)}{\cosh(w_1^* \alpha h_2)} \gamma_7^* \left\{ -g_0(\alpha) - \text{sign}(w_1^*) \beta_9^* h_0(\alpha) \right. \right. \\ & - \lambda_1 \sqrt{\beta_5^*} \frac{\gamma_1^*}{|w_1^*|} m_0(\alpha) - \lambda_2 \frac{\gamma_{11}^* \sqrt{\beta_5^*}}{|w_1^*|} n_0(\alpha) \Bigg] + \frac{\cosh(w_3^* \alpha x_2)}{\cosh(w_3^* \alpha h_2)} \gamma_8^* \left\{ -g(\alpha) \right. \\ & \left. \left. - \text{sign}(w_1^*) \beta_9^* h(\alpha) - \lambda_1 \sqrt{\beta_5^*} \frac{\gamma_1^*}{|w_1^*|} m(\alpha) - \lambda_2 \frac{\gamma_{11}^*}{|w_1^*|} \sqrt{\beta_5^*} n(\alpha) \right\} \right] \end{aligned}$$

$$\begin{aligned} k_8(x_2, \alpha) = & \frac{1}{2\gamma_{13}^* f(\alpha)} \left[ \frac{\cosh(w_1^* \alpha x_2)}{\cosh(w_1^* \alpha h_2)} \gamma_7^* \left\{ g_0(\alpha) \frac{\gamma_{11}^*}{\gamma_{12}^*} \right. \right. \\ & + \frac{\gamma_{11}^*}{\gamma_{12}^*} \text{sign}(w_3^*) \beta_{10}^* h_0(\alpha) + \sqrt{\beta_5^*} \lambda_1 \frac{\gamma_{11}^*}{\gamma_{12}^*} \frac{\gamma_2^*}{|w_3^*|} m_0(\alpha) \\ & + \frac{\gamma_{11}^* \lambda_2 \sqrt{\beta_5^*}}{|w_3^*|} n_0(\alpha) \Bigg] + \frac{\cosh(w_3^* \alpha x_2)}{\cosh(w_3^* \alpha h_2)} \gamma_8^* \left\{ \frac{\gamma_{11}^*}{\gamma_{12}^*} g(\alpha) + \frac{\gamma_{11}^*}{\gamma_{12}^*} \text{sign}(w_3^*) \right. \\ & \left. \left. \beta_{10}^* h(\alpha) + \lambda_1 \sqrt{\beta_5^*} \frac{\gamma_{11}^*}{\gamma_{12}^*} \frac{\gamma_2^*}{|w_3^*|} m(\alpha) + \frac{\gamma_{11}^* \lambda_2 \sqrt{\beta_5^*}}{|w_3^*|} n(\alpha) \right\} \right] \end{aligned}$$

The kernels  $k_j$  ( $j = 1, 4$ ) used in Eqs. (6.15a,b):

$$K_1(n, y) = \frac{1}{2\gamma_{13}} \int_0^\infty \left[ \gamma_1 e^{-|w_1|\alpha y/\sqrt{\beta_5}} - \gamma_2 \frac{\gamma_{11}}{\gamma_{12}} e^{-|w_3|\alpha y/\sqrt{\beta_5}} \right] \cos \alpha h_1 \sin \alpha n_1$$

$$+ \frac{1}{2\gamma_{13}} \int_0^\infty \left[ e^{-(h_1-n)\alpha \sqrt{\beta_5}/|w_1|} \left[ - \left[ \gamma_3 g_2(\alpha) + \gamma_4 g_1(\alpha) \right] + \text{sign}(w_1) \beta_9 \left[ \gamma_3 h_2(\alpha) \right. \right. \right.$$

$$\left. \left. \left. + \gamma_4 h_1(\alpha) \right] + \sqrt{\beta_5} \frac{\gamma_1}{|w_1|} \left[ \gamma_3 m_2(\alpha) + \gamma_4 m_1(\alpha) \right] - \frac{\gamma_{11}\sqrt{\beta_5}}{|w_1|} \left[ \gamma_3 n_2(\alpha) + \gamma_4 n_1(\alpha) \right] \right]$$

$$+ e^{-(h_1-n)\alpha \sqrt{\beta_5}/|w_3|} \left[ \frac{\gamma_{11}}{\gamma_{12}} \left[ \gamma_3 g_2(\alpha) + \gamma_4 g_1(\alpha) \right] - \frac{\gamma_{11}}{\gamma_{12}} \text{sign}(w_3) \beta_{10} \left[ \gamma_3 h_2(\alpha) \right. \right. \right.$$

$$\left. \left. \left. + \gamma_4 h_1(\alpha) \right] - \sqrt{\beta_5} \frac{\gamma_{11}}{\gamma_{12}} \frac{\gamma_2}{|w_3|} \left[ \gamma_3 m_2(\alpha) + \gamma_4 m_1(\alpha) \right] + \frac{\gamma_{11}\sqrt{\beta_5}}{|w_3|} \left[ \gamma_3 n_2(\alpha) + \gamma_4 n_1(\alpha) \right] \right] \right] \frac{\cos \alpha y}{f(\alpha)} d\alpha$$

$$K_2(n, y) = \frac{1}{2\gamma_{13}^*} \int_0^\infty \left[ \left\{ e^{-(h_2-n)\alpha \sqrt{\beta_5^*}/|w_1^*|} - e^{-(h_2+n)\alpha \sqrt{\beta_5^*}/|w_1^*|} \right\} \right.$$

$$\left. \left\{ - \left[ \gamma_3 g_2(\alpha) + \gamma_4 g_1(\alpha) \right] - \text{sign}(w_1^*) \beta_9^* \left[ \gamma_3 h_2(\alpha) + \gamma_4 h_1(\alpha) \right] \right. \right.$$

$$\left. \left. - \sqrt{\beta_5^*} \frac{\gamma_{11}^*}{|w_1^*|} \left[ \gamma_3 m_2(\alpha) \right. \right. \right.$$

$$\left. \left. \left. + \gamma_4 m_1(\alpha) \right] - \frac{\gamma_{11}^* \sqrt{\beta_5^*} \lambda_2}{|w_1^*|} \left[ \gamma_3 n_2(\alpha) + \gamma_4 n_1(\alpha) \right] \right] + \left\{ e^{-(h_2-n)\alpha \sqrt{\beta_5^*}/|w_3^*|} \right.$$

$$\left. - e^{-(h_2+n)\alpha \sqrt{\beta_5^*}/|w_3^*|} \right\} \left\{ \frac{\gamma_{11}^*}{\gamma_{12}^*} \left[ \gamma_3 g_2(\alpha) + \gamma_4 g_1(\alpha) \right] \right. \right.$$

$$\left. \left. + \frac{\gamma_{11}^*}{\gamma_{12}^*} \text{sign}(w_3^*) \beta_{10}^* \left[ \gamma_3 h_2(\alpha) \right. \right. \right]$$

$$\left. \begin{aligned} & + \gamma_4 h_1(\alpha) \\ & + \gamma_4 n_1(\alpha) \end{aligned} \right\} + \sqrt{\beta_5^*} \frac{\gamma_{11}^*}{\gamma_{12}^*} \frac{\gamma_2^* \lambda_1}{|w_3^*|} \left[ \gamma_3 m_2(\alpha) + \gamma_4 m_1(\alpha) \right] + \frac{\gamma_{11}^* \sqrt{\beta_5^*} \lambda_2}{|w_3^*|} \left[ \gamma_3 n_2(\alpha) \right]$$

$$K_3(n, y) = \frac{1}{2\gamma_{13}} \int_0^\infty \gamma_{11} \left[ e^{-|w_1|\alpha y/\sqrt{\beta_5}} - e^{-|w_3|\alpha y/\sqrt{\beta_5}} \right] \sin a h_1 \sin a n_2$$

$$+ \frac{1}{2\gamma_{13}} \int_0^\infty \left[ e^{-(h_1-n)\alpha\sqrt{\beta_5}/|w_1|} \left( -\left[ \gamma_9 \tanh(w_1 \alpha h_1) g_2(\alpha) \right. \right. \right.$$

$$\left. \left. \left. + \gamma_{10} \tanh(w_3 \alpha h_1) g_1(\alpha) \right] \right. \right)$$

$$+ \text{sign}(w_1) \beta_9 \left[ \gamma_9 \tanh(w_1 \alpha h_1) h_2(\alpha) + \gamma_{10} \tanh(w_3 \alpha h_1) h_1(\alpha) \right]$$

$$+ \sqrt{\beta_5} \frac{\gamma_1}{|w_1|} \left[ \gamma_9 \tanh(w_1 \alpha h_1) m_2(\alpha) + \gamma_{10} \tanh(w_3 \alpha h_1) m_1(\alpha) \right]$$

$$- \frac{\gamma_{11} \sqrt{\beta_5}}{|w_1|} \left[ \gamma_9 \tanh(w_1 \alpha h_1) n_2(\alpha) \right]$$

$$+ \gamma_{10} \tanh(w_3 \alpha h_1) n_1(\alpha) \right] + e^{-(h_1-n)\alpha\sqrt{\beta_5}/|w_3|} \left( \frac{\gamma_{11}}{\gamma_{12}} \left[ \gamma_9 \tanh(w_1 \alpha h_1) g_2(\alpha) \right. \right.$$

$$\left. \left. + \gamma_{10} \tanh(w_3 \alpha h_1) g_1(\alpha) \right] - \frac{\gamma_{11}}{\gamma_{12}} \text{sign}(w_3) \beta_{10} \left[ \gamma_9 \tanh(w_1 \alpha h_1) h_2(\alpha) \right. \right.$$

$$\left. \left. + \gamma_{10} \tanh(w_3 \alpha h_1) \right] \right)$$

$$h_1(\alpha) - \sqrt{\beta_5} \frac{\gamma_{11}}{\gamma_{12}} \frac{\gamma_2}{|w_3|} \left[ \gamma_9 \tanh(w_1 \alpha h_1) m_2(\alpha) + \gamma_{10} \tanh(w_3 \alpha h_1) m_1(\alpha) \right]$$

$$+ \frac{\gamma_{11}\sqrt{\beta_s}}{|w_3|} \left[ \gamma_9 \tanh(w_1 \alpha h_1) n_2(\alpha) + \gamma_{10} \tanh(w_3 \alpha h_1) n_1(\alpha) \right] \right] \frac{\sin \alpha y}{f(\alpha)} d\alpha$$

$$K_4(n, y) = \frac{1}{2\gamma_{13}^*} \int_0^\infty \left[ \left\{ e^{-(h_2 - n)\alpha \sqrt{\beta_s^*}/|w_1^*|} - e^{-(h_2 + n)\alpha \sqrt{\beta_s^*}/|w_1^*|} \right\} \right.$$

$$\left\{ - \left[ \gamma_9 \tanh(w_1 \alpha h_1) g_2(\alpha) + \gamma_{10} \tanh(w_3 \alpha h_1) g_1(\alpha) \right] - \text{sign}(w_1^*) \beta_9^* \right.$$

$$\left. \left[ \gamma_9 \tanh(w_1 \alpha h_1) h_2(\alpha) + \gamma_{10} \tanh(w_3 \alpha h_1) h_1(\alpha) \right] - \sqrt{\beta_s^*} \frac{\gamma_{11}^* \lambda_1}{|w_1^*|} \left( \gamma_9 \tanh(w_1 \alpha h_1) \right. \right.$$

$$\left. \left. m_2(\alpha) + \gamma_{10} \tanh(w_3 \alpha h_1) m_1(\alpha) \right) - \frac{\gamma_{11}^* \sqrt{\beta_s^*} \lambda_2}{|w_1^*|} \left( \gamma_9 \tanh(w_1 \alpha h_1) n_2(\alpha) \right. \right.$$

$$\left. \left. + \gamma_{10} \tanh(w_3 \alpha h_1) n_1(\alpha) \right) \right\} + \left\{ e^{-(h_2 - n)\alpha \sqrt{\beta_s^*}/|w_3^*|} \right.$$

$$\left. - e^{-(h_2 + n)\alpha \sqrt{\beta_s^*}/|w_3^*|} \right\} \left\{ \frac{\gamma_{11}^*}{\gamma_{12}^*} \left( \gamma_9 \tanh(w_1 \alpha h_1) g_2(\alpha) \right. \right.$$

$$\left. \left. + \gamma_{10} \tanh(w_3 \alpha h_1) g_1(\alpha) \right) \right.$$

$$\left. + \frac{\gamma_{11}}{\gamma_{12}^*} \text{sign}(w_3^*) \beta_{10}^* \left( \gamma_9 \tanh(w_1 \alpha h_1) h_2(\alpha) + \gamma_{10} \tanh(w_3 \alpha h_1) h_1(\alpha) \right) \right]$$

$$+ \sqrt{\beta_s^*} \frac{\gamma_{11}^* \gamma_{12}^* \lambda_1}{|w_3^*|} \left( \gamma_9 \tanh(w_1 \alpha h_1) m_2(\alpha) + \gamma_{10} \tanh(w_3 \alpha h_1) m_1(\alpha) \right)$$

$$+ \frac{\gamma_{11}^* \sqrt{\beta_s^*} \lambda_2}{|w_3^*|} \left( \gamma_9 \tanh(w_1 \alpha h_1) n_2(\alpha) + \gamma_{10} \tanh(w_3 \alpha h_1) n_1(\alpha) \right) \right\} \frac{\sin \alpha y}{f(\alpha)} d\alpha$$

## APPENDIX C

### Evaluation of some Integrals

$$\int_0^{\infty} e^{-at} \sin bt dt = \frac{b}{a^2+b^2}, \quad a > 0$$

$$\int_0^{\infty} e^{-at} \cos bt dt = \frac{a}{a^2+b^2}, \quad a > 0$$

$$\int_0^{\infty} \frac{1}{\beta^2+x^2} \cos ax dx = \frac{\pi}{2\beta} e^{-a\beta}$$

$$\int_0^{\infty} \frac{x}{\beta^2+x^2} \sin ax dx = \frac{\pi}{2} e^{-a\beta}$$

$$\int_0^{\infty} \frac{1}{x(\beta^2+x^2)} \sin ax dx = \frac{\pi}{2\beta^2} (1 - e^{-a\beta})$$

$$\int_0^{\infty} e^{-ax} \cosh ay d\alpha = \frac{a}{a^2-y^2}, \quad a > 0$$

## APPENDIX D

### Derivation of the Stress Intensity Factors

#### D.1 Case of Internal Cracks:

The stress intensity factors are defined as:

$$k_a = \lim_{x_1 \rightarrow a} \sqrt{2(x_1-a)} \sigma_{1y}(x_1,0)$$

$$k_b = \lim_{x_2 \rightarrow b} \sqrt{2(x_2-b)} \sigma_{2y}(x_2,0) \quad (D.1a,b)$$

From Eq. (4.16a) we can write:

$$\sigma_{1y}(x_1,0) = \frac{2\gamma_{14} E_y}{\pi(1-\nu_{xy}\nu_{yx})} \int_a^a \frac{\phi(t)}{t-x_1} dt + \sigma_{1y}^0(x_1,0) \quad (D.2)$$

where  $\sigma_{1y}^0(x_1,0)$  is a bounded function.

$$\phi(t) = \frac{F(t)}{\sqrt{a^2-t^2}} = \frac{F(t)e^{\pi i/2}}{(t-a)^{1/2}(t+a)^{1/2}}$$

Define the sectionally holomorphic function,

$$\psi(z) = \frac{1}{\pi} \int_a^a \frac{\phi(t)}{t-z} dt$$

From [14] we obtain:

$$\psi(z) = \frac{F(-a)e^{\pi i/2}}{(2a)^{1/2}(z+a)^{1/2}} - \frac{F(a)}{(2a)^{1/2}(z-a)^{1/2}} + \psi_0(z) \quad (D.3)$$

Using (D.3) to evaluate (D.2) and with the definitions given in (D.1a,b) we have:

$$k_a = -\frac{2Y_{14}E_y}{(1-v_{xy}^*v_{yx}^*)\sqrt{a}} F(a) = -\frac{2Y_{14}E_y}{(1-v_{xy}^*v_{yx}^*)\sqrt{a}} F_0^*(1)$$

and similarly

$$k_b = -\frac{2Y_{14}^*E_y^*}{(1-v_{xy}^*v_{yx}^*)\sqrt{b}} F^*(b) = -\frac{2Y_{14}^*E_y^*}{(1-v_{xy}^*v_{yx}^*)\sqrt{b}} F_0^*(1)$$

## D.2 Case of Broken Laminates

$$k_a = \lim_{x_2 \rightarrow h_2} 2Y(x_2 + h_2)^Y \sigma_{2y}(x_2, 0) \quad (D.4)$$

From Eq. (5.2b)

$$\sigma_{2y}(x_2, 0) = \frac{2Y^*E_y^*}{(1-v_{xy}^*v_{yx}^*)} \int_{-h_1}^{h_1} k_{21s}(x_2, t) \phi(t) dt + \sigma_{2y}^0(x_2, 0) \quad (D.5)$$

where  $\sigma_{2y}^0(x_2, 0)$  is a bounded function.

$$\phi(t) = \frac{F(t)}{(h_1^2 - t^2)^Y} = \frac{F(t)e^{\pi i Y}}{(t-h_1)^Y(t+h_1)^Y}$$

Define:

$$\psi(z) = \frac{1}{\pi} \int_{-h_1}^{h_1} \frac{\phi(t)}{t-z} dt$$

then:

$$\psi(z) = \frac{F(-h_1)e^{i\pi Y}}{(2h_1)^Y \sin \pi Y} \frac{1}{(z+h_1)^Y} - \frac{F(h_1)}{(2h_1)^Y \sin \pi Y} \frac{1}{(z-h_1)^Y} + \psi_0(z) \quad (D.6)$$

Using (D.4), (D.5) and (D.6) we obtain:

$$k_a = \frac{(h_1)^Y Y_{14}^* E_y^* F_0^*(1)}{(1-v_{xy}^* v_{yx}^*) \sin \pi Y} \left[ \lambda_{101} \frac{|w_1|/\sqrt{B_s}}{\left[ \frac{|w_1^*||w_1|}{\sqrt{B_s}} \right]^Y} + \lambda_{102} \frac{|w_1|/\sqrt{B_s}}{\left[ \frac{|w_3^*||w_1|}{\sqrt{B_s}} \right]^Y} \right]$$

$$+ \lambda_{103} \left[ \frac{|w_3|/\sqrt{\beta_s}}{\sqrt{\beta_s}} \right] Y + \lambda_{104} \left[ \frac{|w_3|/\sqrt{\beta_s}}{\sqrt{\beta_s}} \right] Y$$

$k_b$  has the same expression as the one we derived in the previous case.

### D.3 Case of a Crack Crossing the Interface

#### D.3.a Singular Behavior at the Interface:

In this case the definitions of the stress intensity factors are:

$$k_b = \lim_{x_2 \rightarrow c} \sqrt{2(c-x_2)} \sigma_{2y}(x_2, 0)$$

$$k_{xx} = \lim_{y \rightarrow 0^+} y^\beta \sigma_{1x}(h_1, y)$$

and

$$k_{xy} = \lim_{y \rightarrow 0^+} y^\beta \tau_{1xy}(h_1, y) \quad (D.7a, b, c)$$

From Eq. (6.2b) we can write:

$$\sigma_{2y}(x_2, 0) = \frac{2\gamma_{14}^* F_y^*}{(1 - v_{xy}^* v_{yx}^*) \pi} \int_c^{h_2} \frac{\phi^*(t)}{t-x_2} dt + \sigma_{2y}^0(x_2, 0) \quad (D.8)$$

where  $\sigma_{2y}^0(x_2, 0)$  is a bounded function.

$$\phi^*(t) = \frac{F^*(t)}{(t-c)^{\frac{1}{2}} (h_2-t)^\beta}$$

$$\begin{aligned}\psi^*(z) &= \frac{1}{\pi} \int_c^{h_2} \frac{\phi^*(t)}{t-z} dt = \frac{F^*(c)e^{i\pi/2}}{(h_2-c)^\beta \sin \pi/2} \frac{1}{(z-c)^{\frac{1}{2}}} \\ &\quad - \frac{F^*(h_2)}{(h_2-c)^{\frac{1}{2}} \sin \pi \beta} \frac{1}{(z-h_2)^\beta} + \psi_0^*(z)\end{aligned}\quad (D.9)$$

Using (D.9) and (D.8), Eq. (D.7a) yields to:

$$k_b = \frac{2\gamma_{14} E_y^*}{(1-v_{xy}^* v_{yx}^*)} \frac{F^*(c)\sqrt{2}}{(h_2-c)^\beta}$$

or

$$k_b = \frac{2^{1-\beta} \gamma_{14} E_y^* (h_2-c)^{\frac{1}{2}}}{(1-v_{xy}^* v_{yx}^*)} F_0^*(-1)$$

Separating the singular parts of Eqs. (6.15a,b) we have:

$$\begin{aligned}\frac{\pi(1-v_{xy}v_{yx})}{2E_x} \sigma_{1x}(h_1, y) &= \int_{-h_1}^{h_1} K_{1s}(n, y) \phi(n) dn \\ &\quad + \int_c^{h_2} K_{2s}(n, y) \phi^*(n) dn + \sigma_{1x}^0(y) \\ \frac{\pi}{2G_{xy}} \tau_{1xy}(h_1, y) &= \int_{-h_1}^{h_1} K_{3s}(n, y) \phi(n) dn \\ &\quad + \int_c^{h_2} K_{4s}(n, y) \phi^*(n) dn + \tau_{1xy}^0(y)\end{aligned}\quad (D.10a, b)$$

where  $\sigma_{1x}^0(y)$  and  $\tau_{1xy}^0(y)$  are bounded.

Let:  $\phi(t) = \frac{F(t)}{(h_1^2 - t^2)^\beta} = \frac{F(t)e^{i\pi\beta}}{(t+h_1)^\beta(t-h_1)^\beta}$

then

$$\psi(z) = \frac{1}{\pi} \int_{-h_1}^{h_1} \frac{\phi(t)}{t-z} dt = \frac{F(-h_1)e^{i\pi\beta}}{(2h_1)^\beta \sin\pi\beta} \frac{1}{(z+h_1)^\beta} - \frac{F(h_1)}{(2h_1)^\beta \sin\pi\beta(z-h_1)^\beta} + \psi_0(z)$$

(D.11)

Using (D.11), (D.9), (D.10a,b) and the definitions (D.7b,c)  
after lengthy algebra we obtain:

$$k_{xx} = \frac{E_x}{(1-v_{xy}v_{yx})} \frac{1}{2^{\beta+1} \sin\pi\beta/2} \left\{ \frac{h_1^\beta}{Y_{13}} \left[ \frac{Y_1}{\left(\frac{|w_1|}{\sqrt{\beta_s}}\right)^\beta} - Y_2 \frac{Y_{11}}{Y_{12}} \frac{1}{\left(\frac{|w_3|}{\sqrt{\beta_s}}\right)^\beta} \right] \right.$$

$$- \frac{|w_1|}{\sqrt{\beta_s}} \frac{(Y_3\lambda_{81} + Y_4\lambda_{82})}{\left(\frac{|w_1|}{\sqrt{\beta_s}}\right)^\beta \lambda_{80}} - \frac{|w_3|}{\sqrt{\beta_s}} \frac{(Y_3\lambda_{83} + Y_4\lambda_{84})}{\left(\frac{|w_3|}{\sqrt{\beta_s}}\right)^\beta \lambda_{80}} \Big] F_0(-1)$$

$$\left. + \frac{(h_2-c)^\beta}{\sqrt{2} Y_{13}^* \lambda_{80}} \left[ \frac{|w_1^*|}{\sqrt{\beta_s^*}} \frac{(Y_3\lambda_{89} + Y_4\lambda_{90})}{\left(\frac{|w_1^*|}{\sqrt{\beta_s^*}}\right)^\beta} + \frac{|w_3^*|}{\sqrt{\beta_s^*}} \frac{(Y_3\lambda_{91} + Y_4\lambda_{92})}{\left(\frac{|w_3^*|}{\sqrt{\beta_s^*}}\right)^\beta} \right] F_0^*(1) \right\}$$

and

$$k_{xy} = \frac{G_{xy}}{2^{\beta+1} \cos \frac{\pi\beta}{2}} \left\{ \frac{h_1^\beta}{Y_{13}} \left[ - \frac{Y_{11}}{\left(\frac{|w_1|}{\sqrt{\beta_s}}\right)^\beta} + \frac{Y_{11}}{\left(\frac{|w_3|}{\sqrt{\beta_s}}\right)^\beta} \right] \right.$$

$$- \frac{(Y_9\lambda_{81} + Y_{10}\lambda_{82})}{\lambda_{80} \left(\frac{|w_1|}{\sqrt{\beta_s}}\right)^\beta} \frac{|w_1|}{\sqrt{\beta_s}} - \frac{(Y_9\lambda_{83} + Y_{10}\lambda_{84})}{\lambda_{80} \left(\frac{|w_3|}{\sqrt{\beta_s}}\right)^\beta} \frac{|w_3|}{\sqrt{\beta_s}} \Big] F_0(-1)$$

$$+ \frac{(h_2 - c)^{\beta}}{\sqrt{2} \gamma_{13}^* \lambda_{80}} \left[ \frac{(\gamma_9 \lambda_{89} + \gamma_{10} \lambda_{90})}{\left( \frac{|w_1^*|}{\sqrt{\beta_s^*}} \right)^{\beta}} \frac{|w_1^*|}{\sqrt{\beta_s^*}} + \frac{(\gamma_9 \lambda_{91} + \gamma_{10} \lambda_{92})}{\left( \frac{|w_3^*|}{\sqrt{\beta_s^*}} \right)^{\beta}} \frac{|w_3^*|}{\sqrt{\beta_s^*}} \right] F_0^*(1) \right\}$$

### D.3.b Non-Singular Behavior at the Interface

$$k_b = \lim_{x_2 \rightarrow \infty} \sqrt{2(x_2 - \ell)} \sigma_{2y}(x_2, 0)$$

The derivation is similar to the one done for the case of internal cracks. Thus, we have:

$$k_b = - \frac{2 \gamma_{14}^* E_y^* \sqrt{\ell}}{(1 - v_{xy}^* v_{yx}^*)} H_0(1)$$

VITA

Mr. Feridun Delale

[REDACTED] graduating the Saint Joseph College in Istanbul in 1966, he attended the Technical University of Istanbul where he received his B.S. in Civil Engineering in July 1970 and M.S. in the same field in August 1971.

From November 1971 to January 1974 Mr. Delale was employed by the Technical University of Istanbul as a Research and Teaching Assistant. He was then awarded a fellowship by the same University to pursue further studies in the U.S.A. He is a member of the Turkish Chamber of Civil Engineers and the co-author of a book entitled "Solution to Problems in Strength of Materials", published by the Technical University of Istanbul in June 1974.

He is presently a Research Assistant in the Department of Mechanical Engineering and Mechanics at Lehigh University, Bethlehem, Pa.